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RESEARCH MEMORANDUM

FORCE AND PRESSURE MEASUREMENTS ON SEVERAL
CANOPY-FUSELAGE CONFIGURATIONS AT
MACH NUMBERS 1.41 AND 2.01

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FORCE AND PRESSURE MEASUREMENTS ON SEVERAL
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MACH NUMBERS 1.41 AND 2.01

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SUMMARY

An investigation has been conducted in the Langley 4- by 4-foot supersonic pressure tunnel on canopy pressures and canopy-fuselage forces and moments under conditions of combined pitch and sideslip. The canopy configurations tested varied in windshield shape (flat, vee-, and round), location on the fuselage, and fineness ratio. All configurations were tested at Mach numbers of 1.41 and 2.01 at Reynolds numbers of 1.74×10^6 and 1.44×10^6 , respectively, based on fuselage major diameter.

Drags of the canopy-fuselage combinations varied from lowest for the flat-windshield configuration to highest for the vee-windshield configuration. For comparable canopies, the configurations with the forward canopy location produced less drag than those with the rearward-located canopies, regardless of windshield shape. The effects on drag of windshield shape and canopy location were diminished with increase in Mach number from 1.41 to 2.01.

INTRODUCTION

Because of the high air loads and temperatures associated with supersonic flight, the best compromise of aerodynamic, structural, and visibility requirements in the design of canopies for military aircraft is critically dependent on the accuracy with which loads and aerodynamic characteristics can be predicted. Since practical methods for the calculation of pressure distributions and forces on such arbitrary shapes are limited, experimental data are required. A few papers showing experimental results are at present available, among them references 1 and 2 which deal with pressure distributions of two rather specialized canopy configurations at supersonic speeds. Reference 3 is concerned with

transonic and supersonic drag comparisons of forward and rearward locations of a canopy on a finned test vehicle. A free-flight drag investigation of windshield-shape effects at transonic and low supersonic speeds is reported in reference 4. Reference 5 deals with the location of a canopy in order to improve the longitudinal development of cross-sectional area for a wing-fuselage combination at transonic speeds.

The present investigation is part of a program of the National Advisory Committee for Aeronautics to determine some of the effects at transonic and supersonic speeds of windshield shape, canopy location, fineness ratio, pitch, sideslip, and Mach number on the aerodynamic characteristics of several canopy-fuselage configurations and on the pressure distributions on the canopies. Reference 6 reports the force and moment characteristics at transonic speeds of some of the configurations of the present investigation. The present tests were made of models with flat, vee-, and round windshield canopies in forward and rearward locations on the fuselage. The fineness ratios of the various canopies were approximately 7.0, 10.0, and 12.0 (based on the ratio of the diameter of an equivalent body of revolution to the length of the canopy in the plane of symmetry). All configurations were tested at Mach numbers of 1.41 and 2.01 at Reynolds numbers of 1.74×10^6 and 1.44×10^6 , respectively, based on fuselage major diameter. Two canopy-fuselage configurations and the fuselage alone were tested for angles of attack from -6° to 12° , and all configurations were tested at 0° , -4° , and -8° sideslip at both 0.4° and 6.5° angle of attack. In all tests, boundary-layer transition was fixed 1/2 inch behind the fuselage nose point by means of a roughness strip.

SYMBOLS

| | |
|----------|---|
| M | free-stream Mach number |
| q | free-stream dynamic pressure |
| p_o | free-stream static pressure |
| p | local pressure |
| P | pressure coefficient, $\frac{p - p_o}{q}$ |
| α | angle of attack, deg |
| β | angle of sideslip, deg |
| x | distance from foremost point of canopy in plane of symmetry in an axial direction |

| | |
|-----------------|---|
| x_b | distance from fuselage nose point in an axial direction |
| l | canopy-profile length in an axial direction |
| l_b | fuselage length |
| ϕ | lateral angle measured from plane of symmetry (see tables X, XI, XVII, and XVIII) |
| A_b | area of base of model |
| A_{\max} | maximum cross-sectional area of canopy or of a body of revolution |
| C_N | normal-force coefficient, $\frac{Z}{qA_b}$ |
| C_c | axial-force coefficient, $\frac{X}{qA_b}$ |
| C_Y | lateral-force coefficient, $\frac{Y}{qA_b}$ |
| C_m | pitching-moment coefficient, $\frac{M'}{qA_b l_b}$ |
| C_n | yawing-moment coefficient, $\frac{N}{qA_b l_b}$ |
| C_l | rolling-moment coefficient, $\frac{L}{qA_b l_b}$ |
| C_{Df} | drag coefficient, $\frac{D_f}{qA_b}$ |
| C_D | drag coefficient, $\frac{D}{qA_b}$ |
| ΔC_D | incremental drag coefficient, $\frac{D - D_f}{qA_b}$ |
| C_{DA} | drag coefficient, $\frac{D}{qA_{\max}}$ |
| ΔC_{DA} | incremental drag coefficient, $\frac{D - D_f}{qA_{\max}}$ |

| | |
|----------------|--|
| X | force along body axis, positive when rearward |
| Y | force along lateral axis, positive when starboard |
| Z | force normal to XY-plane, positive when upward |
| D _f | force on fuselage alone in streamwise direction, positive when rearward |
| D | force in streamwise direction, positive when rearward |
| M' | moment about Y-axis, positive when tending to lift nose |
| N | moment about Z-axis, positive when tending to produce a right turn |
| L | moment about X-axis, positive when tending to produce a right bank |
| K | longitudinal location of maximum cross-sectional area, percent of length |
| P.L. | designation of canopy-fuselage parting line |

MODELS AND INSTRUMENTATION

Basic Model and Canopies

The canopy shapes were tested on a drooped-nose-fuselage forebody having an elliptic cross section. Drawings and dimensions of this body, and the base plug which was used to minimize base-pressure corrections, are shown in figures 1 and 2. The various canopy configurations are described in figures 1 to 5. A family of six canopies of approximately the same size, fineness ratio (7.0), and profile was tested. Canopies with flat, vee-, and round windshields were tested at two longitudinal locations on the fuselage. Two smaller flat-windshield canopies of lower windshield slope having fineness ratios of about 10.0 and 12.0 were tested in forward and rearward locations, respectively, on the fuselage. These configurations, which are described in figures 4 and 5 approximate existing supersonic designs. Photographs of all the models are presented in figure 6.

Instrumentation

The forces and moments on the models were measured by means of a six-component strain-gage balance mounted within the fuselage. Moments were measured about a point on the model axis 14.81 inches from the nose.

Pressure instrumentation was provided in each model. The pressure orifices, which were encircled with ink prior to being photographed, may be seen in figure 6. This instrumentation was provided on only one side of the plane of symmetry so that both positive and negative sideslip angles were tested in order to determine the pressures on both the upstream and the downstream sides of the model for a given sideslip angle. The locations of the orifices for each model may be determined from tables X to XVIII.

Small prisms were mounted on the surface of the fuselage so that either angle of attack or angle of sideslip might be measured by a spectrometer head.

TESTS

Test Conditions

| | |
|--|--------------------|
| Mach numbers | 1.41 and 2.01 |
| Reynolds number per foot at M = 1.41 | 4.18×10^6 |
| Reynolds number per foot at M = 2.01 | 3.46×10^6 |
| Stagnation pressure, atm | 0.95 |
| Stagnation temperature, °F | 100 |

Corrections and Accuracy

Although force and moment data were taken at both positive and negative sideslip angles, the subsequent tabulations and plots show only one value for forces and moments and, essentially, only negative sideslip angles. Both sets of values, however, have been used; the data for all positive sideslip angles greater than 0.3° have been folded and averaged with data for negative angles.

Where angles of attack or sideslip could not be measured optically, the calibrated deflections of the balance under load were applied to the no-wind calibration of the angle mechanism so that the estimated angle accuracy was within $\pm 0.15^\circ$.

Base-pressure measurements were made and axial-force data were corrected to correspond to a base pressure equal to free-stream static pressure.

The force and moment coefficients are believed to be correct within the following limits:

| | |
|-------|---------|
| C_N | ±0.0080 |
| C_C | ±0.0040 |
| C_m | ±0.0020 |
| C_l | ±0.0015 |
| C_n | ±0.0040 |
| C_Y | ±0.0095 |
| C_D | ±0.0040 |

RESULTS AND DISCUSSION

Force and Moment Data

The six force and moment coefficients based on the body-axis system plus the drag coefficient based on the wind axis are tabulated and presented in tables I to IX for all model configurations. Because of the large amount of data and because drag considerations appear of greatest general interest, incremental drag coefficients (difference between the drag coefficients for the body alone and those for a canopy-fuselage combination) are the only force data discussed.

Figure 7 shows incremental drag coefficients plotted against sideslip angle for all canopy-fuselage configurations at various Mach numbers and angles of attack. Drags of the configurations with the three windshield shapes varied from the lowest for the flat-windshield configurations to the highest for the vee-windshield configurations except for the configurations with the forward-located canopies at $M = 2.01$ where the differences were about the same as the estimated possible inaccuracies of the data. For example, at $M = 1.41$ and $\alpha = 0.4^\circ$ for the forward-located canopy, the incremental drag coefficient for the flat-windshield canopy was about 75 percent of that for the vee-windshield canopy. For the large canopies, the configurations with the forward-located canopies produced less drag than those with the rearward-located canopies, regardless of windshield shape. The effects of both windshield shape and canopy location were less at $M = 2.01$ than at $M = 1.41$.

For the small canopies, the effects of location are not readily apparent in figure 7 because of differences in fineness ratio and size. In order to obtain an indication of the effects of position and fineness ratio for the flat-windshield canopies, incremental drag coefficients for zero angle of attack and sideslip were based on the maximum cross-sectional areas of the canopies themselves and are given in the following table:

| Flat-windshield canopy | | | | ΔC_{DA} at - | |
|------------------------|----------|----------------|-----------------------|----------------------|----------|
| Size | Location | Fineness ratio | A_{max} , sq in. | M = 1.41 | M = 2.01 |
| Large | Forward | 6.91 | 2.59 | 0.360 | 0.436 |
| Small | Forward | 10.04 | 1.49 | .237 | .312 |
| Large | Rearward | 7.06 | 2.46 | .535 | .543 |
| Small | Rearward | 12.06 | 1.03 | .351 | .381 |

It is apparent from this table that the forward location was also the more favorable for the small canopies. Reference 3 which presents transonic and supersonic drag comparisons of forward and rearward locations of a canopy on a finned test vehicle indicates that in the low supersonic range a rearward canopy location produces less drag. This is in contrast to the indications of the present investigation.

The $M = 1.41$ values from the preceding table have been plotted for all the flat-windshield configurations in figure 8 which also shows from reference 7 some $M = 1.40$ drag values for bodies of revolution having various locations of maximum cross-sectional area and various fineness ratios. It should be noted that the data from reference 7 are concerned with drags of bodies alone; whereas, the present data relating to canopies include mutual interference effects. Figure 8 seems to indicate that interference effects for the forward-located canopies were small compared to interference effects for the rearward location. Figure 8 also appears to show that the drag differential between the large and small canopy configurations is principally a fineness-ratio effect. The location of maximum cross-sectional area (K in fig. 8), which would in most cases be closely related to windshield slope, would be governed largely by visibility requirements. It would appear that an efficient canopy shape on a canopy-fuselage combination would require a low windshield slope and a fineness ratio of 10 or more.

Pressure Data

All pressure coefficient data for each configuration are presented in tables X to XVIII from which plots of pressure coefficient may be readily made along longitudinal meridians or radially about a particular station. Plots of these coefficients along longitudinal meridians (see tables X, XI, XVII, and XVIII for description) are presented against axial location for various angles of attack and sideslip and for Mach numbers of 1.41 and 2.01 in figures 9 to 17.

Figure 9, 10, 11, and 12 show the pressure-coefficient distributions for the large canopies at Mach numbers of 1.41 and 2.01 and indicate that pressure distributions over the aft portions of the canopies were generally not significantly influenced by windshield shape. Local peak suction were generally highest for the vee-windshield configurations although the large flat-windshield configurations began to show appreciable peaks as sideslip angle increased.

Figures 13 and 14 show pressure-coefficient distribution for the small canopies and for the fuselage alone. These, in addition to figures 9 to 12, show that suction peaks in pressure-coefficient distributions at $M = 2.01$ are generally smaller than those at $M = 1.41$, although the character of the remainder of these distributions at low sideslip angles, especially for positive coefficients, remained much the same. Figures 15, 16, and 17 show the effects of angle of attack on pressure-coefficient distributions for a forward-located round-windshield canopy, a rearward-located round-windshield canopy, and the fuselage alone, respectively. The variation of pressure coefficients over this range of angle of attack (-6.0° to 12.0°) appears to be systematic for these configurations.

Force and Pressure Correlation

A comparison of force and pressure-measurement results was made where there existed identical conditions of pitch and sideslip near zero angle of attack for both force and pressure data. Measured fuselage-alone axial-force data were diminished by the axial forces integrated from the limited pressure data on the fuselage within the area which would be covered by the canopies. The axial forces from pressures on the canopies were added to these corrected fuselage axial forces so that integrated configuration drags for the canopy-fuselage combinations resulted. These integrated values are compared with drag coefficients from force measurements in the following table:

| Canopy configuration | Drag coefficient, C_D | | | |
|----------------------|-------------------------|------------|------------|------------|
| | $M = 1.41$ | | $M = 2.01$ | |
| | Measured | Integrated | Measured | Integrated |
| Large forward flat | 0.1695 | 0.1719 | 0.1900 | 0.1813 |
| Large forward vee- | .1879 | .1883 | .1971 | .1900 |
| Large forward round | .1781 | ---- | .1954 | .1800 |
| Large rearward flat | .1927 | .1772 | .2034 | .1831 |
| Large rearward vee- | .2178 | .1893 | .2160 | .1933 |
| Large rearward round | .1982 | .1802 | .2087 | ---- |
| Small forward flat | .1328 | .1417 | .1475 | .1424 |
| Small rearward flat | .1342 | .1289 | .1420 | .1365 |

The appreciable difference between measured and calculated forces for most of the rearward-located canopies gives credence to the supposition of larger fuselage interference effects for these rearward locations in the previous discussion of force data. In the tabulation both force and pressure-measurement results indicate that the flat-windshield canopy configurations produced less drag than the vee-configurations. The lower chord force for the flat-windshield canopy is associated with the expansions around the edges of the windshield resulting in lower pressures over the remaining two-thirds (approximately) of the canopy frontal projection. This effect is seen in figures 18, 19, and 20 which show pressure contours on half the frontal projections of the forward-located large canopies, the rearward-located large canopies, and on the small canopies, respectively. In contrast to those for the flat canopies, it is indicated by the vee-canopy contours that the expansion around the edges of the vee-windshield has little effect on forces in an axial direction. In reference 4 the drag increments for the flat-windshield canopies of comparable windshield-profile slopes were higher than for the vee-windshield canopies, in contrast to present results; however, the frontal-areas of the flat windshields of reference 4 contributed nearly all of the total canopy frontal-area so that expansions around the windshield edges could not produce reductions in canopy drags.

CONCLUSIONS

Force and pressure measurements have been made on several canopy-fuselage configurations which varied in windshield shape (flat, vee-, and round), canopy location on the fuselage, and fineness ratio. All configurations were tested in pitch and sideslip at Mach numbers of 1.41 and 2.01 for values of Reynolds number based on fuselage major diameter of 1.74×10^6 and 1.44×10^6 , respectively. The results of the tests on these configurations indicate the following conclusions:

1. For canopies which varied only in windshield shape, drags were lowest for the flat-windshield configuration and highest for the vee-configuration.
2. For comparable canopies, the configurations with the forward canopy locations produced less drag than those with the rearward-located canopies, regardless of windshield shape.

3. Both the effect of windshield shape and of canopy location were diminished with the increasing of Mach number from 1.41 to 2.01.

Langley Aeronautical Laboratory,
National Advisory Committee for Aeronautics,
Langley Field, Va., August 11, 1955.

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TABLE I.- FORCE AND MOMENT COEFFICIENTS FOR BODY ALONE

| M | α , deg | β , deg | C_N | C_c | C_m | C_l | C_n | C_Y | C_D |
|------|----------------|---------------|--------|-------|--------|--------|--------|--------|-------|
| 1.41 | 0.4 | 0 | .0052 | .1101 | -.0274 | -.0002 | .0013 | .0104 | .1101 |
| 1.41 | 0.4 | -4 | .0051 | .1087 | -.0278 | -.0020 | .0437 | .2082 | .1230 |
| 1.41 | 0.4 | -8 | .0026 | .1044 | -.0293 | -.0041 | .0870 | .4486 | .1659 |
| 1.41 | 0.4 | 0 | .0000 | .1094 | -.0274 | -.0002 | .0013 | .0124 | .1094 |
| 1.41 | 6.5 | 0 | .2234 | .1075 | .0179 | .0001 | -.0049 | -.0052 | .1321 |
| 1.41 | 6.5 | -4 | .2195 | .1033 | .0180 | -.0061 | .0427 | .2166 | .1423 |
| 1.41 | 6.5 | -8 | .2310 | .0968 | .0166 | -.0121 | .0831 | .4692 | .1865 |
| 1.41 | 6.5 | 0 | .2234 | .1072 | .0180 | .0002 | -.0049 | -.0052 | .1318 |
| 1.41 | -6.0 | 0.3 | -.2312 | .1104 | -.0741 | -.0001 | -.0007 | -.0188 | .1341 |
| 1.41 | -3.0 | 0.3 | -.1233 | .1130 | -.0509 | -.0001 | -.0021 | -.0188 | .1194 |
| 1.41 | 0 | 0.3 | -.0154 | .1100 | -.0276 | .0000 | -.0035 | -.0177 | .1101 |
| 1.41 | 3.0 | 0.3 | .0899 | .1098 | -.0052 | .0002 | -.0052 | -.0167 | .1144 |
| 1.41 | 6.0 | 0.3 | .1979 | .1067 | .0177 | .0004 | -.0065 | -.0177 | .1269 |
| 1.41 | 9.0 | 0.3 | .3210 | .0986 | .0377 | .0007 | -.0076 | -.0188 | .1477 |
| 1.41 | 12.0 | 0.3 | .4675 | .0922 | .0558 | .0013 | -.0095 | -.0271 | .1875 |
| 2.01 | 0.4 | 0 | -.0250 | .1189 | -.0330 | -.0001 | -.0009 | -.0050 | .1187 |
| 2.01 | 0.4 | -4 | -.0203 | .1220 | -.0335 | -.0020 | .0391 | .2348 | .1380 |
| 2.01 | 0.4 | -8 | -.0343 | .1179 | -.0351 | -.0041 | .0745 | .5284 | .1901 |
| 2.01 | 6.5 | 0 | .2399 | .1162 | .0083 | .0001 | -.0032 | -.0088 | .1426 |
| 2.01 | 6.5 | -4 | .2463 | .1111 | .0073 | -.0060 | .0382 | .2581 | .1559 |
| 2.01 | 6.5 | -8 | .2635 | .1069 | .0048 | -.0116 | .0663 | .5686 | .2139 |
| 2.01 | -6.0 | 0.3 | -.2908 | .1296 | -.0763 | .0001 | -.0005 | -.0063 | .1593 |
| 2.01 | -3.0 | 0.3 | -.1532 | .1219 | -.0547 | .0000 | -.0021 | -.0127 | .1298 |
| 2.01 | 0 | 0.3 | -.0312 | .1178 | -.0338 | .0000 | -.0040 | -.0164 | .1179 |
| 2.01 | 3.0 | 0.3 | | .1205 | -.0123 | .0002 | -.0059 | -.0203 | |
| 2.01 | 6.0 | 0.3 | .2188 | .1169 | .0073 | .0005 | -.0074 | -.0266 | .1393 |
| 2.01 | 9.0 | 0.3 | .3749 | .1102 | .0239 | .0008 | -.0084 | -.0355 | .1677 |
| 2.01 | 12.0 | 0.3 | .5439 | .0908 | .0354 | .0012 | -.0084 | -.0469 | .2021 |
| 2.01 | 0 | 0.3 | -.0312 | .1178 | -.0338 | .0000 | -.0037 | -.0152 | .1179 |

TABLE II.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
FORWARD-LOCATED FLAT-WINDSHIELD CANOPY

| M | α , deg | β , deg | C_N | C_C | C_m | C_L | C_n | C_Y | C_D |
|------|----------------|---------------|--------|-------|--------|--------|--------|--------|-------|
| 1.41 | 0.4 | 0 | -.0077 | .1696 | -.0072 | .0001 | .0011 | .0073 | .1695 |
| 1.41 | 0.4 | -4 | -.0116 | .1780 | -.0070 | .0002 | .0739 | .2243 | .1931 |
| 1.41 | 0.4 | -8 | -.0013 | .1747 | -.0078 | -.0011 | .1482 | .5085 | .2438 |
| 1.41 | 6.5 | 0 | .2198 | .1617 | .0413 | .0001 | -.0040 | .0062 | .1855 |
| 1.41 | 6.5 | -4 | .2160 | .1645 | .0463 | -.0025 | .0661 | .2774 | .2068 |
| 1.41 | 6.5 | -8 | .2198 | .1722 | .0469 | -.0044 | .1263 | .6454 | .2839 |
| 1.41 | 6.5 | 0 | .2173 | .1625 | .0414 | .0001 | -.0005 | .0032 | .1861 |
| 2.01 | 0.4 | 0 | -.0374 | .1903 | -.0258 | .0000 | -.0010 | -.0063 | .1900 |
| 2.01 | 0.4 | -4 | -.0406 | .1945 | -.0271 | .0009 | .0689 | .2544 | .2115 |
| 2.01 | 0.4 | -8 | -.0686 | .2019 | -.0292 | .0024 | .1330 | .5816 | .2804 |
| 2.01 | 0.4 | 0 | -.0437 | .1931 | -.0260 | .0000 | -.0015 | -.0088 | .1928 |
| 2.01 | 6.5 | 0 | .2534 | .1739 | .0119 | .0000 | -.0037 | -.0127 | .2015 |
| 2.01 | 6.5 | -4 | .2635 | .1733 | .0099 | -.0041 | .0606 | .3316 | .2247 |
| 2.01 | 6.5 | -8 | .2315 | .1826 | .0105 | -.0057 | .1123 | .7567 | .3110 |
| 2.01 | 6.5 | 0 | .2596 | .1764 | .0119 | .0001 | -.0038 | -.0127 | .2047 |

TABLE III.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
FORWARD-LOCATED VEE-WINDSHIELD CANOPY

| M | α , deg | β , deg | C_N | C_C | C_m | C_l | C_n | C_Y | C_D |
|------|----------------|---------------|--------|-------|--------|--------|--------|--------|-------|
| 1.41 | 0.4 | 0 | -.0179 | .1880 | -.0096 | .0001 | .0022 | .0073 | .1879 |
| 1.41 | 0.4 | -4 | -.0154 | .1965 | -.0092 | .0003 | .0703 | .2293 | .2119 |
| 1.41 | 0.4 | -8 | -.0499 | .1954 | -.0051 | .0011 | .1434 | .5045 | .2634 |
| 1.41 | 0.4 | 0 | -.0179 | .1901 | -.0093 | .0001 | .0005 | .0051 | .1900 |
| 1.41 | 6.5 | 0 | .2148 | .1776 | .0397 | .0001 | -.0031 | .0083 | .2008 |
| 1.41 | 6.5 | -4 | .2148 | .1823 | .0414 | -.0049 | .0646 | .2873 | .2250 |
| 1.41 | 6.5 | -8 | .1752 | .1825 | .0483 | -.0076 | .1241 | .6685 | .2923 |
| 1.41 | 6.5 | 0 | .2173 | .1792 | .0397 | .0001 | -.0032 | .0073 | .2027 |
| 2.01 | 0.4 | 0 | -.0532 | .1975 | -.0268 | .0000 | -.0010 | -.0063 | .1971 |
| 2.01 | 0.4 | -4 | -.0516 | .1964 | -.0277 | .0007 | .0670 | .2625 | .2139 |
| 2.01 | 0.4 | -8 | -.0734 | .2016 | -.0293 | .0020 | .1280 | .5878 | .2810 |
| 2.01 | 0.4 | 0 | -.0500 | .1994 | -.0268 | -.0001 | -.0012 | -.0102 | .1991 |
| 2.01 | 6.5 | -4 | .2593 | .1751 | .0083 | -.0041 | .0577 | .3231 | .2254 |
| 2.01 | 6.5 | -8 | .2358 | .1834 | .0100 | -.0062 | .1110 | .7505 | .3114 |
| 2.01 | 6.5 | 0 | .2561 | .1797 | .0099 | .0001 | -.0050 | -.0228 | .2075 |

TABLE IV.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
FORWARD-LOCATED ROUND-WINDSHIELD CANOPY

| M | α , deg | β , deg | C_N | C_C | C_m | C_l | C_n | C_Y | C_D |
|------|----------------|---------------|--------|-------|--------|--------|--------|--------|-------|
| 1.41 | 0.4 | 0 | -.0128 | .1782 | -.0054 | .0001 | -.0024 | .0083 | .1781 |
| 1.41 | 0.4 | -4 | -.0167 | .1860 | -.0054 | .0003 | .0737 | .2316 | .2016 |
| 1.41 | 0.4 | -8 | -.0256 | .1881 | -.0046 | -.0012 | .1443 | .5106 | .2572 |
| 1.41 | 6.5 | 0 | .2170 | .1686 | .0425 | .0001 | -.0075 | .0031 | .1921 |
| 1.41 | 6.5 | -4 | .2158 | .1704 | .0442 | -.0025 | .0666 | .2860 | .2132 |
| 1.41 | 6.5 | -8 | .2043 | .1743 | .0502 | -.0044 | .1262 | .6301 | .2821 |
| 1.41 | 6.5 | 0 | .2195 | .1696 | .0425 | .0002 | -.0075 | .0010 | .1934 |
| 1.41 | 0 | 0.3 | .0410 | .1812 | -.0052 | -.0002 | -.0113 | -.0073 | .1812 |
| 1.41 | 3.0 | 0.3 | .0693 | .1768 | .0187 | .0001 | -.0131 | -.0094 | .1802 |
| 1.41 | 6.0 | 0.3 | .1926 | .1707 | .0417 | .0006 | -.0152 | -.0125 | .1900 |
| 1.41 | 9.0 | 0.3 | .3286 | .1655 | .0627 | .0010 | -.0153 | -.0406 | .2151 |
| 1.41 | 12.0 | 0.3 | .5032 | .1494 | .0779 | .0015 | -.0136 | -.0750 | .2511 |
| 1.41 | -3.0 | 0.3 | -.1489 | .1850 | -.0291 | -.0003 | -.0089 | -.0135 | .1926 |
| 1.41 | -6.0 | 0.3 | -.2596 | .1857 | -.0533 | -.0002 | -.0063 | -.0125 | .2129 |
| 1.41 | 0 | 0.3 | -.0385 | .1805 | -.0051 | -.0002 | -.0114 | -.0156 | .1806 |
| 2.01 | 0.4 | 0 | -.0499 | .1957 | -.0249 | -.0001 | -.0044 | -.0038 | .1954 |
| 2.01 | 0.4 | -4 | -.0468 | .1958 | -.0262 | .0008 | .0701 | .2671 | .2136 |
| 2.01 | 0.4 | -8 | -.0687 | .2059 | -.0284 | .0028 | .1322 | .6021 | .2872 |
| 2.01 | 0.4 | 0 | -.0437 | .1957 | -.0249 | -.0001 | -.0044 | -.0038 | .1954 |
| 2.01 | 6.5 | 0 | .2531 | .1766 | .0124 | -.0001 | -.0063 | -.0152 | .2043 |
| 2.01 | 6.5 | -4 | .2640 | .1791 | .0105 | -.0041 | .0603 | .3385 | .2310 |
| 2.01 | 6.5 | -8 | .2406 | .1850 | .0112 | -.0060 | .1107 | .7459 | .3128 |
| 2.01 | 6.5 | 0 | .2593 | .1775 | .0125 | -.0001 | -.0064 | -.0164 | .2057 |
| 2.01 | 0 | 0.3 | -.0593 | .1952 | -.0257 | -.0002 | -.0100 | -.0216 | .1953 |
| 2.01 | 3.0 | 0.3 | .0781 | .1856 | -.0045 | .0000 | -.0126 | -.0317 | .1896 |
| 2.01 | 6.0 | 0.3 | .2281 | .1770 | .0118 | .0003 | -.0143 | -.0443 | .2001 |
| 2.01 | 9.0 | 0.3 | .4125 | .1706 | .0216 | .0006 | -.0141 | -.0672 | .2334 |
| 2.01 | 12.0 | 0.3 | .6406 | .1625 | .0219 | .0006 | -.0110 | -.1015 | .2926 |
| 2.01 | -3.0 | 0.3 | -.1937 | .2055 | -.0475 | -.0001 | -.0068 | -.0140 | .2154 |
| 2.01 | -6.0 | 0.3 | -.3343 | .2187 | -.0705 | .0001 | -.0037 | -.0076 | .2525 |
| 2.01 | 0 | 0.3 | -.0593 | .1942 | -.0257 | -.0002 | -.0100 | -.0229 | .1943 |

TABLE V.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
REARWARD-LOCATED FLAT-WINDSHIELD CANOPY

| M | α , deg | β , deg | C_N | C_C | C_m | C_l | C_n | C_Y | C_D |
|------|----------------|---------------|--------|-------|--------|--------|--------|--------|-------|
| 1.41 | 0.4 | 0 | -.0077 | .1928 | -.0137 | .0000 | .0010 | .0115 | .1927 |
| 1.41 | 0.4 | -4 | -.0064 | .2023 | -.0144 | .0021 | .0760 | .2448 | .2189 |
| 1.41 | 0.4 | -8 | -.0307 | .2027 | -.0142 | .0064 | .1450 | .5459 | .2765 |
| 1.41 | 0.4 | 0 | -.0077 | .1953 | -.0136 | .0000 | .0010 | .0104 | .1952 |
| 1.41 | 6.5 | 0 | .2425 | .1842 | .0243 | .0001 | -.0075 | -.0010 | .2105 |
| 1.41 | 6.5 | -4 | .2349 | .1851 | .0250 | -.0036 | .0627 | .3232 | .2326 |
| 1.41 | 6.5 | -8 | .2234 | .1846 | .0272 | -.0042 | .1160 | .7126 | .3058 |
| 2.01 | 0.4 | 0 | -.0374 | .2037 | -.0366 | .0001 | .0001 | -.0063 | .2034 |
| 2.01 | 0.4 | -4 | -.0343 | .2065 | -.0391 | .0024 | .0646 | .2900 | .2260 |
| 2.01 | 0.4 | -8 | -.0763 | .2115 | -.0412 | .0074 | .1170 | .6691 | .3021 |
| 2.01 | 0.4 | 0 | -.0374 | .2027 | -.0368 | -.0002 | -.0003 | -.0088 | .2024 |
| 2.01 | 6.5 | 0 | .2716 | .1857 | -.0103 | -.0004 | -.0025 | -.0127 | .2153 |
| 2.01 | 6.5 | -4 | .2700 | .1870 | -.0099 | -.0018 | .0456 | .3755 | .2420 |
| 2.01 | 6.5 | -8 | .2248 | .1891 | -.0049 | -.0007 | .0817 | .8279 | .3265 |
| 2.01 | 6.5 | 0 | .2778 | .1858 | -.0103 | -.0004 | -.0023 | -.0127 | .2161 |

TABLE VI.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
REARWARD-LOCATED VEE-WINDSHIELD CANOPY

| M | α , deg | β , deg | C_N | C_C | C_m | C_l | C_n | C_Y | C_D |
|------|----------------|---------------|--------|-------|--------|--------|--------|--------|-------|
| 1.41 | 0.4 | 0 | -.0179 | .2179 | -.0124 | .0002 | .0010 | .0083 | .2178 |
| 1.41 | 0.4 | -4 | -.0167 | .2210 | -.0134 | .0020 | .0734 | .2503 | .2378 |
| 1.41 | 0.4 | -8 | -.0461 | .2141 | -.0114 | .0068 | .1400 | .5736 | .2916 |
| 1.41 | 0.4 | 0 | -.0154 | .2179 | -.0121 | .0002 | .0011 | .0083 | .2178 |
| 1.41 | 6.5 | 0 | .2429 | .1985 | .0236 | .0002 | -.0028 | .0094 | .2247 |
| 1.41 | 6.5 | -4 | .2442 | .2016 | .0238 | .0033 | .0572 | .3362 | .2509 |
| 1.41 | 6.5 | -8 | .2097 | .2013 | .0284 | .0028 | .1053 | .7659 | .3282 |
| 1.41 | 6.5 | 0 | .2455 | .1960 | .0236 | .0004 | -.0049 | .0000 | .2225 |
| 2.01 | 0.4 | 0 | -.0469 | .2163 | -.0350 | .0001 | .0001 | -.0050 | .2160 |
| 2.01 | 0.4 | -4 | -.0438 | .2145 | -.0364 | .0012 | .0629 | .2929 | .2341 |
| 2.01 | 0.4 | -8 | -.0782 | .2188 | -.0373 | .0034 | .1164 | .6725 | .3098 |
| 2.01 | 0.4 | 0 | -.0406 | .2182 | -.0350 | .0001 | -.0001 | -.0026 | .2179 |
| 2.01 | 6.5 | 0 | .2724 | .1948 | -.0093 | -.0001 | -.0011 | .0000 | .2244 |
| 2.01 | 6.5 | -4 | .2693 | .1964 | -.0098 | -.0009 | .0442 | .3785 | .2515 |
| 2.01 | 6.5 | -8 | .2223 | .1941 | -.0028 | -.0001 | .8451 | .8451 | .3336 |
| 2.01 | 6.5 | 0 | .2755 | .1919 | -.0093 | -.0001 | -.0017 | -.0039 | .2219 |

TABLE VII.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
 REARWARD-LOCATED ROUND-WINDSHIELD CANOPY

| M | α , deg | β , deg | C_N | C_c | C_m | C_l | C_n | C_Y | C_D |
|------|----------------|---------------|--------|-------|--------|--------|--------|--------|-------|
| 1.41 | 0.4 | 0 | -.0179 | .1983 | -.0109 | -.0001 | -.0014 | .0062 | .1982 |
| 1.41 | 0.4 | -4 | -.0154 | .2048 | -.0116 | .0017 | .0732 | .2433 | .2212 |
| 1.41 | 0.4 | -8 | -.0359 | .2043 | -.0114 | .0055 | .1397 | .5503 | .2787 |
| 1.41 | 0.4 | 0 | -.0153 | .1973 | -.0110 | -.0001 | -.0015 | .0062 | .1972 |
| 1.41 | 6.5 | 0 | .2380 | .1870 | .0271 | -.0001 | -.0050 | .0083 | .2127 |
| 1.41 | 6.5 | -4 | .2418 | .1896 | .0271 | -.0041 | .0603 | .3217 | .2377 |
| 1.41 | 6.5 | -8 | .2277 | .1901 | .0288 | -.0057 | .1111 | .6946 | .3093 |
| 1.41 | 0 | 0.3 | -.0332 | .1975 | -.0102 | -.0003 | -.0105 | -.0177 | .1976 |
| 1.41 | 3.0 | 0.3 | .0870 | .1917 | .0104 | .0000 | -.0015 | -.0218 | .1961 |
| 1.41 | 6.0 | 0.3 | .2149 | .1863 | .0268 | .0003 | -.0116 | -.0280 | .2079 |
| 1.41 | 9.0 | 0.3 | .3633 | .1810 | .0385 | .0006 | -.0111 | -.0384 | .2358 |
| 1.41 | 12.0 | 0.3 | .5577 | .1700 | .0418 | .0009 | -.0099 | | |
| 1.41 | -3.0 | 0.3 | -.1509 | .1993 | -.0319 | -.0006 | -.0092 | -.0156 | .2070 |
| 1.41 | -6.0 | 0.3 | -.2635 | .2007 | -.0537 | -.0006 | -.0071 | -.0145 | .2272 |
| 1.41 | 0 | 0.3 | -.0332 | .1975 | -.0101 | -.0003 | -.0105 | -.0177 | .1976 |
| 2.01 | 0.4 | 0 | -.0469 | .2090 | -.0334 | -.0004 | -.0040 | -.0039 | .2087 |
| 2.01 | 0.4 | -4 | -.0469 | .2103 | -.0357 | .0021 | .0635 | .2898 | .2297 |
| 2.01 | 0.4 | -8 | -.0766 | .2148 | -.0374 | .0064 | .1142 | .6669 | .3051 |
| 2.01 | 0.4 | 0 | -.0406 | .2129 | -.0334 | -.0005 | -.0041 | -.0063 | .2126 |
| 2.01 | 6.5 | 0 | .2724 | .1898 | -.0069 | -.0002 | -.0132 | -.0203 | .2194 |
| 2.01 | 6.5 | -4 | .2662 | .1896 | -.0066 | -.0024 | .0455 | .3683 | .2437 |
| 2.01 | 6.5 | -8 | .2301 | .1928 | -.0021 | -.0019 | .0802 | .8077 | .3279 |
| 2.01 | 0 | 0.3 | -.0596 | .2076 | -.0341 | -.0005 | -.0068 | -.0204 | .2077 |
| 2.01 | 3.0 | 0.3 | .0847 | .1996 | -.0183 | -.0003 | -.0079 | -.0293 | .2039 |
| 2.01 | 6.0 | 0.3 | .2415 | .1884 | -.0064 | -.0002 | -.0083 | -.0420 | .2128 |
| 2.01 | 9.0 | 0.3 | .4078 | .1845 | .0015 | -.0001 | -.0074 | -.0585 | .2463 |
| 2.01 | 12.0 | 0.3 | .6055 | .1830 | .0072 | -.0001 | -.0046 | -.0815 | .3053 |
| 2.01 | -3.0 | 0.3 | -.1977 | .2144 | -.0523 | -.0004 | -.0051 | -.0141 | .2245 |
| 2.01 | -6.0 | 0.3 | -.3483 | .2258 | -.0719 | -.0003 | -.0031 | -.0089 | .2610 |
| 2.01 | 0 | 0.3 | -.0565 | .2047 | -.0340 | -.0004 | -.0068 | -.0204 | .2048 |

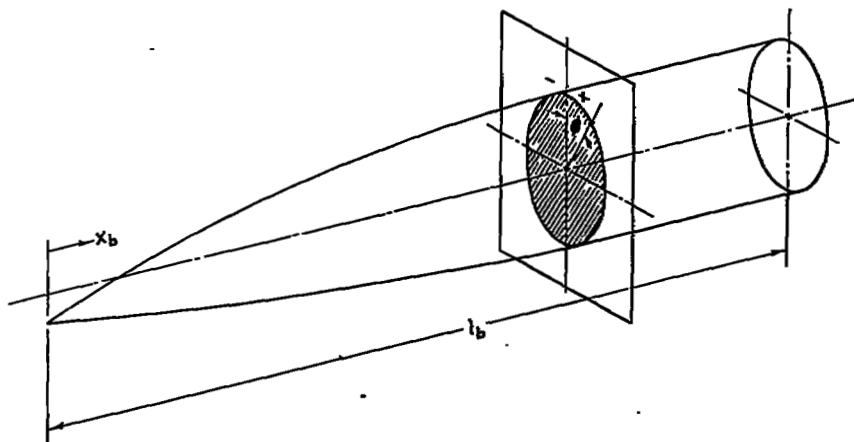
TABLE VIII.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
FORWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY

| M | α , deg | β , deg | C_N | C_c | C_m | C_l | C_n | C_Y | C_D |
|------|----------------|---------------|--------|-------|--------|--------|--------|--------|-------|
| 1.41 | 0.4 | 0 | -.0077 | .1329 | -.0168 | -.0001 | .0005 | .0115 | .1328 |
| 1.41 | 0.4 | -4 | -.0051 | .1338 | -.0173 | -.0009 | .0588 | .2217 | .1489 |
| 1.41 | 0.4 | -8 | -.0321 | .1305 | -.0169 | -.0011 | .1151 | .4772 | .1954 |
| 1.41 | 0.4 | 0 | -.0077 | .1321 | -.0170 | -.0001 | .0004 | .0104 | .1320 |
| 1.41 | 6.5 | 0 | .2202 | .1347 | .0299 | .0002 | -.0051 | .0021 | .1588 |
| 1.41 | 6.5 | -4 | .2151 | .1307 | .0312 | -.0058 | .0557 | .2426 | .1708 |
| 1.41 | 6.5 | -8 | .2074 | .1226 | .0354 | -.0107 | .1065 | .5460 | .2199 |
| 1.41 | 6.5 | 0 | .2228 | .1332 | .0299 | .0002 | -.0049 | .0011 | .1576 |
| 2.01 | 0.4 | 0 | -.0312 | .1477 | -.0285 | -.0001 | -.0024 | -.0089 | .1475 |
| 2.01 | 0.4 | -4 | -.0344 | .1462 | -.0295 | -.0007 | .0549 | .2506 | .1631 |
| 2.01 | 0.4 | -8 | -.0579 | .1475 | -.0315 | -.0012 | .1087 | .5653 | .2244 |
| 2.01 | 0.4 | 0 | -.0312 | .1487 | -.0285 | -.0001 | -.0023 | -.0102 | .1485 |
| 2.01 | 6.5 | 0 | .2441 | .1394 | .0109 | .0000 | -.0047 | -.0127 | .1661 |
| 2.01 | 6.5 | -4 | .2488 | .1446 | .0105 | -.0052 | .0498 | .2870 | .1914 |
| 2.01 | 6.5 | -8 | .2410 | .1453 | .0094 | -.0093 | .0934 | .6619 | .2621 |
| 2.01 | 6.5 | 0 | .2472 | .1420 | .0109 | .0001 | -.0047 | -.0140 | .1691 |

TABLE IX.- FORCE AND MOMENT COEFFICIENTS FOR CONFIGURATION WITH
REARWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY

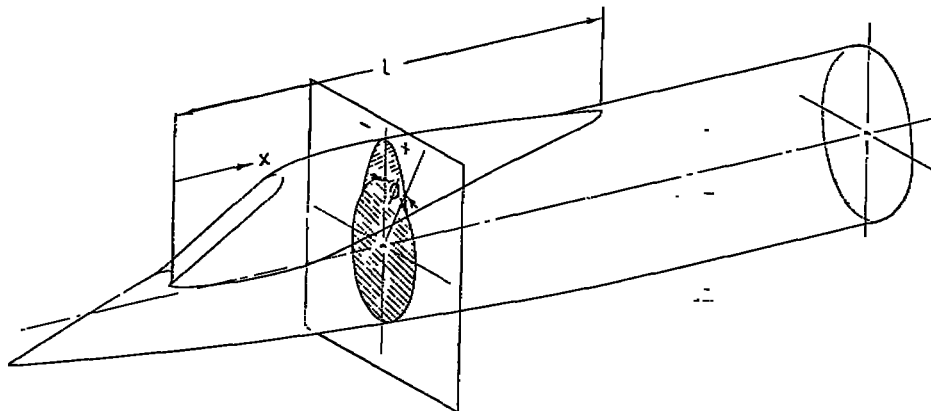
| M | α , deg | β , deg | C_N | C_c | C_m | C_L | C_n | C_Y | C_D |
|------|----------------|---------------|--------|-------|--------|--------|--------|--------|-------|
| 1.41 | 0.4 | 0 | -.0025 | .1342 | -.0227 | -.0001 | .0014 | .0041 | .1342 |
| 1.41 | 0.4 | -4 | -.0039 | .1361 | -.0229 | -.0003 | .0577 | .2220 | .1512 |
| 1.41 | 0.4 | -8 | -.0231 | .1394 | -.0213 | -.0001 | .1119 | .4914 | .2063 |
| 1.41 | 0.4 | 0 | -.0025 | .1326 | -.0226 | .0000 | .0013 | .0041 | .1326 |
| 1.41 | 6.5 | 0 | .2227 | .1259 | .0204 | -.0002 | -.0023 | .0073 | .1503 |
| 1.41 | 6.5 | -4 | .2253 | .1237 | .0215 | -.0055 | .0522 | .2482 | .1654 |
| 1.41 | 6.5 | -8 | .2099 | .1207 | .0254 | -.0096 | .0998 | .5622 | .2206 |
| 1.41 | 6.5 | 0 | .2252 | .1251 | .0204 | -.0001 | -.0023 | .0073 | .1498 |
| 2.01 | 0.4 | 0 | -.0282 | .1422 | -.0342 | -.0001 | -.0004 | -.0076 | .1420 |
| 2.01 | 0.4 | -4 | -.0282 | .1427 | -.0351 | -.0005 | .0525 | .2604 | .1603 |
| 2.01 | 0.4 | -8 | -.0517 | .1464 | -.0371 | .0001 | .0964 | .5804 | .2254 |
| 2.01 | 0.4 | 0 | -.0219 | .1461 | -.0342 | -.0001 | -.0005 | -.0102 | .1459 |
| 2.01 | 6.5 | 0 | .2531 | .1291 | .0014 | -.0001 | -.0022 | -.0076 | .1569 |
| 2.01 | 6.5 | -4 | .2594 | .1296 | .0006 | .0043 | .0420 | .2966 | .1784 |
| 2.01 | 6.5 | -8 | .2359 | .1304 | .0021 | -.0075 | .0729 | .6914 | .2510 |
| 2.01 | 6.5 | 0 | .2562 | .1291 | .0012 | .0000 | -.0023 | -.0101 | .1573 |

TABLE X.- PRESSURE COEFFICIENTS FOR THIN-LINE FLOW

(a) $M=1.41$

| $\frac{x}{c}$ | $\frac{y}{c}$ | .160 | .200 | .240 | .280 | .320 | .360 | .400 | .440 | .480 | .520 | .560 | .600 | .640 | .680 | .720 | .760 | .800 | .840 | .880 | .920 | .960 |
|--------------------------------------|---------------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | | | | | | |
| 0 | | .223 | .171 | .114 | .069 | .028 | .008 | .001 | .003 | -.021 | -.036 | -.044 | -.048 | -.048 | -.047 | -.045 | -.042 | -.039 | -.034 | -.028 | -.021 | -.015 |
| 30 | | .212 | .164 | .113 | .076 | .038 | .025 | .008 | -.003 | -.030 | -.042 | -.049 | -.050 | -.048 | -.044 | -.039 | -.032 | -.024 | -.016 | -.008 | -.001 | -.005 |
| $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | | | | | | |
| 0 | | .209 | .155 | .108 | .076 | .042 | .021 | .000 | -.013 | -.036 | -.053 | -.079 | -.100 | -.099 | -.102 | -.091 | -.079 | -.069 | -.060 | -.049 | -.039 | -.029 |
| 30 | | .223 | .187 | .162 | .107 | .084 | .051 | .030 | .015 | -.021 | -.039 | -.047 | -.043 | -.043 | -.071 | -.065 | -.063 | -.061 | -.061 | -.048 | -.036 | -.026 |
| -30 | | .267 | .219 | .082 | .024 | .008 | -.024 | -.034 | -.044 | -.070 | -.088 | -.091 | -.097 | -.106 | -.093 | -.076 | -.065 | -.060 | -.053 | -.045 | -.035 | -.025 |
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | | | | | | |
| 0 | | .177 | .117 | .099 | .062 | .004 | -.030 | -.043 | -.056 | -.081 | -.097 | -.122 | -.148 | -.156 | -.160 | -.149 | -.136 | -.126 | -.115 | -.105 | -.095 | -.081 |
| 30 | | .215 | .195 | .172 | .123 | .103 | .068 | .040 | .022 | -.006 | -.036 | -.050 | -.070 | -.092 | -.089 | -.088 | -.093 | -.097 | -.099 | -.088 | -.075 | -.065 |
| -30 | | .207 | .058 | .014 | -.036 | -.052 | -.081 | -.094 | -.098 | -.122 | -.137 | -.139 | -.139 | -.144 | -.122 | -.100 | -.087 | -.077 | -.069 | -.058 | -.049 | -.039 |
| $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | | | | | | |
| 0 | | .129 | .089 | .071 | .026 | .000 | -.019 | -.028 | -.035 | -.054 | -.064 | -.081 | -.093 | -.085 | -.080 | -.040 | -.046 | -.037 | -.030 | -.019 | -.011 | -.001 |
| 30 | | .129 | .088 | .076 | .086 | .011 | -.011 | -.025 | -.038 | -.058 | -.072 | -.077 | -.090 | -.094 | -.081 | -.049 | -.059 | -.051 | -.044 | -.030 | -.022 | -.012 |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | | | | | | |
| 0 | | .116 | .072 | .056 | .012 | -.019 | -.036 | -.046 | -.052 | -.071 | -.081 | -.099 | -.111 | -.104 | -.098 | -.076 | -.063 | -.052 | -.043 | -.034 | -.029 | -.019 |
| 30 | | .128 | .092 | .079 | .034 | .018 | -.011 | -.029 | -.045 | -.066 | -.085 | -.096 | -.113 | -.127 | -.120 | -.112 | -.108 | -.101 | -.091 | -.089 | -.083 | -.073 |
| -30 | | .101 | .062 | .046 | -.021 | -.031 | -.031 | -.042 | -.049 | -.068 | -.076 | -.080 | -.086 | -.088 | -.071 | -.057 | -.045 | -.038 | -.032 | -.020 | -.013 | -.003 |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | | | | | | |
| 0 | | .072 | .026 | .017 | -.031 | -.041 | -.087 | -.098 | -.103 | -.123 | -.133 | -.151 | -.167 | -.160 | -.158 | -.125 | -.105 | -.087 | -.072 | -.061 | -.054 | -.046 |
| 30 | | .112 | .088 | .077 | .031 | .016 | -.018 | -.038 | -.061 | -.082 | -.109 | -.127 | -.144 | -.168 | -.170 | -.170 | -.175 | -.175 | -.164 | -.148 | -.138 | -.128 |
| -30 | | .062 | .013 | -.006 | -.045 | -.051 | -.070 | -.076 | -.078 | -.093 | -.096 | -.099 | -.099 | -.099 | -.082 | -.069 | -.068 | -.059 | -.050 | -.045 | -.035 | -.025 |
| $\alpha=-6.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | | | | | | | | | |
| 0 | | .320 | .269 | .235 | .170 | .137 | .097 | .079 | .066 | .034 | .014 | -.018 | -.045 | -.051 | -.059 | -.054 | -.050 | -.041 | -.034 | -.022 | -.015 | -.005 |
| 30 | | .301 | .244 | .195 | .125 | .103 | .081 | .037 | -.025 | -.013 | -.039 | -.049 | -.043 | -.082 | -.077 | -.069 | -.061 | -.056 | -.056 | -.048 | -.042 | -.032 |
| $\alpha=-3.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | | | | | | | | | |
| 0 | | .266 | .214 | .186 | .127 | .096 | .057 | .044 | .032 | .005 | -.014 | -.045 | -.048 | -.071 | -.076 | -.068 | -.058 | -.047 | -.038 | -.026 | -.010 | -.000 |
| 30 | | .251 | .198 | .158 | .096 | .076 | .039 | .020 | .007 | -.025 | -.047 | -.056 | -.070 | -.084 | -.079 | -.068 | -.055 | -.040 | -.034 | -.023 | -.019 | -.008 |
| $\alpha=3.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | | | | | | | | | |
| 0 | | .170 | .130 | .108 | .061 | .031 | .004 | -.007 | -.014 | -.033 | -.044 | -.068 | -.084 | -.081 | -.081 | -.044 | -.048 | -.039 | -.030 | -.020 | -.015 | -.005 |
| 30 | | .172 | .128 | .101 | .052 | .036 | .007 | -.008 | -.018 | -.040 | -.058 | -.064 | -.078 | -.098 | -.076 | -.043 | -.056 | -.045 | -.041 | -.031 | -.022 | -.012 |

TABLE XI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED FLAT-WINDSHIELD CANOPY

(a) $M=141$

| x/l | .000 | .001 | .004 | .006 | .017 | .016 | .052 | .164 | .216 | .238 | .240 | .260 | .321 | .356 | .440 | .600 | .828 | .988 |
|---------------------|------------------------------------|------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|------|------|
| β, deg | $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | |
| P.L. | .497 | | | | | .254 | .119 | .156 | | | .104 | | -.039 | -.092 | | -.025 | .046 | |
| 57½ | .497 | | | | .454 | .216 | .173 | | | | | | -.039 | | | | | |
| 45 | -.001 | | | | | | | | | | | | -.059 | -.092 | | | | |
| 30 | | .901 | | .565 | .432 | .313 | | | | | .100 | | -.159 | -.122 | | | | |
| 15 | | | | | .547 | .287 | .176 | | | | .027 | -.121 | -.172 | -.183 | | -.025 | | |
| 10 | | | | | .505 | | | | | | | | | | | | | |
| 7 | | | | | | .313 | | | | | | | | | | | | |
| 3 | | | | | | .556 | .244 | | | -.041 | -.150 | -.188 | -.157 | -.102 | -.002 | | | |
| 0 | | | .820 | | | .635 | | | | .220 | -.180 | | | -.054 | .046 | | | |
| β, deg | $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | |
| P.L. | .632 | | | | | .361 | .240 | .216 | | | .163 | | .023 | -.039 | | -.104 | .022 | |
| 57½ | .632 | | | | .570 | .311 | .272 | | | | | | .023 | | | | | |
| 45 | -.017 | | | | | | | | | | | | .017 | -.039 | | | | |
| 30 | | .814 | | .643 | .563 | .422 | | | | | .151 | | -.089 | -.072 | | | | |
| 15 | | | | | .616 | .419 | .322 | | | | .138 | -.059 | -.126 | -.155 | | -.104 | | |
| 10 | | | | | .568 | | | | | | | | | | | | | |
| 7 | | | | | | .440 | | | | | | | | | | | | |
| 3 | | | | | | .571 | .276 | | | .020 | -.148 | -.205 | -.204 | -.168 | -.032 | | | |
| 0 | | | .818 | | .626 | | | | | .231 | -.200 | | | -.156 | | .022 | | |
| -3 | | | | | .531 | | .212 | | -.087 | -.152 | -.236 | -.192 | -.148 | -.023 | | | | |
| -7 | | | | | | .250 | | | | | | | | | | | | |
| -10 | | | | | | .431 | | | | | | | | | | | | |
| -15 | | | | | .454 | .151 | .023 | | -.113 | -.207 | -.226 | -.215 | | -.002 | | | | |
| -30 | | .782 | | .476 | .257 | .186 | | | -.005 | -.220 | -.161 | | | | | | | |
| -45 | .008 | | | | | | | | | -.132 | | -.135 | | | | | | |
| -57½ | .305 | | .321 | | .107 | .073 | | | | -.095 | | | | | | | | |
| -P.L. | .205 | | | | .155 | .063 | .074 | | .030 | -.095 | -.135 | | -.002 | .017 | | | | |

TABLE XI - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED FLAT-WINGED CANOPY - Continued

(a) $M=1.41$

| α/β | .000 | .001 | .004 | .006 | .017 | .048 | .092 | .146 | .215 | .293 | .380 | .460 | .531 | .596 | .650 | .690 | .718 |
|------------------------------------|------|-------|------|------|-------|------|-------|-------|------|------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | |
| $P_{0.1}$ | | .718 | | | | .466 | .348 | .245 | | | .270 | | .093 | .084 | | -.251 | -.090 |
| $57\frac{1}{2}$ | | .718 | | | .671 | .415 | .385 | | | | | | .099 | .084 | | | |
| 45 | | -.018 | | | | | | | | | | | | | | | |
| 30 | | | .615 | .706 | .662 | .521 | | | | | .281 | -.011 | -.006 | | | | |
| 15 | | | | | | .656 | .532 | .460 | | | .236 | -.003 | -.074 | -.112 | | -.251 | |
| 10 | | | | | | | .699 | | | | | | | | | | |
| 7 | | | | | | | | .503 | | | | | | | | | |
| 3 | | | | | | | | .575 | .296 | | .071 | -.140 | -.242 | -.310 | -.343 | -.088 | |
| 0 | | | .827 | | | .607 | | | | .734 | -.525 | | | | -.354 | | -.080 |
| -3 | | | | | | | .512 | | .203 | | -.095 | -.157 | -.380 | -.415 | -.496 | -.104 | |
| -7 | | | | | | | | .198 | | | | | | | | | |
| -10 | | | | | | | .365 | | | | | | | | | | |
| -15 | | | | | | .552 | -.223 | -.293 | | | -.745 | -.513 | -.321 | -.253 | | -.737 | |
| -30 | | | .780 | .118 | -.143 | .056 | | | | | -.115 | | -.269 | -.194 | | | |
| -45 | | | .083 | | | | | | | | | -.194 | | | -.167 | | |
| $-57\frac{1}{2}$ | | | .057 | .130 | | .002 | -.318 | | | | | -.747 | | | | | |
| $-P_{0.1}$ | | | .057 | | | .059 | -.713 | .003 | | | -.035 | | -.147 | -.167 | | -.037 | -.081 |
| $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | |
| $P_{0.1}$ | | .466 | | | | .262 | .152 | .100 | | | .308 | .061 | -.034 | -.122 | | -.093 | .007 |
| $57\frac{1}{2}$ | | .466 | | | .417 | .199 | .148 | | | | | | -.034 | | | | |
| 45 | | .652 | | | | | | .170 | | | | | -.040 | -.122 | | | |
| 30 | | | .683 | .466 | .348 | .261 | | | | | .084 | | -.152 | -.119 | | | |
| 15 | | | | | | .432 | .212 | .142 | | | .003 | -.161 | -.213 | -.206 | | -.093 | |
| 10 | | | | | | | .370 | | | | | | | | | | |
| 7 | | | | | | | | .240 | | | | | | | | | |
| 3 | | | | | | | .405 | | .131 | | -.127 | -.227 | -.216 | -.108 | -.092 | | |
| 0 | | | .694 | | | .483 | | | | .107 | | -.240 | | | -.088 | -.152 | .007 |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | |
| $P_{0.1}$ | | .570 | | | | .340 | .233 | .238 | | | .177 | .124 | .022 | -.074 | | -.130 | .003 |
| $57\frac{1}{2}$ | | .570 | | | .512 | .285 | .259 | | | | | | .022 | | | | |
| 45 | | .667 | | | | | | .251 | | | | | .027 | -.074 | | | |
| 30 | | | .687 | .534 | .474 | .358 | | | | | .158 | | -.099 | -.104 | | | |
| 15 | | | | | | .491 | .326 | .260 | | | .092 | -.117 | -.179 | -.201 | | -.120 | |
| 10 | | | | | | | .477 | | | | | | | | | | |
| 7 | | | | | | | | .321 | | | | | | | | | |
| 3 | | | | | | | .400 | | .162 | | -.076 | -.223 | -.277 | -.216 | -.138 | | |
| 0 | | | .690 | | | .488 | | | | .115 | -.273 | | | -.116 | -.038 | .003 | |
| -3 | | | | | | | .383 | | .103 | | -.171 | -.293 | -.268 | -.212 | -.119 | | |
| -7 | | | | | | | | .142 | | | | | | | | | |
| -10 | | | | | | | .297 | | | | | | | | | | |
| -15 | | | | | | .343 | .085 | -.005 | | | -.113 | -.216 | -.212 | -.211 | | -.026 | |
| -30 | | | .668 | .391 | .225 | .152 | | | | | .006 | | -.214 | -.172 | | | |
| -45 | | | .619 | | | | | .090 | | | | -.102 | | -.159 | | | |
| $-57\frac{1}{2}$ | | | .313 | .313 | | .122 | .086 | | | | | | -.083 | | | | |
| $-P_{0.1}$ | | | .313 | | | .176 | .077 | .088 | | | .142 | .000 | -.023 | -.159 | | -.026 | -.011 |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | |
| $P_{0.1}$ | | .636 | | | | .420 | .324 | .321 | | | .258 | .196 | .085 | -.071 | | -.209 | -.046 |
| $57\frac{1}{2}$ | | .615 | | | .592 | .378 | .354 | | | | | | .085 | | | | |
| 45 | | .656 | | | | | | .316 | | | | | .098 | -.017 | | | |
| 30 | | | .680 | .579 | .555 | .441 | | | | | .237 | -.023 | -.052 | | | | |
| 15 | | | | | | .518 | .420 | .355 | | | .169 | -.080 | -.140 | -.159 | | -.209 | |
| 10 | | | | | | | .460 | | | | | | | | | | |
| 7 | | | | | | | | .369 | | | | | | | | | |
| 3 | | | | | | | .427 | | .179 | | -.027 | -.212 | -.320 | -.406 | -.257 | | |
| 0 | | | .696 | | | .468 | | | | .120 | | -.420 | | -.232 | -.015 | -.046 | |
| -3 | | | | | | | .367 | | .091 | | -.180 | -.349 | -.439 | -.314 | -.212 | | |
| -7 | | | | | | | | .083 | | | | | | | | | |
| -10 | | | | | | | .213 | | | | | | | | | | |
| -15 | | | | | | .219 | -.066 | -.111 | | | -.142 | -.285 | -.252 | -.237 | | -.056 | |
| -30 | | | .665 | .420 | -.039 | .018 | | | | | -.077 | | -.758 | -.192 | | | |
| -45 | | | .589 | | | | | .011 | | | | -.153 | | -.191 | | | |
| $-57\frac{1}{2}$ | | | .125 | .195 | | .085 | .015 | | | | | | -.127 | | | | |
| $-P_{0.1}$ | | | .125 | | | .088 | .015 | .006 | | | -.074 | -.049 | -.127 | -.191 | | -.056 | -.071 |

TABLE XC - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED FLAT-WINGED CANYON - Continued

(b) $M=2.01$

| α/β | .000 | .001 | .004 | .006 | .017 | .046 | .092 | .164 | .226 | .236 | .240 | .260 | .311 | .356 | .440 | .600 | .828 | .968 |
|------------------------------------|------|------|------|------|-------|------|------|-------|-------|------|------|------|-------|-------|-------|-------|-------|-------|
| $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | | |
| P_{uL} | | .380 | | | | | .221 | | .112 | | | | .115 | .093 | .039 | .002 | -.012 | .045 |
| $57\frac{1}{2}$ | | .380 | | | | | .211 | .157 | | | | | | .039 | | | | |
| 45 | .771 | | | | .419 | | | | .166 | | | | | .022 | | .002 | | |
| 30 | | | .878 | | .624 | .415 | .331 | | | | | | .125 | -.029 | -.047 | | | |
| 15 | | | | | | | .565 | .340 | .233 | | | | .116 | -.007 | -.051 | -.075 | -.012 | |
| 10 | | | | | | | | .514 | | | | | | | | | | |
| 7 | | | | | | | | | .435 | | | | | | | | | |
| 3 | | | | | | | | .603 | .365 | | | | .118 | -.056 | -.081 | -.065 | -.045 | .015 |
| 0 | | | | .908 | | | .657 | | | .349 | | | | -.056 | | -.061 | .015 | .045 |
| $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | | |
| P_{uL} | | .430 | | | | | .319 | | .216 | | | | .165 | .156 | .097 | .051 | -.096 | .021 |
| $57\frac{1}{2}$ | | .420 | | | | .512 | .306 | .248 | | | | | | | .097 | | | |
| 45 | .725 | | | | | | | | .253 | | | | | .087 | | .051 | | |
| 30 | | | .652 | | .680 | .549 | .440 | | | | | | .218 | .022 | .002 | | | |
| 15 | | | | | | | .638 | .433 | .336 | | | | .200 | .049 | -.003 | -.048 | -.096 | |
| 10 | | | | | | | | .594 | | | | | | | | | | |
| 7 | | | | | | | | | .494 | | | | | | | | | |
| 3 | | | | | | | | .609 | .382 | | | | .144 | -.026 | -.044 | -.064 | -.101 | -.027 |
| 0 | | | .692 | | | .652 | | | | .349 | | | | -.075 | | -.099 | -.014 | .021 |
| -3 | | | | | | | .591 | | .347 | | | | .095 | -.004 | -.114 | -.090 | -.096 | -.010 |
| -7 | | | | | | | | .379 | | | | | | | | | | |
| -10 | | | | | | | | .490 | | | | | | | | | | |
| -15 | | | | | | | .485 | .255 | .135 | | | | .023 | -.073 | -.109 | -.115 | | .029 |
| -30 | | | .874 | | .574 | .173 | .228 | | | | | | .038 | | -.086 | -.096 | | |
| -45 | .778 | | | | | | | | .081 | | | | | -.031 | | -.044 | | |
| -57 $\frac{1}{2}$ | | .237 | | | .311 | | .118 | .075 | | | | | | -.010 | | | | |
| - P_{uL} | | .237 | | | | | .129 | .074 | | | | | .051 | .033 | -.010 | -.061 | .019 | .012 |
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | | |
| P_{uL} | | .579 | | | | | .420 | | .299 | | | | .263 | .226 | .158 | .101 | -.122 | -.076 |
| $57\frac{1}{2}$ | | .579 | | | | .580 | .399 | .341 | | | | | | | .158 | | | |
| 45 | .652 | | | | | | | | .347 | | | | | .153 | | .101 | | |
| 30 | | | .784 | | .716 | .647 | .537 | | | | | | .306 | .075 | .051 | | | |
| 15 | | | | | | | .696 | .519 | .430 | | | | .285 | -.096 | .040 | -.017 | -.122 | |
| 10 | | | | | | | | .629 | | | | | | | | | | |
| 7 | | | | | | | | | .539 | | | | | | | | | |
| 3 | | | | | | | | .612 | .393 | | | | .167 | .001 | -.080 | -.139 | -.194 | -.096 |
| 0 | | | .864 | | | .658 | | | | .312 | | | | -.069 | | -.195 | -.094 | -.076 |
| -3 | | | | | | | .536 | | .341 | | | | .387 | -.107 | -.173 | -.150 | -.179 | -.115 |
| -7 | | | | | | | | .350 | | | | | | | | | | |
| -10 | | | | | | | | .455 | | | | | | | | | | |
| -15 | | | | | | | .446 | -.035 | .050 | | | | -.018 | -.121 | -.158 | -.158 | | .040 |
| -30 | | .869 | | .562 | -.037 | .152 | | | | | | | -.051 | | -.139 | -.137 | | |
| -45 | .770 | | | | | | | | -.002 | | | | | -.085 | | -.085 | | |
| -57 $\frac{1}{2}$ | | .078 | | .159 | | .028 | .000 | | | | | | | -.053 | | | | |
| - P_{uL} | | .078 | | | | .047 | | .014 | | | | | -.003 | -.016 | -.053 | -.085 | -.040 | -.095 |
| $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | | |
| P_{uL} | | .337 | | | | | .214 | | .131 | | | | .106 | .081 | .022 | -.027 | -.049 | -.022 |
| $57\frac{1}{2}$ | | .337 | | | | .342 | .166 | .145 | | | | | | | .022 | | | |
| 45 | .614 | | | | | | | | .152 | | | | | .014 | | -.027 | | |
| 30 | | | .667 | | .490 | .341 | .262 | | | | | | .102 | -.050 | -.064 | | | |
| 15 | | | | | | | .421 | .240 | .157 | | | | .059 | -.050 | -.087 | -.114 | -.049 | |
| 10 | | | | | | | | .398 | | | | | | | | | | |
| 7 | | | | | | | | | .305 | | | | | | | | | |
| 3 | | | | | | | | .405 | .232 | | | | .085 | -.121 | -.138 | -.105 | -.076 | -.024 |
| 0 | | | .697 | | | .487 | | | | .215 | | | | -.126 | | -.071 | -.023 | -.022 |

TABLE VI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED FLAT-WINDSHIELD CANOPY - Concluded

(b) $M=2.0$

| x/l ϕ, deg | .000 | .001 | .004 | .006 | .017 | .035 | .092 | .164 | .216 | .238 | .240 | .260 | .311 | .356 | .440 | .600 | .823 | .988 |
|--|------|------|------|------|------|-------|------|-------|-------|------|------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha = 6.5^\circ; \beta = -4^\circ$ | | | | | | | | | | | | | | | | | | |
| P_{oL} | .420 | | | | | | .297 | | .176 | | | .168 | .138 | .068 | .011 | | -.164 | -.060 |
| $57\frac{1}{2}$ | .420 | | | | | | .262 | .217 | | | | | | .068 | | | | |
| 45 | .562 | | | | | | | | .224 | | | | .069 | | .011 | | | |
| 30 | | | .555 | | .528 | .440 | .350 | | | | | .170 | | -.008 | -.029 | | | |
| 15 | | | | | | | .185 | .322 | .247 | | | .137 | -.011 | -.054 | -.096 | | -.164 | |
| 10 | | | | | | | | .439 | | | | | | | | | | |
| 7 | | | | | | | | | .357 | | | | | | | | | |
| 3 | | | | | | | | .438 | | .244 | | .045 | -.096 | -.145 | -.153 | -.131 | -.073 | |
| 0 | | | | .681 | | | .479 | | | .211 | | | -.137 | | | -.108 | -.105 | -.060 |
| -3 | | | | | | | | .424 | | .215 | | .004 | -.155 | -.167 | -.130 | -.101 | -.093 | |
| -7 | | | | | | | | | .241 | | | | | | | | | |
| -10 | | | | | | | | .339 | | | | | | | | | | |
| -15 | | | | | | | .346 | .155 | .058 | | | -.027 | -.111 | -.137 | -.133 | | -.020 | |
| -30 | | | .666 | | .440 | .151 | .164 | | | | | .022 | | -.099 | -.106 | | | |
| -45 | .616 | | | | | | | | .073 | | | | -.039 | | -.068 | | | |
| $-57\frac{1}{2}$ | | .221 | | | | | .102 | .067 | | | | | | -.024 | | | | |
| $-P_{oL}$ | .221 | | | | | | .134 | | .071 | | | .045 | .025 | -.023 | -.068 | -.020 | -.091 | |
| $\alpha = 6.5^\circ; \beta = -8^\circ$ | | | | | | | | | | | | | | | | | | |
| P_{oL} | .500 | | | | | | .386 | | .278 | | | .243 | .210 | .135 | .066 | | -.163 | -.076 |
| $57\frac{1}{2}$ | .500 | | | | | | .351 | .305 | | | | | | .135 | | | | |
| 45 | .499 | | | | | | | | .314 | | | | .139 | | .066 | | | |
| 30 | | | .583 | | .550 | .522 | .441 | | | | | .254 | | .054 | .026 | | | |
| 15 | | | | | | | .536 | .405 | .338 | | | .216 | .034 | -.018 | -.059 | | -.163 | |
| 10 | | | | | | | | .464 | | | | | | | | | | |
| 7 | | | | | | | | | .394 | | | | | | | | | |
| 3 | | | | | | | | .450 | | .263 | | .077 | -.062 | -.133 | -.190 | -.202 | -.113 | |
| 0 | | | | .662 | | | .494 | | | .214 | | | -.125 | | | -.195 | -.109 | -.076 |
| -3 | | | | | | | | .419 | | .209 | | -.001 | -.158 | -.221 | -.207 | -.198 | -.168 | |
| -7 | | | | | | | | | .210 | | | | | | | | | |
| -10 | | | | | | | | .297 | | | | | | | | | | |
| -15 | | | | | | | .300 | -.097 | -.052 | | | -.082 | -.163 | -.186 | -.160 | | -.060 | |
| -30 | | | .666 | | .422 | -.064 | .083 | | | | | -.054 | | -.135 | -.134 | | | |
| -45 | .570 | | | | | | | | .004 | | | | -.083 | | -.100 | | | |
| $-57\frac{1}{2}$ | | .071 | | | .147 | | .027 | .006 | | | | | | -.063 | | | | |
| $-P_{oL}$ | .071 | | | | | | .057 | | .017 | | | -.004 | -.022 | -.063 | -.120 | -.060 | -.083 | |

TABLE A11.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WINDFIELD CANOPY

(a) $M=1.41$

| x/λ | .008 | .014 | .076 | .124 | .140 | .164 | .212 | .244 | .260 | .295 | .311 | .324 | .345 | .356 | .440 | .600 | .822 | .988 |
|----------------|------------------------------------|------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|------|
| θ , deg | $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | |
| P.L. | .463 | .404 | .380 | .390 | .376 | .367 | .217 | .181 | .061 | | | | | .038 | -.073 | -.012 | -.013 | .069 |
| 57½ | | | | | | .367 | | | | | | | | .038 | | | | |
| 51 | | | | | | .408 | .228 | | | | | | | | | | | |
| 48 | | | | | | | .250 | | | | | | | | | | | |
| 45 | | | | .390 | | | | .173 | -.071 | | | | | | -.073 | | | |
| 41 | | | | | | | | .258 | | | | | | | | | | |
| 34 | | | | | | | | | | -.231 | | | | | | | | |
| 30 | | | .380 | | | | .368 | | | .124 | -.222 | | | -.266 | -.227 | -.012 | | |
| 15 | | .404 | | .444 | | .425 | | .426 | | .318 | | -.195 | -.346 | -.295 | -.064 | -.043 | | |
| 9 | | | | | | | | | | .299 | | | | | | | | |
| 3 | .463 | .456 | .450 | | | .493 | | .471 | .186 | .114 | | | | -.064 | -.186 | -.165 | -.022 | .069 |
| 1 | | | | | | | .513 | | | | | | | | | | | |
| 0 | | | | | | | | | | | | | | | | -.155 | -.020 | |
| θ , deg | $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | |
| P.L. | .548 | .190 | .487 | .490 | .469 | .453 | .303 | .258 | .133 | | | | | .105 | -.014 | .003 | -.149 | .043 |
| 57½ | | | | | | .453 | | | | | | | | .105 | | | | |
| 51 | | | | | | .501 | .326 | | | | | | | | | | | |
| 48 | | | | | | | .349 | | | | | | | | | | | |
| 45 | | | | .490 | | | | .289 | .282 | .006 | | | | | -.014 | | | |
| 41 | | | | | | | | | .356 | | | | | | | | | |
| 34 | | | | | | | | | | -.130 | | | | | | | | |
| 30 | | | .487 | | | .446 | | | | .167 | -.226 | | | -.150 | -.161 | .003 | | |
| 15 | | .490 | | .527 | | .519 | | .491 | | .374 | | -.174 | -.329 | -.206 | -.050 | -.149 | | |
| 9 | | | | | | | | | | .346 | | | | | | | | |
| 3 | .548 | .526 | .527 | | | .575 | | .488 | .085 | .028 | | | | -.093 | -.200 | -.210 | -.060 | .043 |
| 1 | | | | | | | .559 | | | | | | | | | | | |
| 0 | | | | | | | | | | | | | | | | -.240 | -.050 | |
| -1 | | | | | | | .353 | | | | | | | | | | | |
| -3 | .345 | .351 | .344 | | | .383 | | .383 | .205 | .121 | | | | -.076 | -.199 | -.245 | -.043 | .033 |
| -9 | | | | | | | | | | .243 | | | | | | | | |
| -15 | | .307 | | .338 | | .324 | | .345 | | .256 | | -.226 | -.366 | -.335 | -.034 | -.001 | | |
| -30 | | | .271 | | | | .282 | | | .070 | -.316 | | -.367 | -.288 | -.034 | | | |
| -34 | | | | | | | | | | -.313 | | | | | | | | |
| -41 | | | | | | | | .158 | | | | | | | | | | |
| -45 | | | | .264 | | | .084 | .066 | -.150 | | | | | | -.130 | | | |
| -48 | | | | | | | .143 | | | | | | | | | | | |
| -51 | | | | | | .276 | .123 | | | | | | | | | | | |
| -57½ | | | | | | .254 | | | | | | | | | -.026 | | | |
| -P.L. | .345 | .307 | .271 | .264 | .252 | .254 | .126 | .095 | -.009 | | | | | -.026 | -.130 | -.034 | -.001 | .033 |

TABLE XII. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WING-CHINLED CANNY - Continued

| (a) $M=1.41$ | | | | | | | | | | | | | | | | | | | |
|------------------------------------|----------------|------|-------|-------|-------|------|-------|-------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|------|
| α/β | α/β | .008 | .044 | .076 | .124 | .160 | .164 | .212 | .244 | .260 | .295 | .311 | .324 | .345 | .356 | .440 | .600 | .828 | .968 |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | | | |
| F.L. | | .416 | .399 | .406 | .412 | .418 | .405 | .398 | .244 | .125 | | | | | .099 | -.019 | -.113 | -.167 | |
| 57 1/2 | | | | | | | .405 | | | | | | | | .099 | | | | |
| 51 | | | | | | | .441 | .315 | | | | | | | | | | | |
| 45 | | | | | | | | .330 | | | | | | | | | | | |
| 45 | | | | | .412 | | | .295 | .274 | .021 | | | | | | -.019 | | | |
| 41 | | | | | | | | | .330 | | | | | | | | | | |
| 34 | | | | | | | | | | -.061 | | | | | | | | | |
| 30 | | | | .406 | | | | .368 | | .105 | -.162 | | | | -.127 | -.142 | -.113 | | |
| 15 | | | .399 | | .443 | | | | .401 | .313 | | | -.202 | -.154 | -.266 | -.122 | -.167 | | |
| 9 | | | | | | | | | | .261 | | | | | | | | | |
| 3 | | .416 | .421 | .405 | .428 | | .450 | | .354 | -.046 | -.087 | | | | -.171 | -.264 | -.242 | -.085 | |
| 1 | | | | | | | | .471 | | | | | | | | | | | |
| 0 | | | | | | | | | | | | | | | | | -.178 | -.094 | |
| -2 | | | | | | | | .166 | | | | | | | | | | | |
| -3 | | .223 | .246 | .234 | .259 | | .277 | | .227 | .077 | .019 | | | | -.147 | -.262 | -.141 | -.095 | |
| -9 | | | | | | | | | | | .174 | | | | | | | | |
| -15 | | | .221 | | .269 | | | | .278 | | .210 | | | -.250 | -.389 | -.355 | -.069 | -.043 | |
| -30 | | | | .211 | | | | .240 | | | .017 | -.322 | | | -.294 | -.252 | -.084 | | |
| -34 | | | | | | | | | | | -.253 | | | | | | | | |
| -41 | | | | | | | | | .290 | | | | | | | | | | |
| -45 | | | | | .238 | | | | .111 | .101 | -.113 | | | | | -.119 | | | |
| -48 | | | | | | | | .158 | | | | | | | | | | | |
| -51 | | | | | | | .283 | .143 | | | | | | | | | | | |
| -57 1/2 | | | | | | | .249 | | | | | | | | | -.026 | | | |
| -P.L. | | .223 | .221 | .211 | .238 | .240 | .249 | .146 | .111 | .006 | | | | | -.076 | -.119 | -.084 | -.043 | |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | | | |
| F.L. | | .163 | .187 | .455 | .405 | .455 | .491 | .384 | .321 | .194 | | | | | .162 | .034 | -.119 | -.285 | |
| 57 1/2 | | | | | | | .191 | | | | | | | | .162 | | | | |
| 51 | | | | | | | .519 | .404 | | | | | | | | | | | |
| 45 | | | | | | | | .416 | | | | | | | | | | | |
| 45 | | | | | .495 | | | | .390 | .362 | .113 | | | | | .024 | | | |
| 41 | | | | | | | | | | .416 | | | | | | | | | |
| 34 | | | | | | | | | | | .032 | | | | | | | | |
| 30 | | | | .455 | | | | .485 | | | .354 | -.065 | | | -.014 | -.083 | -.119 | | |
| 15 | | | .487 | | .525 | | | | .467 | .351 | | | -.286 | -.342 | -.267 | -.135 | -.285 | | |
| 9 | | | | | | | | | | .291 | | | | | | | | | |
| 3 | | .463 | .163 | .438 | .463 | | .479 | | .325 | -.194 | -.218 | | | | -.283 | -.263 | -.366 | -.137 | |
| 1 | | | | | | | | .410 | | | | | | | | | | | |
| 0 | | | | | | | | | | | | | | | | | -.241 | -.157 | |
| -2 | | | | | | | | -.036 | | | | | | | | | | | |
| -3 | | .063 | -.049 | -.116 | -.065 | | -.092 | | .047 | -.002 | -.037 | | | | -.200 | -.294 | -.254 | -.247 | |
| -9 | | | | | | | | | | | .158 | | | | | | | | |
| -15 | | | .118 | | .180 | | | | .205 | .158 | | | -.265 | -.106 | -.392 | -.092 | -.090 | | |
| -30 | | | | .105 | | | | .166 | | | -.023 | -.259 | | -.384 | -.298 | -.091 | | | |
| -34 | | | | | | | | | | | -.218 | | | | | | | | |
| -41 | | | | | | | | | .067 | | | | | | | | | | |
| -45 | | | | | .132 | | | .026 | .014 | -.177 | | | | | -.166 | | | | |
| -48 | | | | | | | | .071 | | | | | | | | | | | |
| -51 | | | | | | | .182 | .060 | | | | | | | | | | | |
| -57 1/2 | | | | | | | .164 | | | | | | | | | -.073 | | | |
| -P.L. | | .063 | .118 | .105 | .132 | .139 | .164 | .073 | .050 | -.052 | | | | | -.073 | -.166 | -.091 | -.090 | |

TABLE VII. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WING SHIELD CANOPY - Continued

(b) $M=2.01$

| α/γ β, deg | .008 | .044 | .076 | .124 | .140 | .164 | .212 | .244 | .260 | .295 | .311 | .324 | .345 | .356 | .440 | .600 | .828 | .988 |
|--|------|------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | | |
| P.L. | .398 | .379 | .344 | .312 | .297 | .191 | .163 | .077 | | | | | | .065 | .018 | -.028 | -.049 | .063 |
| 57 $\frac{1}{2}$ | | | | | .297 | | | | | | | | | .065 | | | | |
| 51 | | | | | .335 | .201 | | | | | | | | | | | | |
| 45 | | | | | .227 | | | | | | | | | | | | | |
| 45 | | | | | | | .325 | .167 | | -.003 | | | | | .018 | | | |
| 41 | | | | | | | .253 | | | | | | | | | | | |
| 34 | | | | | | | | | | | -.056 | | | | | | | |
| 30 | | | .344 | | | .354 | | | | .217 | -.057 | | | -.115 | -.105 | -.028 | | |
| 15 | | .379 | | .399 | | .374 | | .418 | | .395 | | .050 | -.072 | -.162 | -.020 | -.049 | | |
| 9 | | | | | | | | | | .380 | | | | | | | | |
| 3 | .398 | .421 | .439 | .455 | | .469 | | .465 | .226 | .163 | | | | .048 | -.015 | -.115 | .001 | .063 |
| 1 | | | | | | .492 | | | | | | | | | | | | |
| 0 | | | | | | | | | | | | | | | | -.113 | .006 | |
| $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | | |
| P.L. | .507 | .474 | .445 | .413 | .397 | .281 | .243 | .149 | | | | | | .133 | .072 | -.021 | -.079 | .034 |
| 57 $\frac{1}{2}$ | | | | | .397 | | | | | | | | | .133 | | | | |
| 51 | | | | | .439 | .300 | | | | | | | | | | | | |
| 45 | | | | | .327 | | | | | | | | | | | | | |
| 45 | | | | | | .424 | .253 | | .068 | | | | | .072 | | | | |
| 41 | | | | | | .345 | | | | | | | | | | | | |
| 34 | | | | | | | | | .004 | | | | | | | | | |
| 30 | | | .145 | | | .460 | | | .277 | -.023 | | | | -.047 | -.046 | -.021 | | |
| 15 | | .474 | | .492 | | .403 | | .490 | .160 | | .061 | -.050 | -.145 | -.005 | -.079 | | | |
| 9 | | | | | | | | | .447 | | | | | | | | | |
| 3 | .507 | .503 | .503 | .524 | | .540 | | .494 | .157 | .103 | | | | .003 | -.042 | -.096 | -.078 | .034 |
| 1 | | | | | | .544 | | | | | | | | | | | | |
| 0 | | | | | | | | | | | | | | | | -.108 | -.047 | |
| -1 | | | | | | .325 | | | | | | | | | | | | |
| -3 | .271 | .328 | .344 | .361 | | .375 | | .397 | .235 | .170 | | | | .034 | -.024 | -.130 | -.036 | -.022 |
| -9 | | | | | | | | | .310 | | | | | | | | | |
| -15 | | .279 | | .283 | | .286 | | .332 | .322 | | .009 | -.107 | -.158 | -.100 | .009 | | | |
| -30 | | | .246 | | | .244 | | | .159 | -.095 | | -.174 | -.173 | -.025 | | | | |
| -34 | | | | | | | | | -.098 | | | | | | | | | |
| -42 | | | | | | | | .168 | | | | | | | | | | |
| -45 | | | | | | | .217 | .068 | | -.075 | | | | -.039 | | | | |
| -48 | | | | | | .137 | | | | | | | | | | | | |
| -52 | | | | | .255 | .108 | | | | | | | | | | | | |
| -57 $\frac{1}{2}$ | | | | | .194 | | | | | | | | | .001 | | | | |
| -F.L. | .271 | .279 | .246 | .287 | .194 | .302 | | .085 | | .030 | | | | .001 | -.039 | -.025 | .009 | -.022 |

TABLE XII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEE-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

| α/β | .008 | .014 | .076 | .124 | .140 | .164 | .212 | .244 | .260 | .295 | .311 | .324 | .345 | .356 | .440 | .600 | .828 | .988 |
|-----------------------------|------------------------------------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|
| β, deg | $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | |
| P.L. | .590 | .577 | .554 | .512 | .498 | .377 | .333 | .229 | | | | | | .207 | .136 | .023 | -.089 | -.050 |
| 57 1/2 | | | | | .498 | | | | | | | | | .207 | | | | |
| 51 | | | | | .553 | .407 | | | | | | | | | | | | |
| 45 | | | | | | .433 | | | | | | | | | | | | |
| 45 | | | | | | | .527 | .361 | | .118 | | | | | .136 | | | |
| 41 | | | | | | | | .446 | | | | | | | | | | |
| 34 | | | | | | | | | | .089 | | | | | | | | |
| 30 | | | .554 | | | .531 | | | | .342 | .024 | | | .036 | .023 | .023 | | |
| 15 | | .577 | | .581 | .549 | | .567 | .533 | | .119 | -.023 | -.127 | -.005 | -.089 | | | | |
| 9 | | | | | | | | .507 | | | | | | | | | | |
| 3 | .590 | .566 | .552 | .581 | | .599 | | .513 | .051 | -.006 | | | | -.079 | -.108 | -.116 | -.142 | -.050 |
| 1 | | | | | | .572 | | | | | | | | | | | | |
| 0 | | | | | | | | | | | | | | | | -.134 | -.121 | |
| -1 | | | | | | .077 | | | | | | | | | | | | |
| -3 | .159 | .210 | .140 | .256 | | .220 | | .168 | .158 | .114 | | | | -.001 | -.062 | -.139 | -.123 | -.049 |
| -9 | | | | | | | | | | .248 | | | | | | | | |
| -15 | | .192 | | .220 | .204 | | .242 | .246 | | -.027 | -.132 | -.146 | -.185 | -.046 | | | | |
| -30 | | | .168 | | | .182 | | | .105 | -.117 | | -.202 | -.211 | -.054 | | | | |
| -34 | | | | | | | | | | -.132 | | | | | | | | |
| -41 | | | | | | | | .095 | | | | | | | | | | |
| -45 | | | | | | | .121 | -.018 | | -.132 | | | | | -.081 | | | |
| -45 | | | | | | .051 | | | | | | | | | | | | |
| -51 | | | | | .126 | .022 | | | | | | | | | | | | |
| -57 1/2 | | | | | .099 | | | | | | | | | -.053 | | | | |
| -P.L. | .159 | .192 | .168 | .108 | .099 | .026 | .016 | | -.016 | | | | | -.053 | -.084 | -.054 | -.046 | -.049 |
| $\alpha=6.5^\circ; \beta=0$ | | | | | | | | | | | | | | | | | | |
| P.L. | .277 | .281 | .267 | .165 | .270 | .194 | .165 | .079 | | | | | | .055 | -.007 | -.081 | -.082 | .015 |
| 57 1/2 | | | | | .270 | | | | | | | | | .055 | | | | |
| 51 | | | | | .301 | .202 | | | | | | | | | | | | |
| 45 | | | | | | .212 | | | | | | | | | | | | |
| 45 | | | | | | | .268 | .169 | | .007 | | | | | -.007 | | | |
| 41 | | | | | | | | .224 | | | | | | | | | | |
| 34 | | | | | | | | | | -.050 | | | | | | | | |
| 30 | | | .267 | | | .291 | | | .157 | -.082 | | | | -.099 | -.086 | -.081 | | |
| 15 | | .281 | | .309 | .290 | | .319 | .296 | | .004 | -.109 | -.135 | -.051 | -.082 | | | | |
| 9 | | | | | | | | .280 | | | | | | | | | | |
| 3 | .277 | .315 | .219 | .334 | .343 | | .333 | .110 | .057 | | | | | -.024 | -.071 | -.126 | -.033 | .015 |
| 1 | | | | | | .354 | | | | | | | | | | | | |
| 0 | | | | | | | | | | | | | | | | -.129 | -.027 | |

TABLE XCI. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED VEH-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

| α/β # deg | .008 | .044 | .076 | .124 | .160 | .164 | .212 | .244 | .260 | .295 | .311 | .324 | .345 | .356 | .440 | .600 | .828 | .988 |
|------------------------------------|------|------|-------|------|------|------|-------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | | |
| $P_{o, L}$ | .367 | .368 | .357 | | .349 | .353 | .270 | | .234 | | .144 | | | .116 | .043 | -.044 | -.134 | -.007 |
| 57½ | | | | | | .353 | | | | | | | | .116 | | | | |
| 51 | | | | | | .391 | .260 | | | | | | | | | | | |
| 45 | | | | | | .305 | | | | | | | | | | | | |
| 41 | | | | | | | | .352 | .248 | | .072 | | | | .043 | | | |
| 34 | | | | | | | | | .305 | | | | | | | | | |
| 30 | | | | | | | | | | | .021 | | | | | | | |
| 15 | | | .357 | | | .365 | | | | .207 | -.040 | | -.035 | -.039 | -.044 | | | |
| 9 | | | .368 | .390 | | .365 | | | .380 | .354 | | .033 | -.087 | -.175 | -.077 | -.134 | | |
| 3 | | | | | | | | | | .378 | | | | | | | | |
| 1 | .367 | .360 | .376 | .390 | | .402 | | | .357 | .032 | -.006 | | | -.104 | -.100 | -.115 | -.078 | -.007 |
| 0 | | | | | | .399 | | | | | | | | | | | | |
| -1 | | | | | | | .208 | | | | | | | | | -.128 | -.076 | |
| -3 | .174 | .222 | .234 | .254 | | .256 | | | .264 | .124 | .076 | | | -.029 | -.077 | -.114 | -.063 | -.081 |
| -9 | | | | | | | | | | .226 | | | | | | | | |
| -15 | | .193 | | .223 | | .216 | | | .249 | .237 | | -.026 | -.128 | -.156 | -.060 | -.022 | | |
| -30 | | | .160 | | | .218 | | | | .107 | -.112 | | -.152 | -.130 | -.089 | | | |
| -34 | | | | | | | | | | -.103 | | | | | | | | |
| -41 | | | | | | | | .150 | | | | | | | | | | |
| -45 | | | | | | | | .167 | .094 | | -.044 | | | | -.044 | | | |
| -49 | | | | | | | | .112 | | | | | | | | | | |
| -51 | | | | | | .219 | .186 | | | | | | | | | | | |
| -57½ | | | | | | .192 | | | | | | | | .008 | | | | |
| -P _{o, L} | .174 | .193 | .180 | | .167 | .192 | .129 | | .107 | | .031 | | | .008 | -.044 | -.089 | -.022 | -.081 |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | | |
| $P_{o, L}$ | .436 | .465 | .460 | | .428 | .435 | .355 | | .313 | .215 | | | | .180 | .100 | .003 | -.154 | -.091 |
| 57½ | | | | | | .425 | | | | | | | | .180 | | | | |
| 51 | | | | | | .477 | .382 | | | | | | | | | | | |
| 45 | | | | | | | .398 | | | | | | | | | | | |
| 41 | | | | | | | | .437 | .345 | | .145 | | | | .100 | | | |
| 34 | | | | | | | | | .396 | | | | | | | | | |
| 30 | | | | | | | | | | | .105 | | | | | | | |
| 15 | | | .460 | | | .450 | | | | .270 | .031 | | .037 | .019 | .003 | | | |
| 9 | | .465 | | .477 | | .446 | | | .452 | .419 | | .069 | -.061 | -.132 | -.083 | -.154 | | |
| 3 | | | | | | | | | | .391 | | | | | | | | |
| 1 | .436 | .446 | .421 | .39 | | .442 | | | .380 | -.055 | -.073 | | | -.143 | -.174 | -.163 | -.138 | -.091 |
| 0 | | | | | | .416 | | | | | | | | | | | | |
| -1 | | | | | | | -.019 | | | | | | | | | -.154 | -.173 | |
| -3 | .057 | .023 | -.042 | .050 | | .007 | | | .046 | .032 | .005 | | | -.039 | -.152 | -.157 | -.211 | -.104 |
| -9 | | | | | | | | | | .179 | | | | | | | | |
| -15 | | .101 | | .110 | | .127 | | | .159 | .170 | | -.062 | -.155 | -.207 | -.112 | -.064 | | |
| -30 | | | .089 | | | .136 | | | | .055 | -.132 | | -.202 | -.181 | -.065 | | | |
| -34 | | | | | | | | | | -.147 | | | | | | | | |
| -41 | | | | | | | | .072 | | | | | | | | | | |
| -45 | | | | | | | .093 | .016 | | -.202 | | | | | -.087 | | | |
| -49 | | | | | | .059 | | | | | | | | | | | | |
| -51 | | | | | | .125 | .045 | | | | | | | | | | | |
| -57½ | | | | | | .107 | | | | | | | | -.047 | | | | |
| -P _{o, L} | .057 | .171 | .039 | | .095 | .107 | .060 | | .044 | | -.026 | | | -.047 | -.097 | -.285 | -.064 | -.104 |

TABLE XIII. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH 10° AND 10°-10° BOUND-WINGFIELD CANOPY

(a) $M=14$

| x/l deg | .002 | .036 | .066 | .124 | .160 | .212 | .260 | .311 | .356 | .410 | .600 | .828 | .968 |
|------------------------------------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | |
| $P_{t/L}$ | .728 | .512 | .427 | .366 | .331 | .275 | .231 | .059 | .030 | -.050 | -.023 | -.116 | .025 |
| $57\frac{1}{2}$ | | | | | .331 | .251 | .026 | .030 | | | | | |
| 45 | | | .427 | | .332 | .295 | .147 | -.020 | -.061 | -.050 | | | |
| 30 | | .512 | .480 | .454 | .378 | .354 | .186 | -.028 | -.119 | -.066 | -.023 | | |
| 15 | | .614 | .576 | .546 | .497 | .426 | .228 | -.056 | -.148 | -.167 | -.053 | -.116 | |
| 3 | | | .631 | | .450 | .220 | -.106 | -.193 | -.211 | -.182 | -.061 | .025 | |
| 0 | .728 | | .602 | | .577 | | .189 | | -.217 | | -.180 | -.047 | .026 |
| -3 | | | .569 | | .369 | .126 | -.154 | -.254 | -.222 | -.197 | -.050 | .003 | |
| -15 | | .424 | .347 | .310 | .257 | .180 | .026 | -.200 | -.253 | -.216 | -.061 | -.031 | |
| -30 | | .243 | .227 | .210 | .148 | .126 | .004 | -.147 | -.244 | -.187 | -.030 | | |
| -45 | | | .190 | | .120 | .103 | -.017 | -.152 | -.239 | -.152 | | | |
| $-57\frac{1}{2}$ | | | | | .131 | | -.008 | -.106 | -.095 | | | | |
| $-P_{t/L}$ | .603 | .243 | .196 | .143 | .131 | .089 | .015 | -.071 | -.095 | -.152 | -.030 | -.031 | .003 |
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | |
| $P_{t/L}$ | .721 | .629 | .540 | .471 | .432 | .374 | .318 | .130 | .101 | .007 | .001 | -.219 | -.038 |
| $57\frac{1}{2}$ | | | | | .432 | | .231 | .099 | .101 | | | | |
| 45 | | | .540 | | .434 | .391 | .226 | .064 | .018 | .007 | | | |
| 30 | | .629 | .596 | .550 | .483 | .449 | .269 | .047 | -.018 | -.001 | .001 | | |
| 15 | | .670 | .642 | .619 | .583 | .513 | .302 | .007 | -.102 | -.124 | -.057 | -.219 | |
| 3 | | | .602 | | .451 | .222 | -.113 | -.205 | -.273 | -.311 | -.119 | -.098 | |
| 0 | .721 | | .539 | | .527 | | .148 | | -.265 | | -.335 | -.114 | -.017 |
| -3 | | | .479 | | .280 | .044 | -.220 | -.312 | -.326 | -.341 | -.159 | -.037 | |
| -15 | | .297 | .192 | .146 | .107 | .033 | -.101 | -.290 | -.320 | -.287 | -.104 | -.047 | |
| -30 | | .120 | .073 | .063 | .021 | .003 | -.089 | -.237 | -.303 | -.250 | -.042 | | |
| -45 | | | .089 | | .017 | .008 | -.094 | -.214 | -.257 | -.190 | | | |
| $-57\frac{1}{2}$ | | | | | .037 | | -.077 | -.159 | -.146 | | | | |
| $-P_{t/L}$ | .627 | .100 | .069 | .045 | .037 | .013 | -.049 | -.124 | -.146 | -.190 | -.042 | -.047 | -.037 |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | |
| $P_{t/L}$ | .603 | .441 | .349 | .259 | .221 | .273 | .085 | .059 | .024 | -.088 | -.099 | -.176 | -.005 |
| $57\frac{1}{2}$ | | | | | .221 | | .004 | .030 | .024 | | | | |
| 45 | | | .349 | | .210 | .283 | .144 | -.004 | -.042 | -.088 | | | |
| 30 | | .441 | .426 | .397 | .330 | .314 | .158 | -.039 | -.132 | -.096 | -.099 | | |
| 15 | | .499 | .463 | .439 | .401 | .346 | .151 | -.108 | -.203 | -.179 | -.103 | -.176 | |
| 3 | | | .480 | | .375 | .325 | -.145 | -.265 | -.264 | -.181 | -.089 | -.005 | |
| 0 | .603 | | .457 | | .435 | | .071 | | -.279 | | -.115 | -.093 | -.005 |
| -3 | | | .424 | | .262 | .016 | -.219 | -.296 | -.247 | -.139 | -.129 | -.075 | |
| -15 | | .327 | .273 | .238 | .200 | .134 | -.007 | -.225 | -.277 | -.228 | -.054 | -.058 | |
| -30 | | .203 | .198 | .188 | .133 | .127 | .004 | -.166 | -.241 | -.186 | -.071 | | |
| -45 | | | .121 | | .128 | .113 | .000 | -.132 | -.243 | -.173 | | | |
| $-57\frac{1}{2}$ | | | | | .146 | | .015 | -.090 | -.068 | | | | |
| $-P_{t/L}$ | .551 | .203 | .181 | .165 | .111 | .016 | -.042 | -.038 | -.273 | -.071 | -.058 | -.075 | |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | |
| $P_{t/L}$ | .579 | .541 | .451 | .452 | .410 | .360 | .208 | .124 | .083 | -.037 | -.074 | -.210 | -.047 |
| $57\frac{1}{2}$ | | | | | .410 | | .007 | .098 | .083 | | | | |
| 45 | | | .451 | | .396 | .360 | .216 | .041 | .023 | -.037 | | | |
| 30 | | .541 | .525 | .480 | .410 | .396 | .023 | .225 | -.074 | -.044 | -.074 | | |
| 15 | | .541 | .523 | .475 | .468 | .416 | .216 | -.062 | -.168 | -.190 | -.125 | -.210 | |
| 3 | | | .412 | | .362 | .096 | -.207 | -.293 | -.262 | -.294 | -.156 | -.047 | |
| 0 | .579 | | .346 | | .376 | | .025 | | -.344 | | -.390 | -.201 | -.048 |
| -3 | | | .332 | | .184 | -.064 | -.289 | -.373 | -.354 | -.299 | -.294 | -.084 | |
| -15 | | .202 | .129 | .096 | .061 | -.003 | -.271 | -.276 | -.225 | -.241 | -.043 | -.058 | |
| -30 | | .067 | .072 | .074 | .031 | .026 | -.076 | -.222 | -.267 | -.218 | -.055 | | |
| -45 | | | .080 | | .013 | .020 | -.066 | -.144 | -.211 | -.206 | | | |
| $-57\frac{1}{2}$ | | | | | .066 | | -.002 | -.136 | -.133 | | | | |
| $-P_{t/L}$ | .504 | .067 | .080 | .074 | .066 | .012 | .000 | -.108 | -.133 | -.206 | -.055 | -.058 | -.084 |

TABLE XIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

| $\frac{x}{l}$ θ, deg | .002 | .036 | .088 | .124 | .160 | .212 | .260 | .311 | .356 | .440 | .600 | .828 | .988 |
|--|------|------|------|------|------|------|------|-------|-------|-------|-------|-------|------|
| $\alpha = -6.0^\circ; \beta = 0.3^\circ$ | | | | | | | | | | | | | |
| PoLe | .830 | .451 | .329 | .265 | .217 | .150 | .104 | -.036 | -.043 | -.073 | .033 | -.005 | .079 |
| 57 $\frac{1}{2}$ | | | | | .217 | | .159 | -.078 | -.043 | | | | |
| 45 | | | .329 | | .217 | .195 | .041 | -.119 | -.182 | -.073 | | | |
| 30 | | .451 | .392 | .363 | .279 | .248 | .090 | -.101 | -.185 | -.153 | .038 | | |
| 15 | | .644 | .571 | .519 | .461 | .363 | .175 | -.094 | -.161 | -.138 | .018 | -.006 | |
| 3 | | | | .762 | | .581 | .301 | -.020 | -.130 | -.135 | -.155 | .009 | .079 |
| 0 | .830 | | .786 | | .735 | | .324 | | -.119 | | -.127 | .017 | .095 |
| $\alpha = -3.0^\circ; \beta = 0.3^\circ$ | | | | | | | | | | | | | |
| PoLe | .774 | .608 | .312 | .257 | .217 | .157 | .163 | -.029 | -.044 | -.096 | .007 | -.031 | .058 |
| 57 $\frac{1}{2}$ | | | | | .217 | | .158 | -.066 | -.044 | | | | |
| 15 | | | .312 | | .215 | .193 | .045 | -.107 | -.164 | -.096 | | | |
| 30 | | .408 | .366 | .343 | .265 | .241 | .086 | -.106 | -.189 | -.143 | .007 | | |
| 15 | | .582 | .516 | .471 | .414 | .328 | .146 | -.118 | -.185 | -.162 | -.004 | -.031 | |
| 3 | | | | .688 | | .513 | .239 | -.070 | -.172 | -.167 | -.153 | -.011 | .058 |
| 0 | .774 | | .706 | | .661 | | .257 | | -.162 | | -.135 | -.003 | .070 |
| $\alpha = 0^\circ; \beta = 0.3^\circ$ | | | | | | | | | | | | | |
| PoLe | .719 | .375 | .301 | .253 | .219 | .165 | .167 | -.024 | -.043 | -.112 | -.021 | -.047 | .044 |
| 57 $\frac{1}{2}$ | | | | | .219 | | .161 | -.056 | -.043 | | | | |
| 15 | | | .301 | | .214 | .192 | .052 | -.097 | -.147 | -.112 | | | |
| 30 | | .375 | .344 | .325 | .254 | .234 | .083 | -.109 | -.192 | -.144 | -.021 | | |
| 15 | | .529 | .463 | .432 | .381 | .301 | .125 | -.134 | -.205 | -.181 | -.021 | -.047 | |
| 3 | | | | .619 | | .452 | .183 | -.110 | -.204 | -.190 | -.138 | -.024 | .044 |
| 0 | .719 | | .635 | | .592 | | .198 | | -.197 | | -.124 | -.018 | .057 |
| $\alpha = 3.0^\circ; \beta = 0.3^\circ$ | | | | | | | | | | | | | |
| PoLe | .646 | .345 | .289 | .250 | .222 | .173 | .152 | -.021 | -.044 | -.126 | -.053 | -.064 | .029 |
| 57 $\frac{1}{2}$ | | | | | .222 | | .145 | -.049 | -.044 | | | | |
| 45 | | | .249 | | .212 | .192 | .056 | -.088 | -.133 | -.126 | | | |
| 30 | | .345 | .326 | .308 | .240 | .225 | .077 | -.112 | -.154 | -.147 | -.053 | | |
| 15 | | .477 | .417 | .392 | .345 | .272 | .104 | -.151 | -.223 | -.198 | -.043 | -.064 | |
| 3 | | | | .553 | | .393 | .132 | -.149 | -.236 | -.208 | -.122 | -.035 | .029 |
| 0 | .646 | | .569 | | .526 | | .144 | | -.230 | | -.110 | -.031 | .044 |
| $\alpha = 6.0^\circ; \beta = 0.3^\circ$ | | | | | | | | | | | | | |
| PoLe | .609 | .319 | .278 | .246 | .221 | .177 | .154 | -.016 | -.046 | -.142 | -.087 | -.082 | .013 |
| 57 $\frac{1}{2}$ | | | | | .221 | | .148 | -.044 | -.046 | | | | |
| 45 | | | .278 | | .208 | .167 | .057 | -.082 | -.121 | -.142 | | | |
| 30 | | .319 | .307 | .290 | .224 | .213 | .070 | -.116 | -.157 | -.156 | -.087 | | |
| 15 | | .427 | .370 | .350 | .308 | .241 | .079 | -.169 | -.243 | -.213 | -.062 | -.082 | |
| 3 | | | | .427 | | .330 | .082 | -.187 | -.262 | -.223 | -.108 | -.049 | .013 |
| 0 | .609 | | .502 | | .454 | | .090 | | -.260 | | -.100 | -.043 | .035 |

TABLE XIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

| x/l | .002 | .036 | .088 | .124 | .160 | .212 | .260 | .311 | .356 | .440 | .600 | .828 | .988 |
|---------------------|------|------|------|------|------|------|------|------|------|------|------|------|------|
| β, deg | | | | | | | | | | | | | |

$\alpha=90^\circ; \beta=0.3^\circ$

| P_o/L_o | .553 | .297 | .267 | .243 | .220 | .181 | .156 | -.008 | -.045 | -.151 | -.124 | -.098 | -.009 |
|-----------------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|
| $57\frac{1}{2}$ | | | | | .220 | | .151 | -.036 | -.045 | | | | |
| 45 | | | .267 | | .204 | .183 | .057 | -.077 | -.111 | -.151 | | | |
| 30 | | .297 | .291 | .274 | .212 | .202 | .062 | -.118 | -.197 | -.165 | -.124 | | |
| 15 | | .383 | .329 | .315 | .273 | .213 | .056 | -.185 | -.260 | -.222 | -.080 | -.098 | |
| 3 | | | .425 | | .275 | .036 | -.217 | -.284 | -.230 | -.099 | -.062 | -.009 | |
| 0 | .553 | | .438 | | .392 | .042 | | -.283 | | -.091 | -.054 | .009 | |

$\alpha=12.0^\circ; \beta=0.3^\circ$

| P_o/L_o | .496 | .276 | .254 | .237 | .216 | .179 | .159 | -.002 | -.043 | -.152 | -.104 | -.106 | -.037 |
|-----------------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|
| $57\frac{1}{2}$ | | | | | .216 | | .154 | -.025 | -.043 | | | | |
| 45 | | | .254 | | .199 | .178 | .057 | -.071 | -.102 | -.152 | | | |
| 30 | | .276 | .276 | .256 | .198 | .152 | .057 | -.117 | -.156 | -.170 | -.164 | | |
| 15 | | .345 | .294 | .280 | .241 | .189 | .040 | -.154 | -.270 | -.229 | -.094 | -.106 | |
| 3 | | | .367 | | .229 | .000 | -.240 | -.296 | -.230 | -.054 | -.077 | -.037 | |
| 0 | .496 | | .382 | | .334 | .004 | | -.297 | | -.087 | -.068 | -.011 | |

(b) $M=2.01$

| | | | | | | | | | | | | | |
|------------------------------------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|--|
| $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | |
| P_o/L_o | .725 | .466 | .390 | .350 | .305 | .237 | .174 | .100 | .089 | .033 | .020 | .020 | |
| $57\frac{1}{2}$ | | | | | .305 | | | .079 | .089 | | | | |
| 45 | | | .390 | | .302 | .283 | .186 | .066 | .007 | .033 | | | |
| 30 | | .466 | .443 | .418 | .360 | .330 | .232 | .085 | -.007 | .027 | .020 | | |
| 15 | | .603 | .551 | .528 | .493 | .434 | .293 | .063 | -.021 | -.039 | .003 | | |
| 3 | | | .635 | | .555 | .304 | .038 | -.061 | -.090 | -.103 | -.057 | .020 | |
| 0 | .725 | | .612 | | .598 | .279 | | -.072 | | -.102 | -.036 | .014 | |
| -3 | | | .541 | | .426 | .136 | -.036 | -.175 | -.118 | -.192 | -.142 | -.060 | |
| -15 | | .347 | .263 | .216 | .185 | .129 | .032 | -.117 | -.157 | -.150 | -.136 | | |
| -30 | | .413 | .083 | .079 | .050 | .039 | -.016 | -.107 | -.153 | -.158 | -.054 | | |
| -45 | | | .063 | | .032 | .024 | -.035 | -.111 | -.150 | -.135 | | | |
| -57 $\frac{1}{2}$ | | | | | .032 | | | -.088 | -.073 | | | | |
| - P_o/L_o | .628 | .413 | .063 | .056 | .032 | .004 | -.023 | -.061 | -.073 | -.105 | -.054 | -.060 | |
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | |
| P_o/L_o | .709 | .579 | .502 | .466 | .419 | .338 | .261 | .176 | .158 | .090 | .061 | -.073 | |
| $57\frac{1}{2}$ | | | | | .419 | | | .152 | .158 | | | | |
| 45 | | | .502 | | .408 | .383 | .273 | .139 | .079 | .090 | | | |
| 30 | | .579 | .566 | .528 | .465 | .439 | .320 | .155 | .055 | .104 | .061 | | |
| 15 | | .665 | .636 | .614 | .578 | .524 | .377 | .121 | .023 | .001 | .028 | | |
| 3 | | | .633 | | .570 | .322 | .043 | -.058 | -.111 | -.158 | -.123 | -.073 | |
| 0 | .709 | | .576 | | .572 | .264 | | -.090 | | -.181 | -.112 | -.061 | |
| -3 | | | .616 | | .555 | .304 | .037 | -.063 | -.090 | -.102 | -.058 | .020 | |
| -15 | | .605 | .552 | .530 | .494 | .435 | .295 | .063 | -.021 | -.038 | .005 | | |
| -30 | | .467 | .450 | .420 | .364 | .332 | .234 | .087 | -.006 | .029 | .022 | | |
| -45 | | | .391 | | .304 | .284 | .189 | .067 | .009 | .033 | | | |
| -57 $\frac{1}{2}$ | | | | | .309 | | | .060 | .091 | | | | |
| - P_o/L_o | .724 | .467 | .391 | .352 | .309 | .238 | .176 | .103 | .091 | .033 | .022 | .020 | |

TABLE XIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

| α/β | .002 | .036 | .089 | .124 | .160 | .212 | .260 | .311 | .356 | .440 | .600 | .828 | .988 |
|--------------------------------------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | |
| P.L. | .557 | .385 | .340 | .315 | .298 | .235 | .177 | .096 | .072 | .002 | -.033 | -.150 | -.033 |
| 57 $\frac{1}{2}$ | | | | | .298 | | | .076 | .072 | | | | |
| 45 | | | .340 | | .281 | .261 | .173 | .060 | .011 | .002 | | | |
| 30 | | .385 | .384 | .357 | .306 | .284 | .193 | .057 | -.027 | .014 | -.033 | | |
| 15 | | .479 | .434 | .413 | .381 | .332 | .214 | .009 | -.066 | -.082 | -.561 | -.150 | |
| 3 | | | .469 | | .399 | .184 | -.043 | -.122 | -.140 | -.125 | -.078 | -.033 | |
| 0 | .557 | | .455 | | .434 | .159 | -.129 | -.129 | -.116 | -.081 | -.056 | | |
| -3 | | | .425 | | .327 | .115 | -.080 | -.146 | -.140 | -.124 | -.111 | -.115 | |
| -15 | | .328 | .256 | .229 | .204 | .159 | .059 | -.096 | -.142 | -.134 | -.054 | -.035 | |
| -30 | | .181 | .162 | .165 | .124 | .106 | .043 | -.059 | -.117 | -.094 | -.071 | | |
| -45 | | | .148 | | .106 | .100 | .035 | -.052 | -.089 | -.083 | | | |
| -57 $\frac{1}{2}$ | | | | | .119 | | -.034 | -.032 | | | | | |
| -P.L. | .517 | .181 | .148 | .128 | .119 | .086 | .045 | -.016 | -.032 | -.083 | -.071 | -.035 | -.115 |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | |
| P.L. | .530 | .478 | .442 | .423 | .397 | .325 | .255 | .164 | .136 | .050 | .003 | -.148 | -.068 |
| 57 $\frac{1}{2}$ | | | | | .397 | | | .143 | .136 | | | | |
| 45 | | | .442 | | .379 | .346 | .249 | .126 | .075 | .050 | | | |
| 30 | | .478 | .487 | .454 | .395 | .375 | .270 | .116 | .024 | .075 | .003 | | |
| 15 | | .527 | .501 | .482 | .449 | .406 | .279 | .055 | -.033 | -.050 | -.044 | -.148 | |
| 3 | | | .455 | | .405 | .193 | -.042 | -.121 | -.177 | -.226 | -.149 | -.068 | |
| 0 | .530 | | .443 | | .402 | .138 | -.154 | -.154 | -.212 | -.161 | -.086 | | |
| -3 | | | .378 | | .270 | .070 | -.112 | -.180 | -.196 | -.218 | -.160 | -.084 | |
| -15 | | .234 | .169 | .129 | .101 | .056 | -.023 | -.147 | -.179 | -.163 | -.068 | -.046 | |
| -30 | | .087 | .052 | .065 | .043 | .028 | -.024 | -.107 | -.154 | -.130 | -.068 | | |
| -45 | | | .061 | | .035 | .036 | -.019 | -.095 | -.125 | -.112 | | | |
| -57 $\frac{1}{2}$ | | | | | .049 | | -.071 | -.069 | | | | | |
| -P.L. | .460 | .087 | .061 | .054 | .049 | .032 | -.003 | -.053 | -.069 | -.112 | -.068 | -.046 | -.084 |
| $\alpha=-6.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | |
| P.L. | .900 | .445 | .321 | .269 | .219 | .154 | .100 | .046 | .042 | .002 | .045 | -.002 | .077 |
| 57 $\frac{1}{2}$ | | | | | .219 | | .085 | .007 | .012 | | | | |
| 45 | | | .321 | | .234 | .117 | -.002 | -.067 | .002 | | | | |
| 30 | | .445 | .386 | .371 | .313 | .276 | .177 | .033 | -.048 | -.057 | .045 | | |
| 15 | | .679 | .587 | .547 | .512 | .438 | .286 | .047 | -.030 | -.032 | -.027 | -.002 | |
| 3 | | | .608 | | .609 | .396 | .102 | -.010 | -.030 | -.055 | .025 | .077 | |
| 0 | .900 | | .811 | | .801 | .405 | | .002 | -.043 | .030 | .087 | | |
| $\alpha=-3.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | |
| P.L. | .804 | .390 | .291 | .243 | .203 | .148 | .096 | .036 | .033 | -.009 | .020 | -.022 | .060 |
| 57 $\frac{1}{2}$ | | | | | .203 | | | .010 | .033 | | | | |
| 45 | | | .291 | | .210 | .157 | .108 | .001 | -.061 | -.009 | | | |
| 30 | | .390 | .344 | .335 | .278 | .244 | .155 | .025 | -.056 | -.056 | .020 | | |
| 15 | | .600 | .517 | .477 | .446 | .379 | .243 | .022 | -.048 | -.055 | -.016 | -.022 | |
| 3 | | | .710 | | .610 | .337 | .064 | -.038 | -.055 | -.067 | .010 | .060 | |
| 0 | .804 | | .723 | | .698 | .351 | | -.028 | | -.057 | .017 | .069 | |

TABLE XIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED ROUND-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

| x/γ θ, deg | .002 | .036 | .088 | .124 | .160 | .212 | .260 | .311 | .356 | .440 | .600 | .828 | .968 |
|--------------------------------------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | |
| P.L. | .715 | .357 | .281 | .245 | .207 | .156 | .105 | .044 | .033 | -.012 | -.004 | | .041 |
| 57½ | | | | | .207 | | | .022 | .033 | | | | |
| 45 | | | .281 | | .207 | .194 | .112 | .005 | -.047 | -.012 | | | |
| 30 | | .357 | .322 | .316 | .260 | .233 | .150 | .023 | -.054 | -.041 | -.004 | | |
| 15 | | .536 | .462 | .431 | .402 | .345 | .214 | .007 | -.063 | -.071 | -.012 | | |
| 3 | | | | .622 | | .529 | .278 | .025 | -.067 | -.082 | -.086 | -.010 | .041 |
| 0 | .715 | | .628 | | .604 | | .285 | | -.066 | | -.079 | -.005 | .049 |
| $\alpha=3.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | |
| P.L. | .628 | .306 | .247 | .213 | .190 | .145 | .092 | .030 | .018 | -.033 | -.030 | -.045 | .026 |
| 57½ | | | | | .190 | | .069 | .011 | .018 | | | | |
| 45 | | | .247 | | .185 | .175 | .097 | -.003 | -.048 | -.033 | | | |
| 30 | | .306 | .285 | .275 | .225 | .201 | .005 | .126 | -.067 | -.046 | -.030 | | |
| 15 | | .464 | .398 | .368 | .341 | .287 | .169 | -.020 | -.081 | -.087 | -.026 | -.045 | |
| 3 | | | | .536 | | .450 | .216 | -.010 | -.090 | -.092 | -.088 | -.014 | .026 |
| 0 | .628 | | .546 | | .523 | | .226 | | -.085 | | -.081 | -.011 | .034 |
| $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | | |
| P.L. | .555 | .282 | .239 | .215 | .201 | .155 | .104 | .035 | .014 | -.046 | -.064 | -.081 | -.009 |
| 57½ | | | | | .201 | | | .016 | .014 | | | | |
| 45 | | | .239 | | .186 | .175 | .098 | -.002 | -.045 | -.046 | | | |
| 30 | | .282 | .273 | .258 | .212 | .192 | .116 | -.004 | -.076 | -.047 | -.064 | | |
| 15 | | .412 | .351 | .329 | .300 | .249 | .139 | -.042 | -.105 | -.112 | -.057 | -.081 | |
| 3 | | | | .457 | | .373 | .154 | -.056 | -.128 | -.119 | -.098 | -.040 | -.009 |
| 0 | .555 | | .466 | | .438 | | .158 | | -.124 | | -.095 | -.037 | -.001 |
| $\alpha=9.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | |
| P.L. | .491 | .243 | .211 | .196 | .187 | .145 | .098 | .030 | .006 | -.058 | -.081 | -.082 | -.030 |
| 57½ | | | | | .187 | | .075 | .011 | .006 | | | | |
| 45 | | | .211 | | .171 | .157 | .084 | -.011 | -.049 | -.058 | | | |
| 30 | | .243 | .240 | .226 | .183 | .168 | -.019 | .097 | -.086 | -.053 | -.081 | | |
| 15 | | .354 | .296 | .277 | .251 | .207 | .104 | -.064 | -.121 | -.127 | -.070 | -.082 | |
| 3 | | | | .389 | | .311 | .110 | -.082 | -.144 | -.128 | -.090 | -.046 | -.030 |
| 0 | .491 | | .400 | | .377 | | .117 | | -.141 | | -.089 | -.046 | -.028 |
| $\alpha=12.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | |
| P.L. | .431 | .218 | .198 | .191 | .185 | .148 | .102 | .033 | .006 | -.062 | -.097 | -.089 | -.048 |
| 57½ | | | | | .165 | | .078 | .017 | .006 | | | | |
| 45 | | | .198 | | .170 | .152 | .083 | -.008 | -.043 | -.062 | | | |
| 30 | | .218 | .226 | .210 | .170 | .158 | -.025 | .088 | -.089 | -.054 | -.097 | | |
| 15 | | .308 | .260 | .241 | .215 | .177 | .082 | -.078 | -.133 | -.137 | -.090 | -.089 | |
| 3 | | | | .327 | | .255 | .068 | -.109 | -.159 | -.135 | -.084 | -.062 | -.048 |
| 0 | .431 | | .339 | | .316 | | .071 | | -.157 | | -.085 | -.063 | -.046 |

TABLE XI. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED FLAT-WINGED FIELD CANOPY

(a) $M=1.41$

| x/l | .002 | .006 | .035 | .072 | .137 | .181 | .206 | .210 | .228 | .254 | .306 | .305 | .444 | .606 | .662 | .990 |
|-------------------|--|------|-------|-------|-------|------|------|------|-------|-------|-------|-------|-------|-------|-------|------|
| $y/l = 0.2$ | $\alpha = 0.4^\circ; \beta = 0^\circ$ | | | | | | | | | | | | | | | |
| $P_{x,0}$ | .601 | .443 | | .059 | | | | | .027 | .305 | -.082 | -.219 | | -.110 | | .267 |
| $33\frac{1}{2}$ | | | | .059 | .041 | | | | | -.090 | -.119 | | | | | |
| $22\frac{1}{2}$ | | .443 | .291 | | .047 | | | | | -.265 | -.241 | -.175 | -.110 | | | |
| $13\frac{1}{2}$ | | .441 | | | | | | | | | | | | | | |
| 12 | .704 | | | .211 | .122 | | | | .003 | -.213 | | -.198 | -.141 | | | |
| 9 | | .594 | | | | | | | | | | | | | | |
| $7\frac{1}{2}$ | | | .553 | | | | | | | | | | | | | |
| 6 | | | | .525 | | | | | | | | | | | | |
| 5 | | | | | .233 | | | | | | | | | | | |
| 3 | .789 | | | .519 | .440 | | | | -.254 | -.186 | -.177 | -.228 | -.261 | -.128 | -.057 | .067 |
| 0 | | .611 | | | | | | .209 | -.205 | -.226 | -.226 | -.215 | | | | |
| $P_{x,0}$ | $\alpha = 0.4^\circ; \beta = -4^\circ$ | | | | | | | | | | | | | | | |
| $P_{x,0}$ | .720 | .556 | | .159 | | | | | .111 | .086 | -.016 | -.058 | | -.119 | | .053 |
| $33\frac{1}{2}$ | | | | .159 | .149 | | | | | -.011 | -.058 | | | | | |
| $22\frac{1}{2}$ | | .556 | .400 | | .171 | | | | | .031 | -.173 | -.138 | -.119 | | | |
| $13\frac{1}{2}$ | | .596 | | | | | | | | | | | | | | |
| 12 | .726 | | | .334 | .245 | | | | .114 | -.158 | | -.126 | -.194 | | | |
| 9 | | .644 | | | | | | | | | | | | | | |
| $7\frac{1}{2}$ | | | .614 | | | | | | | | | | | | | |
| 6 | | | | .512 | | | | | | | | | | | | |
| 5 | | | | | .293 | | | | | | | | | | | |
| 3 | .796 | | | .537 | .421 | | | | -.135 | -.118 | -.230 | -.161 | -.247 | -.221 | -.080 | .053 |
| 0 | | .599 | | | | | | .206 | -.279 | -.270 | -.270 | -.200 | | | | |
| -3 | .771 | | | .501 | .367 | | | | -.356 | -.257 | -.213 | -.161 | -.258 | -.165 | -.121 | .041 |
| -5 | | | | | .196 | | | | | | | | | | | |
| -6 | | | | .204 | | | | | | | | | | | | |
| $-7\frac{1}{2}$ | | | .158 | | | | | | | | | | | | | |
| -9 | | .495 | | | | | | | | | | | | | | |
| -12 | .655 | | | .083 | -.016 | | | | -.213 | -.269 | | -.233 | -.119 | | | |
| $-13\frac{1}{2}$ | | .323 | | | | | | | | | | | | | | |
| -22 $\frac{1}{2}$ | | .317 | .177 | | -.176 | | | | -.167 | | -.305 | -.183 | -.112 | | | |
| -33 $\frac{1}{2}$ | | | | -.036 | -.063 | | | | -.140 | -.177 | | | | | | |
| - $P_{x,0}$ | .440 | .317 | -.036 | | | | | | -.046 | -.009 | -.111 | -.177 | -.112 | | | .041 |
| $P_{x,0}$ | $\alpha = 0.4^\circ; \beta = -8^\circ$ | | | | | | | | | | | | | | | |
| $P_{x,0}$ | .790 | .652 | | .265 | | | | | .159 | .170 | .052 | .011 | | -.130 | | .004 |
| $33\frac{1}{2}$ | | | | .265 | .258 | | | | | .070 | .011 | | | | | |
| $22\frac{1}{2}$ | | .652 | .464 | | .252 | | | | | .127 | -.303 | -.080 | -.130 | | | |
| $13\frac{1}{2}$ | | .683 | | | | | | | | | | | | | | |
| 12 | .719 | | | .434 | .366 | | | | .208 | -.110 | | -.173 | -.238 | | | |
| 9 | | .706 | | | | | | | | | | | | | | |
| $7\frac{1}{2}$ | | | .657 | | | | | | | | | | | | | |
| 6 | | | | .590 | | | | | | | | | | | | |
| 5 | | | | | .347 | | | | | | | | | | | |
| 3 | .789 | | | .532 | .414 | | | | -.243 | -.117 | -.262 | -.345 | -.240 | -.257 | -.114 | .304 |
| 0 | | .576 | | | | | | .202 | -.242 | -.245 | -.245 | -.245 | -.245 | -.245 | -.245 | |
| -3 | .766 | | | .478 | .367 | | | | -.375 | -.330 | -.343 | -.141 | -.253 | -.272 | -.149 | .005 |
| -5 | | | | | .172 | | | | | | | | | | | |
| -6 | | | | .295 | | | | | | | | | | | | |
| $-7\frac{1}{2}$ | | | .358 | | | | | | | | | | | | | |
| -9 | | .427 | | | | | | | | | | | | | | |
| -12 | .607 | | | -.021 | -.132 | | | | -.220 | -.312 | | -.303 | -.167 | | | |
| $-13\frac{1}{2}$ | | .029 | | | | | | | | | | | | | | |
| -22 $\frac{1}{2}$ | | .183 | .060 | | -.155 | | | | -.271 | | -.256 | -.205 | -.116 | | | |
| -33 $\frac{1}{2}$ | | | | -.126 | -.162 | | | | -.224 | -.235 | | | | | | |
| - $P_{x,0}$ | .216 | .283 | -.126 | | | | | | -.122 | -.141 | -.201 | -.235 | -.116 | | | .005 |

TABLE XIV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED FLAP-WINGFIELD CANOPY - Continued

| (a) $M=1.41$ | | α/β | | | | | | | | | | | | | | | |
|--------------|------|------------------------------------|---------|---------|---------|---------|---------|---------|---------|---------|---------|---------|---------|---------|---------|---------|---------|
| | | -0.002 | 0.006 | 0.035 | 0.072 | 0.137 | 0.181 | 0.206 | 0.210 | 0.228 | 0.264 | 0.306 | 0.365 | 0.444 | 0.606 | 0.862 | 0.990 |
| | | $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | |
| Po.L. | | .497 | .388 | | .081 | | | | | .033 | .011 | -.071 | -.120 | | -.134 | | .154 |
| 33½ | | | | | .081 | .060 | | | | | -.073 | -.120 | | | | | |
| 22½ | | | .388 | .260 | .053 | | | | | -.052 | | -.238 | -.211 | -.134 | | | |
| 13½ | | | .416 | | | | | | | | | | | | | | |
| 12 | .560 | | .407 | .152 | .074 | | | | | -.076 | | -.213 | -.208 | -.154 | | | |
| 9 | | | .506 | | | | | | | | | | | | | | |
| 7½ | | | | .446 | | | | | | | | | | | | | |
| 6 | | | | | .330 | | | | | | | | | | | | |
| 5 | | | | | | .131 | | | | | | | | | | | |
| 3 | .633 | | .397 | .293 | .165 | | | -.303 | -.219 | -.248 | -.259 | -.224 | -.156 | | | | .154 |
| 0 | | | .406 | | | | .112 | | -.271 | | -.255 | | -.222 | -.055 | | | |
| | | $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | |
| Po.L. | | .594 | .488 | | .169 | | | | | .103 | .062 | -.019 | -.068 | | -.214 | | .056 |
| 33½ | | | | | .169 | .118 | | | | | | -.009 | -.008 | | | | |
| 22½ | | | .412 | .360 | .160 | | | | | .079 | | -.176 | -.173 | -.114 | | | |
| 13½ | | | .524 | | | | | | | | | | | | | | |
| 12 | .570 | | .392 | .268 | .182 | | | | | .070 | | -.205 | -.219 | -.266 | | | |
| 9 | | | .569 | | | | | | | | | | | | | | |
| 7½ | | | | .504 | | | | | | | | | | | | | |
| 6 | | | | | .388 | | | | | | | | | | | | |
| 5 | | | | | | .190 | | | | | | | | | | | |
| 3 | .633 | | .400 | .284 | .156 | | | -.197 | -.216 | -.204 | -.217 | -.285 | -.182 | | | | .056 |
| C | | | .486 | | | | .104 | | -.228 | | -.206 | | -.186 | | | | -.051 |
| -3 | .616 | | .377 | .267 | .158 | | | -.101 | -.333 | -.264 | -.214 | -.268 | -.108 | | | | .028 |
| -5 | | | | .093 | | | | | | | | | | | | | |
| -6 | | | | .219 | | | | | | | | | | | | | |
| -7½ | | | .353 | | | | | | | | | | | | | | |
| -9 | | | .417 | | | | | | | | | | | | | | |
| -12 | .517 | | .403 | .033 | -.057 | | | -.138 | | -.296 | | -.209 | -.122 | | | | |
| -13½ | | | .273 | | | | | | | | | | | | | | |
| -22½ | | | .280 | .119 | -.060 | | | -.243 | | -.287 | -.224 | -.241 | | | | | |
| -33½ | | | -.007 | -.032 | | | | | | -.110 | -.173 | | | | | | |
| -Po.L. | | .360 | .280 | -.007 | | | | -.037 | -.058 | -.189 | -.173 | | -.111 | | | | .028 |
| | | $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | |
| Po.L. | | .612 | .573 | | .263 | | | | | .182 | .160 | -.047 | -.010 | | -.182 | | .015 |
| 33½ | | | | | .263 | .212 | | | | | | .064 | -.010 | | | | |
| 22½ | | | .573 | .455 | .263 | | | | | .110 | | -.118 | -.127 | -.182 | | | |
| 13½ | | | .594 | | | | | | | | | | | | | | |
| 12 | .546 | | .394 | .260 | .204 | | | | | .148 | | -.168 | -.235 | -.190 | | | |
| 9 | | | .597 | | | | | | | | | | | | | | |
| 7½ | | | | .531 | | | | | | | | | | | | | |
| 6 | | | | | .413 | | | | | | | | | | | | |
| 5 | | | | | | .234 | | | | | | | | | | | |
| 3 | .624 | | .394 | .297 | .159 | | | -.114 | -.195 | -.239 | -.124 | -.107 | -.178 | | | | .015 |
| 0 | | | .470 | | | | .099 | | -.211 | | -.155 | | -.200 | | | | -.049 |
| -3 | .600 | | .351 | .259 | .155 | | | -.121 | -.127 | -.120 | -.194 | -.359 | -.185 | | | | -.002 |
| -5 | | | | .069 | | | | | | | | | | | | | |
| -6 | | | | .178 | | | | | | | | | | | | | |
| -7½ | | | .252 | | | | | | | | | | | | | | |
| -9 | | | .249 | | | | | | | | | | | | | | |
| -12 | .442 | | .393 | -.074 | -.178 | | | -.217 | | -.336 | | -.248 | -.131 | | | | |
| -13½ | | | .056 | | | | | | | | | | | | | | |
| -22½ | | | .171 | .043 | -.169 | | | | -.227 | | -.331 | -.216 | -.110 | | | | |
| -33½ | | | -.067 | -.118 | | | | | | -.197 | -.219 | | | | | | |
| -Po.L. | | .194 | .171 | -.087 | | | | -.099 | -.119 | -.179 | -.219 | | -.110 | | | | -.002 |

TABLE XIV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED FLAT-WINDSHIELD CANOPY - Continued

| (b) M=2.01 | | | | | | | | | | | | | | | | | |
|------------------------------------|---------------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|
| $\frac{x}{L}$ | $\frac{y}{L}$ | .002 | .006 | .035 | .072 | .137 | .181 | .206 | .210 | .228 | .264 | .306 | .365 | .444 | .606 | .862 | .990 |
| $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | |
| Full | | .532 | .401 | | .090 | | | | .058 | .043 | .001 | .023 | | .003 | | | .054 |
| 33½ | | | | | .090 | .065 | | | | | | .015 | .023 | | | | |
| 22½ | | | .401 | .315 | | .087 | | | | .011 | | .110 | .085 | .083 | | | |
| 13½ | | | .460 | | | | | | | | | | | | | | |
| 12 | | | | .415 | .255 | .182 | | | .105 | | .056 | | .118 | .077 | | | |
| 9 | | | .622 | | | | | | | | | | | | | | |
| 7½ | | | | .544 | | | | | | | | | | | | | |
| 6 | | | | | .488 | | | | | | | | | | | | |
| 5 | | | | | | .362 | | | | | | | | | | | |
| 3 | | .752 | | | .569 | .480 | .385 | | .074 | .060 | .083 | .092 | .112 | .087 | .037 | .054 | |
| 0 | | | .626 | | | | | .363 | .090 | .090 | .089 | .089 | .091 | | | | .044 |
| $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | |
| Full | | .619 | .497 | | .166 | | | | .123 | .106 | .053 | .029 | | .066 | | | .019 |
| 33½ | | | | | .166 | .116 | | | | | .047 | .029 | | | | | |
| 22½ | | | .497 | .421 | | .180 | | | | .088 | | .059 | .045 | .068 | | | |
| 13½ | | | .569 | | | | | | | | | | | | | | |
| 12 | | | | .527 | .342 | .275 | | | .185 | | .012 | | .101 | .103 | | | |
| 9 | | | .686 | | | | | | | | | | | | | | |
| 7½ | | | | .623 | | | | | | | | | | | | | |
| 6 | | | | | .538 | | | | | | | | | | | | |
| 5 | | | | | | .380 | | | | | | | | | | | |
| 3 | | .783 | | | .573 | .490 | .368 | | .008 | .008 | .095 | .131 | .136 | .112 | .077 | .019 | |
| 0 | | | .630 | | | | .350 | | .103 | | .132 | | .132 | | .109 | .001 | |
| -3 | | .776 | | .551 | .469 | .377 | | .096 | .108 | .094 | .112 | .141 | .134 | .109 | .009 | | |
| -5 | | | | .336 | | | | | | | | | | | | | |
| -6 | | | | .415 | | | | | | | | | | | | | |
| -7½ | | | .464 | | | | | | | | | | | | | | |
| -9 | | | .556 | | | | | | | | | | | | | | |
| -12 | | | .237 | .167 | .096 | | | | .023 | | .103 | | .142 | .080 | | | |
| -13½ | | | .315 | | | | | | | | | | | | | | |
| -22½ | | | .288 | .213 | .001 | | | | .067 | | .164 | .124 | .077 | | | | |
| -33½ | | | | .011 | .018 | | | | .077 | .072 | | | | | | | |
| -Full | | .398 | .288 | | .011 | | | | .007 | .021 | .052 | .072 | | .077 | | .009 | |
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | |
| Full | | .671 | .585 | | .263 | | | | .206 | .178 | .126 | .096 | | .026 | | | .016 |
| 33½ | | | | | .263 | .247 | | | | | .125 | .096 | | | | | |
| 22½ | | | .585 | .510 | | .288 | | | | .184 | | .007 | .012 | .026 | | | |
| 13½ | | | .654 | | | | | | | | | | | | | | |
| 12 | | | | .623 | .447 | .379 | | | .277 | | .032 | | .059 | .023 | | | |
| 9 | | | .732 | | | | | | | | | | | | | | |
| 7½ | | | | .667 | | | | | | | | | | | | | |
| 6 | | | | | .527 | | | | | | | | | | | | |
| 5 | | | | | | .427 | | | | | | | | | | | |
| 3 | | .760 | | | .566 | .508 | .406 | | .066 | .030 | .073 | .140 | .188 | .222 | .145 | .046 | |
| 0 | | | .644 | | | | .349 | | .069 | | .149 | | .214 | | .066 | | |
| -3 | | .751 | | .546 | .460 | .367 | | .101 | .163 | .167 | .187 | .202 | .236 | .149 | .062 | | |
| -5 | | | | .316 | | | | | | | | | | | | | |
| -6 | | | | .378 | | | | | | | | | | | | | |
| -7½ | | | .432 | | | | | | | | | | | | | | |
| -9 | | | .521 | | | | | | | | | | | | | | |
| -12 | | | .036 | .016 | .020 | .027 | | | .020 | | .139 | | .176 | .133 | | | |
| -13½ | | | .036 | | | | | | | | | | | | | | |
| -22½ | | | .175 | .111 | .081 | | | | .138 | | .217 | .158 | .097 | | | | |
| -33½ | | | | .063 | .103 | | | | .133 | .122 | | | | | | | |
| -Full | | .222 | .175 | .063 | | | | .068 | .079 | .104 | .122 | | .197 | | .062 | | |

TABLE XIV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED FLAT-WINGED CANOPY - Continued

| (b) $M=2.01$ | | x/l | .002 | .006 | .015 | .072 | .137 | .181 | .204 | .210 | .228 | .261 | .306 | .365 | .444 | .606 | .862 | .990 |
|--------------|--|---------------------|--|------|-------|-------|-------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|------|
| | | β, deg | $\alpha = 6.5^\circ; \beta = 0^\circ$ | | | | | | | | | | | | | | | |
| P/L | | | .431 | .332 | | .089 | | | | .053 | .018 | -.008 | -.028 | | -.115 | | | .044 |
| 31½ | | | | | | .089 | .068 | | | | | -.013 | -.028 | | | | | |
| 22½ | | | | .332 | .266 | | .073 | | | | .007 | | -.104 | -.093 | -.115 | | | |
| 13½ | | | | .378 | | | | | | | | | | | | | | |
| 12 | | | | | .336 | .190 | .126 | | | .058 | | -.089 | | -.112 | -.108 | | | |
| 9 | | | | | .504 | | | | | | | | | | | | | |
| 7½ | | | | | | .429 | | | | | | | | | | | | |
| 6 | | | | | | | .361 | | | | | | | | | | | |
| 5 | | | | | | | | .244 | | | | | | | | | | |
| 3 | | | .605 | | | .422 | .343 | .259 | | -.115 | -.111 | -.133 | -.130 | -.133 | -.077 | -.042 | .014 | |
| 0 | | | | | .490 | | | | .215 | | -.132 | | -.125 | | -.078 | | -.021 | |
| | | β, deg | $\alpha = 6.5^\circ; \beta = -4^\circ$ | | | | | | | | | | | | | | | |
| P/L | | | .500 | .411 | | .154 | | | | .110 | .094 | .037 | .014 | | -.090 | | -.004 | |
| 31½ | | | | | | .154 | .140 | | | | | .043 | .014 | | | | | |
| 22½ | | | | .411 | .349 | | .157 | | | | .075 | | -.057 | -.060 | -.090 | | | |
| 13½ | | | | .466 | | | | | | | | | | | | | | |
| 12 | | | | | .425 | .268 | .210 | | | .130 | | -.053 | | -.127 | -.152 | | | |
| 9 | | | | | .551 | | | | | | | | | | | | | |
| 7½ | | | | | | .494 | | | | | | | | | | | | |
| 6 | | | | | | | .405 | | | | | | | | | | | |
| 5 | | | | | | | | .268 | | | | | | | | | | |
| 3 | | | .598 | | | .426 | .353 | .265 | | -.052 | -.047 | -.141 | -.176 | -.173 | -.123 | -.096 | -.004 | |
| 0 | | | | | .488 | | | | .234 | | -.158 | | -.170 | | -.114 | | -.006 | |
| -3 | | | .588 | | | .406 | .325 | .252 | | -.144 | -.158 | -.146 | -.152 | -.163 | -.123 | -.114 | -.026 | |
| -5 | | | | | | | .220 | | | | | | | | | | | |
| -6 | | | | | | | .291 | | | | | | | | | | | |
| -7½ | | | | | | .347 | | | | | | | | | | | | |
| -9 | | | | .439 | | | | | | | | | | | | | | |
| -12 | | | | | .326 | .099 | .039 | | | -.023 | | -.135 | | -.149 | -.083 | | | |
| -13½ | | | | | .249 | | | | | | | | | | | | | |
| -22½ | | | | .236 | .273 | | -.023 | | | | -.065 | | -.150 | -.131 | -.116 | | | |
| -31½ | | | | | .017 | -.008 | | | | | | -.066 | -.073 | | | | | |
| -P/L | | | .313 | .236 | | .017 | | | | -.005 | -.019 | -.055 | -.073 | | -.116 | | -.086 | |
| | | β, deg | $\alpha = 6.5^\circ; \beta = -8^\circ$ | | | | | | | | | | | | | | | |
| P/L | | | .522 | .480 | | .225 | | | | .171 | .160 | .095 | .069 | | -.059 | | -.013 | |
| 31½ | | | | | | .225 | .218 | | | | | .108 | .069 | | | | | |
| 22½ | | | | .480 | .422 | | .204 | | | | .153 | | -.007 | -.017 | -.059 | | | |
| 13½ | | | | .517 | | | | | | | | | | | | | | |
| 12 | | | | | .406 | .353 | .294 | | | .203 | | -.019 | | -.106 | -.136 | | | |
| 9 | | | | | .565 | | | | | | | | | | | | | |
| 7½ | | | | | | .535 | | | | | | | | | | | | |
| 6 | | | | | | | .439 | | | | | | | | | | | |
| 5 | | | | | | | | .300 | | | | | | | | | | |
| 3 | | | .560 | | | .431 | .363 | .276 | | -.002 | -.045 | -.131 | -.192 | -.235 | -.207 | -.118 | -.029 | |
| 0 | | | | | .493 | | | | .226 | | -.129 | | -.231 | | -.207 | | -.047 | |
| -3 | | | .557 | | | .400 | .315 | .264 | | -.147 | -.203 | -.214 | -.217 | -.231 | -.211 | -.144 | -.032 | |
| -5 | | | | | | | .204 | | | | | | | | | | | |
| -6 | | | | | | | .255 | | | | | | | | | | | |
| -7½ | | | | | | .315 | | | | | | | | | | | | |
| -9 | | | | .406 | | | | | | | | | | | | | | |
| -12 | | | | | -.065 | -.038 | -.035 | | | -.073 | | -.169 | | -.176 | -.130 | | | |
| -13½ | | | | | .009 | | | | | | | | | | | | | |
| -22½ | | | | .136 | .081 | | -.086 | | | | -.127 | | -.187 | -.156 | -.097 | | | |
| -31½ | | | | | | -.044 | -.073 | | | | | -.125 | -.109 | | | | | |
| -P/L | | | .160 | .136 | | -.044 | | | | -.051 | -.062 | | -.109 | | -.097 | | -.032 | |

TABLE XV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED VEE-WINDSHIELD CANOPY

(a) $M=1.41$

| x/l β, deg | .010 | .030 | .083 | .133 | .176 | .224 | .251 | .260 | .269 | .293 | .300 | .360 | .440 | .600 | .856 | .992 |
|--|------|------|------|------|------|-------|------|-------|-------|------|-------|-------|-------|-------|-------|------|
| $\alpha = 0.4^\circ; \beta = 0^\circ$ | | | | | | | | | | | | | | | | |
| P.L. | .426 | .347 | .352 | .318 | .304 | .092 | | -.014 | | | | -.045 | -.114 | | -.015 | .090 |
| 33 $\frac{1}{2}$ | | | | | | | | -.078 | | | | | -.114 | | | |
| 30 | | | | | .304 | | | | | | | | | | | |
| 29 | | | | | | .024 | | | | | | | | | | |
| 27 $\frac{1}{2}$ | | | | | | .212 | | | | | | | | | | |
| 24 | | | | | | | | -.085 | | | | | | | | |
| 22 $\frac{1}{2}$ | | | | .328 | | | | .186 | -.076 | | -.293 | -.330 | -.246 | -.015 | | |
| 12 | | | .352 | .393 | .421 | .424 | | .391 | | .209 | -.064 | -.425 | -.266 | -.176 | | |
| 4 $\frac{1}{2}$ | | | | | | | | .295 | | | | | | | | |
| 3 | | .347 | .419 | | .507 | | .282 | .078 | | | | -.061 | -.134 | -.293 | | .090 |
| 1 $\frac{1}{2}$ | | | | | | .465 | | | | | | | | | | |
| 1 | | .373 | .437 | .478 | | | | | | | | | | | | |
| 0 | | | | | | | | .130 | | | | -.033 | -.176 | | | .077 |
| $\alpha = 0.4^\circ; \beta = -4^\circ$ | | | | | | | | | | | | | | | | |
| P.L. | .535 | .455 | .466 | .424 | .402 | .185 | | .062 | | | .026 | -.051 | | -.014 | | .062 |
| 33 $\frac{1}{2}$ | | | | | | | | .015 | | | | -.051 | | | | |
| 30 | | | | | .402 | | | | | | | | | | | |
| 29 | | | | | | .140 | | | | | | | | | | |
| 27 $\frac{1}{2}$ | | | | | | .296 | | | | | | | | | | |
| 24 | | | | | | | | -.128 | | | | | | | | |
| 22 $\frac{1}{2}$ | | | | .424 | | | | .230 | -.039 | | -.184 | -.236 | -.198 | -.014 | | |
| 12 | | | .466 | .492 | .500 | .479 | | .434 | | .234 | -.045 | -.425 | -.211 | -.186 | | |
| 4 $\frac{1}{2}$ | | | | | | | | .315 | | | | | | | | |
| 3 | | .455 | .511 | | .573 | | .259 | -.100 | | | | -.134 | -.216 | -.269 | | .062 |
| 1 $\frac{1}{2}$ | | | | | | .480 | | | | | | | | | | |
| 1 | | .476 | .510 | .547 | | | | | | | | | | | | |
| 0 | | | | | | | | .094 | | | | -.047 | -.203 | | | .083 |
| -1 | | .234 | .287 | .327 | | | | | | | | | | | | |
| -1 $\frac{1}{2}$ | | | | | | .371 | | | | | | | | | | |
| -3 | | .221 | .306 | | .391 | | .271 | .142 | | | | -.053 | -.195 | -.310 | | .046 |
| -4 $\frac{1}{2}$ | | | | | | | | .243 | | | | | | | | |
| -12 | | | .226 | .275 | .313 | .335 | | .323 | | .162 | -.095 | -.430 | -.359 | -.148 | | |
| -22 $\frac{1}{2}$ | | | | .204 | | | | .142 | -.113 | | -.353 | -.426 | -.276 | .009 | | |
| -24 | | | | | | | | -.126 | | | | | | | | |
| -27 $\frac{1}{2}$ | | | | | | .139 | | | | | | | | | | |
| -29 | | | | | | -.098 | | | | | | | | | | |
| -30 | | | | | .191 | | | | | | | | | | | |
| -33 $\frac{1}{2}$ | | | | | | | | -.187 | | | | -.176 | | | | |
| -P.L. | .273 | .221 | .226 | .204 | .191 | -.009 | | -.097 | | | | -.120 | -.176 | .009 | | .046 |

TABLE XV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED VEE-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

| x/l ϕ, deg | .010 | .030 | .083 | .133 | .176 | .224 | .251 | .260 | .269 | .293 | .300 | .360 | .440 | .600 | .856 | .992 |
|--|-------|-------|-------|-------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|------|-------|
| $\alpha = 0.4^\circ; \beta = -8^\circ$ | | | | | | | | | | | | | | | | |
| P.L. | .629 | .570 | .589 | .525 | .500 | .287 | | .153 | | | .109 | .022 | | -.004 | | .006 |
| 33 $\frac{1}{2}$ | | | | | | | | .116 | | | | .022 | | | | |
| 30 | | | | | .500 | | | | | | | | | | | |
| 29 | | | | | | .264 | | | | | | | | | | |
| 27 $\frac{1}{2}$ | | | | | | .392 | | | | | | | | | | |
| 24 | | | | | | | | -.097 | | | | | | | | |
| 22 $\frac{1}{2}$ | | | | .525 | | | | .290 | .033 | | -.078 | -.138 | -.134 | -.004 | | |
| 12 | | | .589 | .597 | .585 | .538 | | .480 | | .268 | -.020 | -.382 | -.249 | -.189 | | |
| 4 $\frac{1}{2}$ | | | | | | | | .338 | | | | | | | | |
| 3 | | .570 | .583 | | .617 | | .245 | -.273 | | | -.360 | -.264 | -.258 | | | .006 |
| 1 $\frac{1}{2}$ | | | | | | .445 | | | | | | | | | | |
| 1 | | .558 | .548 | .560 | | | | | | | | | | | | |
| 0 | | | | | | | | -.019 | | | -.113 | -.243 | | | | .085 |
| -1 | -.070 | -.078 | -.051 | | | | | | | | | | | | | |
| -1 $\frac{1}{2}$ | | | | | | .086 | | | | | | | | | | |
| -3 | .092 | .109 | | -.001 | | .057 | .014 | | | | -.109 | -.246 | -.317 | | | -.002 |
| -4 $\frac{1}{2}$ | | | | | | | -.034 | | | | | | | | | |
| -12 | | | .101 | .155 | .196 | .228 | .241 | | .127 | -.116 | -.403 | -.394 | -.202 | | | |
| -22 $\frac{1}{2}$ | | | | .078 | | | .087 | -.154 | | -.383 | -.494 | -.371 | .039 | | | |
| -24 | | | | | | | | -.133 | | | | | | | | |
| -27 $\frac{1}{2}$ | | | | | | .060 | | | | | | | | | | |
| -29 | | | | | | -.217 | | | | | | | | | | |
| -30 | | | | | .065 | | | | | | | | | | | |
| -33 $\frac{1}{2}$ | | | | | | | | -.299 | | | -.235 | | | | | |
| -P.L. | .079 | .092 | .101 | .078 | .065 | -.110 | -.174 | | | | -.192 | -.235 | | .039 | | -.002 |
| $\alpha = 6.5^\circ; \beta = 0^\circ$ | | | | | | | | | | | | | | | | |
| P.L. | .321 | .263 | .288 | .285 | .308 | .118 | | .016 | | | -.025 | -.105 | | -.221 | | .102 |
| 33 $\frac{1}{2}$ | | | | | | | | -.037 | | | | -.105 | | | | |
| 30 | | | | | .308 | | | | | | | | | | | |
| 29 | | | | | | .084 | | | | | | | | | | |
| 27 $\frac{1}{2}$ | | | | | | .208 | | | | | | | | | | |
| 24 | | | | | | | | -.121 | | | | | | | | |
| 22 $\frac{1}{2}$ | | | | .285 | | | | .180 | -.151 | | -.238 | -.249 | -.253 | -.221 | | |
| 12 | | | .288 | .335 | .350 | .341 | | .321 | | .159 | -.087 | -.464 | -.254 | -.199 | | |
| 4 $\frac{1}{2}$ | | | | | | | | .198 | | | | | | | | |
| 3 | | .268 | .327 | | .392 | | .193 | -.041 | | | -.139 | -.251 | -.335 | | | .102 |
| 1 $\frac{1}{2}$ | | | | | | .363 | | | | | | | | | | |
| 1 | | .288 | .340 | .381 | | | | | | | | | | | | |
| 0 | | | | | | | | .040 | | | -.110 | -.237 | | -.117 | | .139 |

TABLE XV. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED VEE-WINDSHIELD CANOPY - Continued

| | | (a) M=141 | | | | | | | | | | | | | | | | |
|------------------------------------|---------------------|-----------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|------|--|
| α/l | β, deg | .012 | .030 | .083 | .133 | .176 | .224 | .251 | .260 | .269 | .293 | .300 | .340 | .440 | .600 | .856 | .992 | |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | | |
| $P_{t, \text{Loc}}$ | | .426 | .377 | .396 | .369 | .392 | .156 | | .092 | | | | .014 | -.016 | | -.269 | .060 | |
| $33\frac{1}{2}$ | | | | | | | | | .053 | | | | | -.016 | | | | |
| 30 | | | | | | .392 | | | | | | | | | | | | |
| 29 | | | | | | | .192 | | | | | | | | | | | |
| $27\frac{1}{2}$ | | | | | | | .295 | | | | | | | | | | | |
| 24 | | | | | | | | | -.055 | | | | | | | | | |
| $22\frac{1}{2}$ | | | | | .369 | | | | .247 | -.037 | | | -.142 | -.167 | -.200 | -.269 | | |
| 19 | | | .396 | .434 | .431 | .398 | | | .360 | | .185 | -.071 | -.433 | -.286 | | -.247 | | |
| $1\frac{1}{2}$ | | | | | | | | | .212 | | | | | | | | | |
| 3 | | .377 | .416 | | .466 | | .180 | -.235 | | | | | -.196 | -.275 | -.285 | | .068 | |
| $1\frac{1}{2}$ | | | | | | .362 | | | | | | | | | | | | |
| 1 | | .388 | .407 | .436 | | | | | | | | | | | | | | |
| 0 | | | | | | | | | .017 | | | | -.138 | -.268 | | -.174 | .068 | |
| -1 | | .141 | .200 | .187 | | | | | | | | | | | | | | |
| $-1\frac{1}{2}$ | | | | | | .222 | | | | | | | | | | | | |
| -3 | | .144 | .221 | | .297 | | .187 | .025 | | | | | -.128 | -.254 | -.351 | | .033 | |
| $-1\frac{1}{2}$ | | | | | | | | .151 | | | | | | | | | | |
| -12 | | | .176 | .227 | .258 | .267 | | .262 | | .122 | -.118 | | -.470 | -.331 | -.140 | | | |
| $-27\frac{1}{2}$ | | | | .153 | | | | .147 | -.149 | | | | -.343 | -.345 | -.294 | -.202 | | |
| -24 | | | | | | | | -.144 | | | | | | | | | | |
| $-27\frac{1}{2}$ | | | | | | .126 | | | | | | | | | | | | |
| -29 | | | | | | -.032 | | | | | | | | | | | | |
| -30 | | | | | .216 | | | | | | | | | | | | | |
| $-33\frac{1}{2}$ | | | | | | | | -.125 | | | | | | -.160 | | | | |
| - $P_{t, \text{Loc}}$ | | .174 | .144 | .176 | .183 | .216 | .032 | -.061 | | | | -.095 | -.160 | | -.202 | | .033 | |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | | | |
| $P_{t, \text{Loc}}$ | | .517 | .464 | .500 | .431 | .461 | .259 | .166 | | | | | .123 | .019 | | -.232 | .029 | |
| $33\frac{1}{2}$ | | | | | | | | .153 | | | | | | .019 | | | | |
| 30 | | | | | | .461 | | | | | | | | | | | | |
| 29 | | | | | | | .307 | | | | | | | | | | | |
| $27\frac{1}{2}$ | | | | | | | .390 | | | | | | | | | | | |
| 24 | | | | | | | | .023 | | | | | | | | | | |
| $22\frac{1}{2}$ | | | | | .431 | | | .304 | .055 | | | | -.047 | -.084 | -.140 | -.232 | | |
| 12 | | | .500 | .530 | .501 | .450 | | .404 | | .213 | -.043 | | -.371 | -.324 | -.280 | | | |
| $1\frac{1}{2}$ | | | | | | | | .238 | | | | | | | | | | |
| 3 | | .484 | .485 | | .504 | | .192 | -.346 | | | | | -.409 | -.428 | -.317 | | .029 | |
| $1\frac{1}{2}$ | | | | | | .328 | | | | | | | | | | | | |
| 1 | | .467 | .439 | .438 | | | | | | | | | | | | | | |
| 0 | | | | | | | | -.130 | | | | | -.198 | -.309 | | -.299 | .034 | |
| -1 | | -.171 | -.149 | -.112 | | | | | | | | | | | | | | |
| $-1\frac{1}{2}$ | | | | | | .015 | | | | | | | | | | | | |
| -3 | | .019 | -.014 | | -.103 | | -.025 | -.102 | | | | | -.193 | -.318 | -.344 | | .025 | |
| $-1\frac{1}{2}$ | | | | | | | | -.154 | | | | | | | | | | |
| -12 | | | .051 | .116 | .158 | .182 | | .202 | | .097 | -.125 | | -.473 | -.401 | -.158 | | | |
| $-22\frac{1}{2}$ | | | | .038 | | | | .107 | -.185 | | | -.390 | -.435 | -.322 | -.176 | | | |
| -24 | | | | | | | | -.147 | | | | | | | | | | |
| $-27\frac{1}{2}$ | | | | | | .048 | | | | | | | | | | | | |
| -29 | | | | | | -.135 | | | | | | | | | | | | |
| -30 | | | | | .119 | | | | | | | | | | | | | |
| $-33\frac{1}{2}$ | | | | | | | | -.213 | | | | | -.211 | | | | | |
| - $P_{t, \text{Loc}}$ | | -.009 | .029 | .061 | .038 | .119 | -.051 | -.134 | | | | -.168 | -.211 | | -.176 | | .025 | |

TABLE XV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED VEE-WINDSHIELD CANOPY - Continued

| (b) $M=2.01$ | | | | | | | | | | | | | | | | | |
|------------------------------------|------|------|------|------|------|-------|------|-------|------|------|-------|-------|-------|-------|------|-------|--|
| x/z θ, deg | .010 | .030 | .083 | .133 | .176 | .224 | .251 | .260 | .269 | .293 | .300 | .360 | .440 | .600 | .856 | .992 | |
| $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | | |
| P.L. | .388 | .329 | .320 | .286 | .297 | .103 | | .044 | | | .017 | -.018 | | -.104 | | .077 | |
| 33 $\frac{1}{2}$ | | | | | | | | -.008 | | | | -.018 | | | | | |
| 30 | | | | | .297 | | | | | | | | | | | | |
| 29 | | | | | | .079 | | | | | | | | | | | |
| 27 $\frac{1}{2}$ | | | | | | .287 | | | | | | | | | | | |
| 24 | | | | | | | | .073 | | | | | | | | | |
| 22 $\frac{1}{2}$ | | | | .286 | | | | .289 | .105 | | -.068 | -.143 | -.113 | -.104 | | | |
| 12 | | .320 | .354 | .379 | .396 | | | .433 | | .375 | .183 | -.172 | -.168 | -.060 | | | |
| 4 $\frac{1}{2}$ | | | | | | | | .362 | | | | | | | | | |
| 3 | | .329 | .406 | | .464 | | .346 | .129 | | | .030 | -.007 | -.078 | -.120 | | .077 | |
| 1 $\frac{1}{2}$ | | | | | | .473 | | | | | | | | | | | |
| 1 | | .370 | .443 | .477 | | | | | | | | | | | | | |
| 0 | | | | | | | | .181 | | | | .006 | | -.126 | | .094 | |
| $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | | |
| P.L. | .520 | .442 | .432 | .385 | .404 | .201 | | .133 | | | .104 | .055 | | -.085 | | .036 | |
| 33 $\frac{1}{2}$ | | | | | | | | .091 | | | | .055 | | | | | |
| 30 | | | | | .404 | | | | | | | | | | | | |
| 29 | | | | | | .189 | | | | | | | | | | | |
| 27 $\frac{1}{2}$ | | | | | | .381 | | | | | | | | | | | |
| 24 | | | | | | | | .126 | | | | | | | | | |
| 22 $\frac{1}{2}$ | | | | .385 | | | | .362 | .159 | | -.021 | -.069 | -.034 | -.085 | | | |
| 12 | | .432 | .467 | .485 | .493 | | | .519 | | .442 | .237 | -.150 | -.141 | -.059 | | | |
| 4 $\frac{1}{2}$ | | | | | | | | .422 | | | | | | | | | |
| 3 | | .442 | .508 | | .557 | | .394 | .015 | | | -.071 | -.016 | -.080 | -.091 | | .036 | |
| 1 $\frac{1}{2}$ | | | | | | .513 | | | | | | | | | | | |
| 1 | | .473 | .521 | .550 | | | | | | | | | | | | | |
| 0 | | | | | | | | .152 | | | | -.014 | | -.116 | | .030 | |
| -1 | | .268 | .317 | .331 | | | | | | | | | | | | | |
| -1 $\frac{1}{2}$ | | | | | | .363 | | | | | | | | | | | |
| -3 | | .228 | .311 | | .384 | | .318 | .191 | | | .060 | -.009 | -.085 | -.125 | | -.027 | |
| -4 $\frac{1}{2}$ | | | | | | | | .286 | | | | | | | | | |
| -12 | | | .220 | .257 | .287 | .306 | | .340 | | .295 | .127 | -.165 | -.150 | -.105 | | | |
| -22 $\frac{1}{2}$ | | | | .197 | | | | .220 | .056 | | -.101 | -.183 | -.170 | -.090 | | | |
| -24 | | | | | | | | .023 | | | | | | | | | |
| -27 $\frac{1}{2}$ | | | | | | .201 | | | | | | | | | | | |
| -29 | | | | | | -.006 | | | | | | | | | | | |
| -30 | | | | | .194 | | | | | | | | | | | | |
| -33 $\frac{1}{2}$ | | | | | | | | -.088 | | | | -.084 | | | | | |
| -P.L. | .259 | .228 | .220 | .197 | .194 | .022 | | -.026 | | | -.050 | -.084 | | -.090 | | -.027 | |

TABLE XV - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED VEE-WINDFIELD JAWOFT - Continue:

(b) $M=2.01$

| α/γ θ, deg | .010 | .030 | .083 | .133 | .176 | .224 | .251 | .260 | .269 | .293 | .300 | .360 | .440 | .600 | .856 | .992 |
|---|------|------|------|------|------|-------|------|-------|------|------|-------|-------|-------|-------|------|-------|
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | |
| P.L. | .620 | .562 | .544 | .450 | .492 | .285 | | .217 | | | .187 | .123 | | -.041 | | -.035 |
| 33 $\frac{1}{2}$ | | | | | | | | .189 | | | | .123 | | | | |
| 30 | | | | | .492 | | | | | | | | | | | |
| 29 | | | | | | .305 | | | | | | | | | | |
| 27 $\frac{1}{2}$ | | | | | | .471 | | | | | | | | | | |
| 24 | | | | | | | | .174 | | | | | | | | |
| 22 $\frac{1}{2}$ | | | | .450 | | | | .423 | .206 | | .049 | .008 | .027 | -.041 | | |
| 12 | | | .544 | .581 | .590 | .587 | | .603 | .505 | .280 | -.119 | -.124 | -.086 | | | |
| 4 $\frac{1}{2}$ | | | | | | | | .475 | | | | | | | | |
| 3 | | .562 | .599 | | .640 | | .452 | -.012 | | | -.109 | -.141 | -.143 | -.143 | | -.035 |
| 1 $\frac{1}{2}$ | | | | | | .532 | | | | | | | | | | |
| 1 | | .552 | .566 | .584 | | | | | | | | | | | | |
| 0 | | | | | | | | .045 | | | | -.104 | | -.142 | | -.009 |
| -1 | | .091 | .031 | .050 | | | | | | | | | | | | |
| -1 $\frac{1}{2}$ | | | | | | .143 | | | | | | | | | | |
| -3 | | .125 | .227 | | .232 | | .152 | .108 | | | .030 | -.068 | -.140 | -.144 | | -.034 |
| -4 $\frac{1}{2}$ | | | | | | | | .197 | | | | | | | | |
| -12 | | | .126 | .165 | .195 | .213 | | .245 | .214 | .069 | -.172 | -.148 | -.149 | | | |
| -22 $\frac{1}{2}$ | | | | .101 | | | | .148 | .009 | | -.127 | -.202 | -.196 | -.136 | | |
| -24 | | | | | | | | -.023 | | | | | | | | |
| -27 $\frac{1}{2}$ | | | | | | .122 | | | | | | | | | | |
| -29 | | | | | | -.072 | | | | | | | | | | |
| -30 | | | | | .095 | | | | | | | | | | | |
| -33 $\frac{1}{2}$ | | | | | | | | -.149 | | | | -.179 | | | | |
| -P.L. | .124 | .125 | .126 | .101 | .095 | -.056 | | -.095 | | | -.111 | -.179 | | -.136 | | -.034 |
| $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | |
| P.L. | .287 | .252 | .254 | .239 | .254 | .113 | | .054 | | | .030 | -.016 | | -.123 | | .029 |
| 33 $\frac{1}{2}$ | | | | | | | | .022 | | | | -.016 | | | | |
| 30 | | | | | .254 | | | | | | | | | | | |
| 29 | | | | | | .095 | | | | | | | | | | |
| 27 $\frac{1}{2}$ | | | | | | .251 | | | | | | | | | | |
| 24 | | | | | | | | .033 | | | | | | | | |
| 22 $\frac{1}{2}$ | | | | .239 | | | | .230 | .067 | | -.086 | -.122 | -.087 | -.123 | | |
| 12 | | | .254 | .299 | .316 | .324 | | .347 | .298 | .130 | -.187 | -.175 | -.094 | | | |
| 4 $\frac{1}{2}$ | | | | | | | | .266 | | | | | | | | |
| 3 | | .252 | .321 | | .361 | | .253 | .029 | | | -.049 | -.054 | -.109 | -.091 | | .029 |
| 1 $\frac{1}{2}$ | | | | | | .347 | | | | | | | | | | |
| 1 | | .275 | .332 | .357 | | | | | | | | | | | | |
| 0 | | | | | | | | .101 | | | | -.038 | | -.099 | | .041 |

TABLE XV.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED VEE-WINDSHIELD CANOPY - Concluded

| (b) M=2.01 | | | | | | | | | | | | | | | | |
|------------------------------------|------|-------|-------|-------|------|-------|------|-------|-------|-------|-------|-------|-------|-------|------|-------|
| α/β | .010 | .030 | .083 | .133 | .176 | .224 | .251 | .260 | .269 | .293 | .300 | .360 | .440 | .600 | .856 | .992 |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | |
| PoI. | .404 | .354 | .316 | .304 | .338 | .179 | | .124 | | | .103 | .043 | | -.029 | | .037 |
| 33½ | | | | | | | | .113 | | | | .043 | | | | |
| 30 | | | | | .338 | | | | | | | | | | | |
| 29 | | | | | | .211 | | | | | | | | | | |
| 27½ | | | | | | .346 | | | | | | | | | | |
| 24 | | | | | | | | .092 | | | | | | | | |
| 22½ | | | | .304 | | | | .251 | .109 | | -.050 | -.056 | -.037 | -.089 | | |
| 12 | | | .316 | .398 | .406 | .406 | | .420 | | .353 | .173 | -.170 | -.170 | -.162 | | |
| 1½ | | | | | | | | .322 | | | | | | | | |
| 3 | | .254 | .407 | | .444 | | .294 | -.065 | | | -.140 | -.128 | -.181 | -.116 | | .037 |
| 1½ | | | | | | .384 | | | | | | | | | | |
| 1 | | .378 | .407 | .422 | | | | | | | | | | | | |
| 0 | | | | | | | | .076 | | | | -.066 | | -.128 | | .029 |
| -1 | | .158 | .195 | .203 | | | | | | | | | | | | |
| -1½ | | | | | | .257 | | | | | | | | | | |
| -3 | | .150 | .225 | | .279 | | .219 | .096 | | | -.003 | -.055 | -.116 | -.122 | | -.023 |
| -1½ | | | | | | | | .203 | | | | | | | | |
| -12 | | | .167 | .203 | .229 | .242 | | .270 | | .235 | .083 | -.192 | -.196 | -.079 | | |
| -22½ | | | | .165 | | | | .177 | .031 | | -.115 | -.172 | -.135 | -.131 | | |
| -24 | | | | | | | | .003 | | | | | | | | |
| -27½ | | | | | | .196 | | | | | | | | | | |
| -29 | | | | | | .016 | | | | | | | | | | |
| -30 | | | | | .181 | | | | | | | | | | | |
| -33½ | | | | | | | | -.053 | | | | -.059 | | | | |
| -PoI. | .165 | .150 | .167 | .165 | .181 | .049 | | -.001 | | -.023 | -.059 | | -.131 | | | -.023 |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | | |
| PoI. | .493 | .451 | .423 | .330 | .379 | .230 | | .177 | | .167 | .132 | | -.050 | | | -.031 |
| 33½ | | | | | | | | .203 | | | .102 | | | | | |
| 30 | | | | | .379 | | | | | | | | | | | |
| 29 | | | | | | .304 | | | | | | | | | | |
| 27½ | | | | | | .435 | | | | | | | | | | |
| 24 | | | | | | | | .154 | | | | | | | | |
| 22½ | | | | .330 | | | | .248 | .159 | | .053 | .012 | .003 | -.050 | | |
| 12 | | | .423 | .504 | .497 | .484 | | .491 | | .402 | .209 | -.132 | -.133 | -.132 | | |
| 1½ | | | | | | | | .366 | | | | | | | | |
| 3 | | .451 | .462 | | .504 | | .340 | -.067 | | | -.135 | -.173 | -.169 | -.173 | | -.031 |
| 1½ | | | | | | .389 | | | | | | | | | | |
| 1 | | .450 | .439 | .422 | | | | | | | | | | | | |
| 0 | | | | | | | | -.045 | | | | -.195 | | -.158 | | -.015 |
| -1 | | -.023 | -.064 | -.050 | | | | | | | | | | | | |
| -1½ | | | | | | .040 | | | | | | | | | | |
| -3 | | .016 | .029 | | .064 | | .029 | -.016 | | -.071 | -.163 | -.207 | -.151 | | | -.024 |
| -1½ | | | | | | | | .046 | | | | | | | | |
| -12 | | | .073 | .112 | .139 | .159 | | .186 | | .161 | .031 | -.195 | -.165 | -.110 | | |
| -22½ | | | | .082 | | | | .119 | -.009 | | -.140 | -.177 | -.159 | -.117 | | |
| -24 | | | | | | | | -.041 | | | | | | | | |
| -27½ | | | | | | .111 | | | | | | | | | | |
| -29 | | | | | | -.053 | | | | | | | | | | |
| -30 | | | | | .100 | | | | | | | | | | | |
| -33½ | | | | | | | | -.113 | | | | -.101 | | | | |
| -PoI. | .022 | .016 | .073 | .082 | .106 | -.015 | | -.056 | | -.078 | -.101 | | -.117 | | | -.024 |

TABLE XVI.- PRESSURE COEFFICIENTS ICP CONFIGURATION WITH REARWARD-LOCATED ROUND-WINDSHIELD CANOPY

(a) $M=1.41$

| x/γ ϕ, deg | .010 | .030 | .068 | .096 | .133 | .176 | .224 | .260 | .302 | .360 | .440 | .600 | .856 | .992 |
|------------------------------------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|------|
| $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | |
| P.L. | .687 | .491 | .372 | .308 | .237 | .124 | -.017 | -.061 | -.095 | -.157 | -.152 | -.156 | -.069 | .073 |
| 33 $\frac{1}{2}$ | | | | | | .124 | -.026 | | -.139 | -.157 | | | | |
| 22 $\frac{1}{2}$ | | | | .308 | | .163 | .007 | | -.199 | -.238 | -.164 | -.156 | | |
| 12 | | .491 | .442 | .350 | .235 | .036 | -.072 | -.188 | -.292 | -.235 | -.148 | -.069 | | |
| 3 | .687 | .648 | .594 | .567 | .545 | | .060 | | -.198 | | -.253 | -.187 | -.060 | .073 |
| 0 | | .647 | | .589 | | | .100 | -.009 | | -.262 | | -.154 | | .081 |
| $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | |
| P.L. | .738 | .613 | .501 | .442 | .357 | .233 | .075 | .015 | -.024 | -.097 | -.108 | -.158 | -.144 | .059 |
| 33 $\frac{1}{2}$ | | | | | | .233 | .057 | | -.069 | -.097 | | | | |
| 22 $\frac{1}{2}$ | | | | .442 | | .267 | .097 | | -.119 | -.159 | -.112 | -.158 | | |
| 12 | | .613 | | .563 | .459 | .334 | .133 | .006 | -.134 | -.246 | -.203 | -.162 | -.144 | |
| 3 | .738 | .685 | .612 | .582 | .559 | | .103 | | -.193 | | -.263 | -.287 | -.086 | .059 |
| 0 | | .617 | | .561 | | | .089 | -.005 | | -.290 | | -.220 | | .058 |
| -3 | .601 | .577 | .529 | .514 | .490 | | -.006 | | -.229 | | -.294 | -.212 | -.129 | .044 |
| -12 | | .346 | | .297 | .218 | .118 | -.074 | -.158 | -.254 | -.332 | -.278 | -.125 | -.042 | |
| -22 $\frac{1}{2}$ | | | | .176 | | .054 | -.083 | | -.274 | -.304 | -.200 | -.133 | | |
| -33 $\frac{1}{2}$ | | | | | | .018 | -.118 | | -.206 | -.212 | | | | |
| -P.L. | .601 | .346 | .232 | .176 | .114 | .018 | -.099 | -.133 | -.158 | -.212 | -.193 | -.133 | -.042 | .044 |
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | |
| P.L. | .745 | .718 | .625 | .578 | .467 | .351 | .176 | .100 | .055 | -.028 | -.053 | -.134 | -.269 | .009 |
| 33 $\frac{1}{2}$ | | | | | | .351 | .162 | | .011 | -.028 | | | | |
| 22 $\frac{1}{2}$ | | | | .578 | | .367 | .193 | | -.030 | -.099 | -.065 | -.134 | | |
| 12 | | .718 | | .657 | .551 | .417 | .215 | .077 | -.081 | -.200 | -.173 | -.203 | -.269 | |
| 3 | .746 | .688 | .583 | .550 | .528 | | .109 | | -.076 | | -.332 | -.457 | -.179 | .009 |
| 0 | | .544 | | .482 | | | .035 | -.039 | | -.345 | | -.448 | | .011 |
| -3 | .472 | .463 | .415 | .412 | .388 | | -.096 | | -.299 | | -.414 | -.370 | -.160 | .008 |
| -12 | | .180 | | .127 | .061 | -.029 | -.195 | -.262 | -.335 | -.390 | -.341 | -.167 | -.087 | |
| -22 $\frac{1}{2}$ | | | | .039 | | -.058 | -.179 | | -.348 | -.374 | -.234 | -.130 | | |
| -33 $\frac{1}{2}$ | | | | | | -.057 | -.200 | | -.274 | -.267 | | | | |
| -P.L. | .472 | .180 | .086 | .039 | -.001 | -.087 | -.174 | -.202 | -.221 | -.267 | -.229 | -.130 | -.087 | .008 |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | |
| P.L. | .583 | .502 | .426 | .406 | .327 | .236 | .103 | .037 | -.003 | -.097 | -.128 | -.210 | -.126 | .058 |
| 33 $\frac{1}{2}$ | | | | | | .236 | .120 | | -.048 | -.097 | | | | |
| 22 $\frac{1}{2}$ | | | | .406 | | .253 | .094 | | -.115 | -.170 | -.153 | -.210 | | |
| 12 | | .502 | | .497 | .397 | .278 | .060 | -.046 | -.182 | -.273 | -.237 | -.253 | -.126 | |
| 3 | .583 | .592 | .498 | .464 | .452 | | .012 | | -.267 | | -.289 | -.196 | | .058 |
| 0 | | .541 | | .446 | | | -.013 | -.101 | | -.334 | | -.177 | | .058 |
| -3 | .472 | .491 | .425 | .410 | .384 | | -.088 | | -.284 | | -.261 | -.172 | | .043 |
| -12 | | .275 | | .260 | .183 | .089 | -.092 | -.151 | -.270 | -.344 | -.237 | -.121 | -.060 | |
| -22 $\frac{1}{2}$ | | | | .166 | | .065 | -.059 | | -.251 | -.276 | -.209 | -.160 | | |
| -33 $\frac{1}{2}$ | | | | | | .049 | -.065 | | -.176 | -.199 | | | | |
| -P.L. | .472 | .275 | .198 | .166 | .136 | .049 | -.055 | -.101 | -.128 | -.199 | -.205 | -.160 | -.060 | .043 |

TABLE XVI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED ROUND-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

| α/β deg | .010 | .030 | .068 | .096 | .133 | .176 | .224 | .260 | .302 | .360 | .440 | .600 | .856 | .992 |
|--------------------------------------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|------|
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | |
| P.L. | .589 | .584 | .532 | .534 | .422 | .331 | .193 | .114 | .069 | -.029 | -.072 | -.182 | -.234 | .007 |
| 33 $\frac{1}{2}$ | | | | | | .331 | .221 | .026 | -.029 | | | | | |
| 22 $\frac{1}{2}$ | | | | .534 | | .344 | .178 | -.050 | -.108 | -.106 | -.182 | | | |
| 12 | | .584 | | .575 | .467 | .344 | .130 | .012 | -.142 | -.248 | -.224 | -.274 | -.234 | |
| 3 | .589 | .591 | .466 | .422 | .418 | | .006 | -.307 | | -.414 | -.342 | | | .007 |
| 0 | | .467 | | .364 | | | -.060 | -.144 | | -.405 | | -.335 | | .015 |
| -3 | .362 | .350 | .313 | .306 | .276 | | -.181 | -.362 | | -.390 | -.394 | | | .011 |
| -12 | | .122 | | .104 | .038 | -.042 | -.204 | -.275 | -.338 | -.386 | -.281 | -.115 | -.106 | |
| -22 $\frac{1}{2}$ | | | | .046 | | -.036 | -.142 | | -.320 | -.326 | -.231 | -.128 | | |
| -33 $\frac{1}{2}$ | | | | | | -.042 | -.146 | | -.233 | -.245 | | | | |
| -P.L. | .362 | .122 | .075 | .046 | .033 | -.042 | -.128 | -.160 | -.184 | -.245 | -.228 | -.128 | -.106 | .011 |
| $\alpha=-6.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .827 | .557 | .395 | .311 | .226 | .095 | -.045 | -.069 | -.106 | -.148 | -.118 | -.096 | -.038 | .101 |
| 33 $\frac{1}{2}$ | | | | | | .095 | -.030 | | -.166 | -.148 | | | | |
| 22 $\frac{1}{2}$ | | | | .311 | | .163 | .004 | | -.211 | -.253 | -.133 | -.096 | | |
| 12 | | .557 | | .493 | .399 | .275 | .063 | -.035 | -.156 | -.255 | -.216 | -.111 | -.038 | |
| 3 | .827 | .754 | .715 | .696 | .681 | | .158 | -.120 | | -.203 | -.148 | | | .101 |
| 0 | | .768 | | .728 | | | .215 | .067 | | -.195 | | -.150 | | .106 |
| $\alpha=-3.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .760 | .516 | .373 | .294 | .220 | .101 | -.037 | -.077 | -.103 | -.157 | -.134 | -.123 | -.050 | .090 |
| 33 $\frac{1}{2}$ | | | | | | .101 | -.023 | | -.156 | -.157 | | | | |
| 22 $\frac{1}{2}$ | | | | .294 | | .156 | .001 | | -.212 | -.247 | -.147 | -.123 | | |
| 12 | | .516 | | .460 | .365 | .249 | .041 | -.060 | -.176 | -.274 | -.221 | -.123 | -.050 | |
| 3 | .760 | .702 | .653 | .631 | .617 | | .105 | | -.158 | | -.222 | -.144 | | .090 |
| 0 | | .711 | | .664 | | | .160 | .013 | | -.228 | | -.145 | | .094 |
| $\alpha=3.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .613 | .431 | .326 | .271 | .220 | .124 | -.011 | -.063 | -.093 | -.161 | -.164 | -.176 | -.073 | .080 |
| 33 $\frac{1}{2}$ | | | | | | .124 | -.014 | | -.138 | -.161 | | | | |
| 22 $\frac{1}{2}$ | | | | .271 | | .142 | -.004 | | -.203 | -.237 | -.173 | -.176 | | |
| 12 | | .431 | | .396 | .305 | .195 | .004 | -.104 | -.213 | -.303 | -.225 | -.148 | -.073 | |
| 3 | .613 | .603 | .534 | .505 | .492 | | .010 | | -.227 | | -.248 | -.138 | | .080 |
| 0 | | .607 | | .538 | | | .057 | -.083 | | -.283 | | -.139 | | .095 |
| $\alpha=6.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .543 | .387 | .299 | .266 | .214 | .134 | .004 | -.049 | -.081 | -.155 | -.173 | -.202 | -.079 | .104 |
| 33 $\frac{1}{2}$ | | | | | | .134 | .010 | | -.125 | -.155 | | | | |
| 22 $\frac{1}{2}$ | | | | .266 | | .141 | -.005 | | -.193 | -.232 | -.186 | -.202 | | |
| 12 | | .387 | | .370 | .281 | .175 | -.011 | -.120 | -.228 | -.312 | -.231 | -.159 | -.079 | |
| 3 | .543 | .557 | .482 | .455 | .439 | | -.029 | | -.253 | | -.245 | -.134 | | .104 |
| 0 | | .560 | | .482 | | | .012 | -.123 | | -.299 | | -.135 | | .137 |

TABLE XVI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED FOUNDRY-WINDSHIELD CANOPY - Continued

(a) $M=1.41$

| x/l | .010 | .030 | .068 | .096 | .133 | .176 | .224 | .260 | .302 | .360 | .440 | .600 | .856 | .992 |
|----------------------|--------------------------------------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|------|
| θ, deg | $\alpha=9.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | |
| P.L. | .474 | .344 | .275 | .260 | .205 | .139 | .021 | -.030 | -.064 | -.116 | -.173 | -.227 | -.084 | .110 |
| 33 $\frac{1}{2}$ | | | | | | .139 | .060 | | | -.110 | -.146 | | | |
| 22 $\frac{1}{2}$ | | | | .260 | | .145 | -.001 | | -.130 | -.225 | -.195 | -.227 | | |
| 12 | | .344 | | .353 | .264 | .160 | -.022 | -.130 | -.238 | -.315 | -.240 | -.164 | -.084 | |
| 3 | .474 | .511 | .437 | .420 | .393 | | -.061 | | -.272 | | -.234 | -.129 | | .110 |
| 0 | | .516 | | .435 | | | -.025 | -.154 | | -.309 | | -.131 | | .173 |
| θ, deg | $\alpha=12.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | |
| P.L. | .405 | .301 | .248 | .245 | .193 | .135 | .031 | -.021 | -.057 | -.138 | -.175 | -.256 | -.097 | .164 |
| 33 $\frac{1}{2}$ | | | | | | .135 | .059 | | -.099 | -.138 | | | | |
| 22 $\frac{1}{2}$ | | | | .245 | | .146 | -.002 | | -.175 | -.222 | -.207 | -.256 | | |
| 12 | | .301 | | .334 | .245 | .142 | -.034 | -.142 | -.251 | -.324 | -.258 | -.175 | -.097 | |
| 3 | .405 | .457 | .398 | .365 | .346 | | -.054 | | -.294 | | -.227 | -.131 | | .164 |
| 0 | | .466 | | .389 | | | -.054 | -.186 | | -.317 | | -.134 | | .179 |

(b) $M=2.01$

| x/l | .010 | .030 | .068 | .096 | .133 | .176 | .224 | .260 | .302 | .360 | .440 | .600 | .856 | .992 |
|----------------------|------------------------------------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|
| θ, deg | $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | |
| P.L. | .696 | .535 | .436 | .400 | .330 | .241 | .129 | .080 | .048 | -.003 | -.017 | -.078 | -.128 | .018 |
| 33 $\frac{1}{2}$ | | | | | | .241 | .116 | | .010 | -.003 | | | | |
| 22 $\frac{1}{2}$ | | | | .400 | | .291 | .171 | | -.007 | -.065 | -.038 | -.078 | | |
| 12 | | .535 | | .555 | .464 | .383 | .213 | .109 | -.010 | -.102 | -.099 | -.091 | -.128 | |
| 3 | .696 | .694 | .619 | .604 | .605 | | .200 | | -.045 | | -.123 | -.139 | -.082 | .018 |
| 0 | | .660 | | .586 | | | .190 | .082 | | -.124 | | -.137 | | .017 |
| -3 | .560 | .580 | .530 | .535 | .527 | | .122 | | -.082 | | -.130 | -.138 | -.120 | -.013 |
| -12 | | .290 | | .304 | .245 | .180 | .044 | -.030 | -.108 | -.162 | -.147 | -.095 | -.025 | |
| -22 $\frac{1}{2}$ | | | | .156 | | .094 | .009 | | -.120 | -.170 | -.124 | -.095 | | |
| -33 $\frac{1}{2}$ | | | | | | .045 | -.042 | | -.109 | -.106 | | | | |
| -P.L. | .560 | .290 | .157 | .156 | .122 | .045 | -.031 | -.058 | -.077 | -.106 | -.108 | -.095 | -.025 | -.013 |
| θ, deg | $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | |
| P.L. | .716 | .635 | .551 | .530 | .436 | .351 | .223 | .165 | .123 | .057 | .039 | -.050 | -.129 | -.059 |
| 33 $\frac{1}{2}$ | | | | | | .351 | .211 | | .084 | .057 | | | | |
| 22 $\frac{1}{2}$ | | | | .530 | | .393 | .256 | | .057 | -.004 | .014 | -.050 | | |
| 12 | | .625 | | .660 | .561 | .470 | .290 | .170 | .036 | -.071 | -.068 | -.064 | -.129 | |
| 3 | .716 | .703 | .620 | .595 | .606 | | .214 | | -.051 | | -.154 | -.206 | -.156 | -.059 |
| 0 | | .614 | | .537 | | | .159 | .064 | | -.153 | | -.227 | | -.049 |
| -3 | .456 | .467 | .452 | .466 | .457 | | .072 | | -.119 | | -.185 | -.226 | -.166 | -.061 |
| -12 | | .169 | | .174 | .133 | .076 | -.039 | -.095 | -.157 | -.198 | -.177 | -.145 | -.079 | |
| -22 $\frac{1}{2}$ | | | | .043 | | .002 | -.067 | | -.167 | -.206 | -.160 | -.114 | | |
| -33 $\frac{1}{2}$ | | | | | | -.038 | -.109 | | -.153 | -.146 | | | | |
| -P.L. | .456 | .169 | .086 | .043 | .030 | -.038 | -.091 | -.110 | -.124 | -.140 | -.144 | -.114 | -.075 | -.061 |

TABLE XVI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED ROUND-WINDSEIELD CANOPY - Continued

(b) $M=2.01$

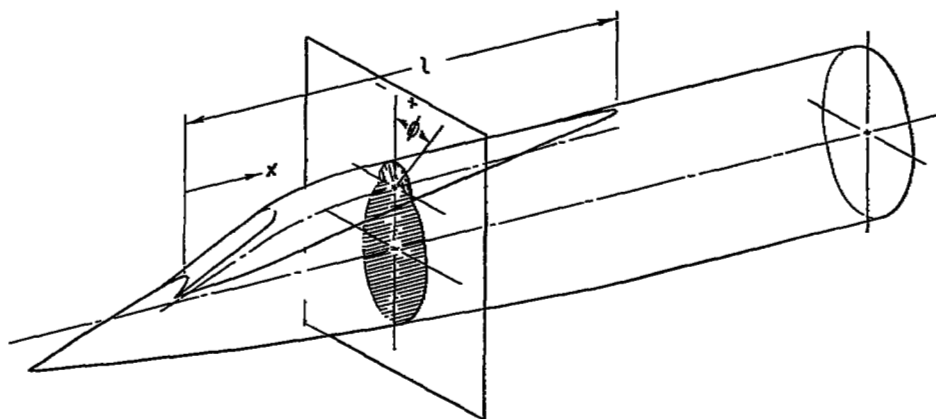
| x/γ θ, deg | .010 | .030 | .068 | .096 | .133 | .176 | .224 | .260 | .302 | .360 | .440 | .600 | .856 | .992 |
|--------------------------------------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | |
| P.L. | .519 | .410 | .338 | .323 | .259 | .200 | .109 | .077 | .043 | -.018 | -.039 | -.113 | -.139 | .001 |
| $33\frac{1}{2}$ | | | | | | .200 | .116 | | .007 | -.018 | | | | |
| $22\frac{1}{2}$ | | | | .323 | | .246 | .136 | | -.034 | -.078 | -.067 | -.113 | | |
| 12 | | .410 | | .456 | .366 | .291 | .144 | .049 | -.063 | -.147 | -.139 | -.144 | -.139 | |
| 3 | .519 | .548 | .474 | .451 | .449 | | .097 | | -.110 | | -.170 | -.152 | -.103 | .001 |
| 0 | | .525 | | .438 | | | .087 | -.002 | | -.170 | | -.137 | | .012 |
| -3 | .405 | .449 | .398 | .394 | .383 | | .033 | | -.134 | | -.154 | -.133 | -.149 | -.002 |
| -12 | | .205 | | .237 | .175 | .118 | -.001 | -.064 | -.133 | -.182 | -.161 | -.097 | -.054 | |
| $-22\frac{1}{2}$ | | | | .125 | | .074 | -.009 | | -.126 | -.164 | -.130 | -.130 | | |
| $-33\frac{1}{2}$ | | | | | | .050 | -.030 | | -.097 | -.106 | | | | |
| -P.L. | .405 | .205 | .144 | .125 | .102 | .050 | -.016 | -.046 | -.066 | -.106 | -.110 | -.130 | -.054 | -.002 |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | |
| P.L. | .532 | .486 | .418 | .423 | .331 | .271 | .177 | .150 | .115 | .040 | .012 | -.082 | -.153 | -.021 |
| $33\frac{1}{2}$ | | | | | | .271 | .208 | | .074 | .040 | | | | |
| $22\frac{1}{2}$ | | | | .423 | | .342 | .215 | | .027 | -.023 | -.018 | -.082 | | |
| 12 | | .486 | | .542 | .444 | .363 | .202 | .097 | -.026 | -.119 | -.117 | -.134 | -.153 | |
| 3 | .532 | .553 | .466 | .437 | .446 | | .103 | | -.120 | | -.207 | -.226 | -.133 | -.021 |
| 0 | | .470 | | .383 | | | .053 | -.023 | | -.202 | | -.223 | | -.026 |
| -3 | .299 | .344 | .311 | .318 | .306 | | -.020 | | -.178 | | -.225 | -.247 | -.157 | -.049 |
| -12 | | .066 | | .103 | .065 | .018 | -.081 | -.134 | -.183 | -.212 | -.185 | -.106 | -.100 | |
| $-22\frac{1}{2}$ | | | | .029 | | -.008 | -.073 | | -.172 | -.200 | -.155 | -.117 | | |
| $-33\frac{1}{2}$ | | | | | | -.022 | -.091 | | -.137 | -.141 | | | | |
| -P.L. | .299 | .086 | .044 | .029 | .025 | -.022 | -.075 | -.094 | -.111 | -.141 | -.140 | -.117 | -.100 | -.049 |
| $\alpha=-6.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .818 | .532 | .395 | .319 | .260 | .146 | .051 | .018 | -.004 | -.039 | -.038 | -.043 | -.021 | .097 |
| $33\frac{1}{2}$ | | | | | | .146 | .037 | | -.050 | -.039 | | | | |
| $22\frac{1}{2}$ | | | | .319 | | .233 | .124 | | -.053 | -.112 | -.070 | -.043 | | |
| 12 | | .532 | | .533 | .450 | .372 | .201 | .104 | -.005 | -.093 | -.087 | -.056 | -.021 | |
| 3 | .818 | .799 | .747 | .749 | .748 | | .287 | | .009 | | -.056 | -.088 | -.018 | .097 |
| 0 | | .834 | | .780 | | | .334 | .200 | | -.052 | | -.087 | | .102 |
| $\alpha=-3.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .732 | .473 | .354 | .292 | .239 | .138 | .044 | .009 | -.016 | -.050 | -.055 | -.073 | -.040 | .076 |
| $33\frac{1}{2}$ | | | | | | .138 | .035 | | -.054 | -.050 | | | | |
| $22\frac{1}{2}$ | | | | .292 | | .207 | .102 | | -.064 | -.119 | -.079 | -.073 | | |
| 12 | | .473 | | .479 | .398 | .327 | .163 | .071 | -.033 | -.115 | -.109 | -.071 | -.040 | |
| 3 | .732 | .725 | .668 | .668 | .665 | | .227 | | -.027 | | -.083 | -.097 | -.032 | .076 |
| 0 | | .758 | | .694 | | | .273 | .146 | | -.083 | | -.096 | | .084 |

TABLE XVI.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED ROUND-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

| α/β ϕ, deg | .010 | .030 | .068 | .096 | .133 | .176 | .224 | .260 | .302 | .360 | .440 | .600 | .856 | .992 |
|--------------------------------------|------|------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|------|
| $\alpha=0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .651 | .421 | .314 | .272 | .224 | .139 | .042 | .007 | -.019 | -.055 | -.063 | -.092 | -.048 | .064 |
| 33 $\frac{1}{2}$ | | | | | | .139 | .034 | | -.053 | -.055 | | | | |
| 22 $\frac{1}{2}$ | | | | .272 | | .191 | .088 | | -.070 | -.124 | -.085 | -.092 | | |
| 12 | | .421 | | .437 | .357 | .287 | .132 | .044 | -.053 | -.131 | -.121 | -.081 | -.048 | |
| 3 | .651 | .658 | .594 | .590 | .586 | | .176 | | -.058 | | -.102 | -.104 | -.040 | .064 |
| 0 | | .683 | | .615 | | | .217 | .100 | | -.107 | | -.106 | | .072 |
| $\alpha=3.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .567 | .370 | .279 | .249 | .204 | .135 | .045 | .007 | -.020 | -.063 | -.072 | -.111 | -.058 | .055 |
| 33 $\frac{1}{2}$ | | | | | | .135 | .031 | | -.052 | -.063 | | | | |
| 22 $\frac{1}{2}$ | | | | .249 | | .175 | .075 | | -.074 | -.125 | -.092 | -.111 | | |
| 12 | | .370 | | .395 | .315 | .249 | .100 | .020 | -.073 | -.144 | -.132 | -.093 | -.058 | |
| 3 | .567 | .588 | .524 | .518 | .512 | | .124 | | -.084 | | -.118 | -.100 | -.050 | .055 |
| 0 | | .612 | | .540 | | | .161 | .051 | | -.131 | | -.104 | | .074 |
| $\alpha=6.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .488 | .323 | .244 | .223 | .182 | .125 | .045 | .013 | -.015 | -.064 | -.075 | -.128 | -.068 | .050 |
| 33 $\frac{1}{2}$ | | | | | | .125 | .036 | | -.051 | -.064 | | | | |
| 22 $\frac{1}{2}$ | | | | .223 | | .161 | .064 | | -.079 | -.125 | -.100 | -.128 | | |
| 12 | | .323 | | .359 | .281 | .215 | .076 | -.003 | -.093 | -.159 | -.145 | -.110 | -.068 | |
| 3 | .488 | .528 | .462 | .453 | .445 | | .082 | | -.109 | | -.131 | -.094 | -.056 | .050 |
| 0 | | .550 | | .474 | | | .112 | .014 | | -.149 | | -.096 | | .072 |
| $\alpha=9.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .417 | .280 | .215 | .201 | .165 | .120 | .045 | .019 | -.007 | -.057 | -.074 | -.135 | -.077 | .029 |
| 33 $\frac{1}{2}$ | | | | | | .120 | .045 | | -.040 | -.057 | | | | |
| 22 $\frac{1}{2}$ | | | | .201 | | .158 | .060 | | -.078 | -.118 | -.100 | -.135 | | |
| 12 | | .280 | | .331 | .255 | .190 | .060 | -.019 | -.104 | -.167 | -.151 | -.124 | -.077 | |
| 3 | .417 | .469 | .412 | .398 | .387 | | .044 | | -.127 | | -.138 | -.092 | -.067 | .029 |
| 0 | | .494 | | .416 | | | .072 | -.016 | | -.163 | | -.092 | | .056 |
| $\alpha=12.0^\circ; \beta=0.3^\circ$ | | | | | | | | | | | | | | |
| P.L. | .348 | .246 | .191 | .183 | .150 | .104 | .039 | .018 | -.007 | -.053 | -.074 | -.142 | -.097 | .019 |
| 33 $\frac{1}{2}$ | | | | | | .104 | .049 | | -.033 | -.053 | | | | |
| 22 $\frac{1}{2}$ | | | | .183 | | .156 | .057 | | -.079 | -.116 | -.102 | -.142 | | |
| 12 | | .246 | | .309 | .232 | .168 | .040 | -.035 | -.117 | -.177 | -.156 | -.147 | -.097 | |
| 3 | .348 | .409 | .368 | .349 | .337 | | .011 | | -.144 | | -.139 | -.092 | -.096 | .019 |
| 0 | | .432 | | .365 | | | .037 | -.046 | | -.170 | | -.089 | | .046 |

TABLE XVII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED SMALL FLAT-WINGED CANOPY

(a) $M=1.41$

| α/β | β, deg | -0.15 | 0 | .004 | .065 | .164 | .223 | .239 | .209 | .382 | .455 | .527 | .655 | .760 | .869 | .993 |
|------------------------------------|---------------------|-------|------|------|------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|------|
| $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | | |
| P_{oL} | | .355 | | .259 | .191 | | .112 | .042 | -.028 | -.082 | -.097 | -.011 | .020 | -.027 | -.036 | |
| 80 | | .355 | | | .185 | | | -.079 | -.047 | | -.097 | | | | | |
| 60 | | .365 | | | .260 | | .190 | -.076 | -.101 | -.110 | | .011 | | | | |
| 40 | | .467 | | .290 | .253 | | .214 | -.033 | -.147 | | -.070 | | .020 | | | |
| 32 | | | | .333 | | | | | | | | | | | | |
| 20 | | | .512 | | | .283 | .203 | | -.197 | -.186 | | .019 | | -.027 | | |
| 11 | | | | | | | .235 | | | | | | | | | |
| 0 | | | .538 | | | | .283 | .060 | -.196 | | | | .016 | | -.036 | |
| $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | | |
| P_{oL} | | .414 | | .346 | .271 | | .192 | .109 | .016 | -.049 | -.105 | -.045 | -.007 | -.044 | -.049 | |
| 80 | | .414 | | | .296 | | | .019 | -.001 | | -.105 | | | | | |
| 60 | | .433 | | | .489 | | .252 | .033 | -.044 | -.098 | | -.045 | | | | |
| 40 | | .491 | | .337 | .358 | | .291 | .028 | -.079 | | -.133 | | -.007 | | | |
| 32 | | | | .375 | | | | | | | | | | | | |
| 20 | | | .530 | | | .303 | .257 | | -.159 | -.191 | | -.022 | | -.044 | | |
| 11 | | | | | | | .256 | | | | | | | | | |
| 0 | | | .531 | | | | .266 | .050 | -.201 | | | | .003 | | -.049 | |
| -11 | | | | | | | .195 | | | | | | | | | |
| -20 | | .551 | | | | .229 | .091 | | -.236 | -.237 | | .014 | | -.039 | | |
| -32 | | | | .253 | | | | | | | | | | | | |
| -40 | | .437 | | .070 | .132 | | .112 | -.132 | -.267 | | -.051 | | .011 | | | |
| -60 | | .186 | | | .155 | | .042 | -.178 | -.179 | -.129 | | .007 | | | | |
| -80 | | .271 | | | .078 | | | -.174 | -.097 | | -.083 | | | | | |
| P_{oU} | | .271 | | .161 | .115 | | .034 | -.035 | -.099 | -.147 | -.083 | .007 | .011 | -.039 | -.045 | |

TABLE XVII. - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH NACA-44-20-100 SHALL FLAT-DOGFIELD CASEY - Continued

(a) $M=1.41$

| α/β | -0.15 | 0 | $.046$ | $.065$ | $.164$ | $.223$ | $.239$ | $.259$ | $.282$ | $.285$ | $.287$ | $.295$ | $.260$ | $.263$ | $.293$ |
|------------------------------------|---------|------|--------|--------|--------|--------|--------|--------|--------|--------|--------|--------|--------|--------|--------|
| β, deg | | | | | | | | | | | | | | | |
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | |
| P_{oL} | .465 | | .431 | .348 | | .272 | .272 | .266 | -.022 | -.125 | -.113 | -.045 | -.092 | -.093 | |
| 80 | .465 | | | .393 | | .304 | .304 | .298 | | -.125 | | | | | |
| 60 | .507 | | | .514 | | .342 | .323 | -.017 | -.097 | | -.113 | | | | |
| 40 | .429 | | .421 | .421 | | .342 | .275 | -.055 | | -.170 | | -.065 | | | |
| 20 | | | .392 | | | | | | | | | | | | |
| 11 | | .512 | | | .303 | .277 | | -.112 | -.203 | | -.097 | | -.092 | | |
| 0 | | | .502 | | | .240 | .015 | -.111 | | | | | -.012 | -.093 | |
| -11 | | | | | | .156 | | | | | | | | | |
| -20 | | .557 | | | .177 | -.135 | | -.271 | -.334 | | -.010 | | -.073 | | |
| -32 | | | .178 | | | | | | | | | | | | |
| -40 | .408 | | -.192 | .022 | | -.013 | -.183 | -.315 | | -.071 | | -.015 | | | |
| -60 | -.010 | | | .023 | | -.069 | -.061 | -.269 | -.151 | | -.075 | | | | |
| -80 | .171 | | | -.028 | | | -.252 | -.173 | | -.202 | | | | | |
| P_{oU} | .171 | | .071 | .032 | | -.011 | -.116 | -.172 | -.230 | -.202 | -.075 | -.315 | -.073 | -.290 | |
| $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | |
| P_{oL} | .291 | | .176 | .139 | | .076 | .011 | -.060 | -.115 | -.129 | -.030 | -.306 | -.012 | -.036 | |
| 80 | .291 | | | .118 | | -.055 | .004 | | -.129 | | | | | | |
| 60 | .271 | | | .172 | | .127 | -.019 | -.114 | -.173 | | -.030 | | | | |
| 40 | .321 | | .371 | .169 | | .142 | -.055 | -.169 | | -.097 | | -.006 | | | |
| 20 | | | .255 | | | | | | | | | | | | |
| 11 | | .424 | | | .170 | .125 | | -.235 | -.182 | | -.013 | | -.012 | | |
| 0 | | | .139 | | | .111 | | | | | | | | | |
| -11 | | | | | | .161 | -.009 | -.215 | | | | -.006 | | -.036 | |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | |
| P_{oL} | .321 | | .253 | .223 | | .142 | .066 | -.020 | -.092 | -.175 | -.098 | | -.051 | -.060 | |
| 80 | .322 | | .247 | .227 | | .015 | | | | -.175 | | | | | |
| 60 | .327 | | | .242 | | .210 | .056 | -.087 | -.135 | | -.098 | | | | |
| 40 | .301 | | .379 | .256 | | .145 | -.025 | -.151 | | -.211 | | | | | |
| 20 | | | .290 | | | | | | | | | | | | |
| 11 | | .405 | | | .185 | .146 | | -.206 | -.233 | | -.252 | | -.061 | | |
| 0 | | | .210 | | | .157 | | | | | | | | | |
| -11 | | | | | | .165 | -.017 | -.215 | | | | -.011 | -.060 | | |
| -20 | | .421 | | | .132 | .025 | | -.276 | -.186 | | -.026 | | -.020 | | |
| -32 | | | .182 | | | | | | | | | | | | |
| -40 | .295 | | .144 | .089 | | .053 | -.115 | -.237 | | -.073 | | | | | |
| -60 | .165 | | | .073 | | .052 | -.106 | -.157 | -.154 | | -.016 | | | | |
| -80 | .235 | | .110 | .065 | | | -.129 | | | -.088 | | | | | |
| P_{oU} | .235 | | .117 | .098 | | .018 | -.011 | -.096 | -.135 | -.068 | -.016 | | -.080 | -.099 | |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | |
| P_{oL} | .312 | | .227 | .323 | | .206 | .123 | .028 | -.060 | -.196 | -.226 | -.036 | -.117 | -.071 | |
| 80 | .242 | | | .302 | | | .079 | .119 | | -.196 | | | | | |
| 60 | .312 | | | .318 | | .265 | .111 | -.077 | -.216 | | -.226 | | | | |
| 40 | .257 | | .363 | .233 | | .218 | .000 | -.170 | | -.211 | | -.016 | | | |
| 20 | | | .301 | | | | | | | | | | | | |
| 11 | | .275 | | | .108 | .107 | | -.215 | -.279 | | -.109 | | -.117 | | |
| 0 | | | | | | .146 | | | | | | | | | |
| -11 | | .390 | | | | .129 | -.061 | -.243 | | | | -.117 | -.072 | | |
| -20 | | | .257 | | | | | | | | | | | | |
| -32 | .433 | | | .079 | -.193 | | -.220 | -.265 | | -.317 | | -.066 | | | |
| -40 | | .105 | | | | | | | | | | | | | |
| -60 | .295 | | -.045 | -.001 | | -.020 | -.222 | -.211 | | -.221 | | | | | |
| -80 | -.039 | | | -.014 | | -.036 | -.199 | -.212 | -.185 | | -.111 | | | | |
| P_{oU} | .157 | | .029 | .012 | | | -.186 | | -.216 | | | | | | |
| | .157 | | .043 | .026 | | -.012 | -.170 | -.143 | -.164 | -.219 | -.111 | -.066 | -.067 | | |

TABLE VII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY - Continued

(b) $M=2.01$

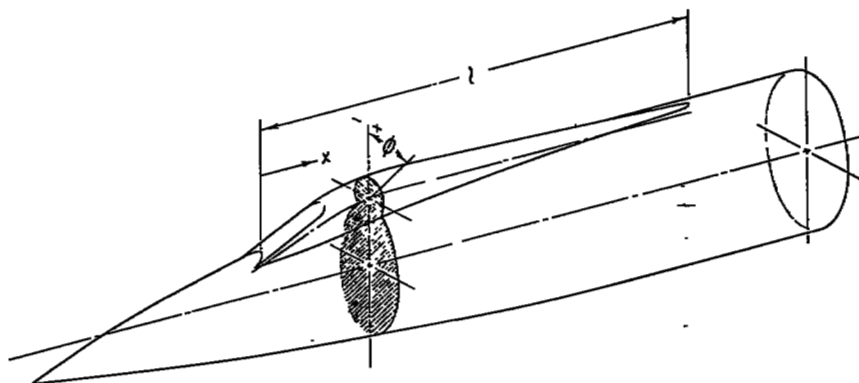
| α/β | $\alpha=0.15^\circ$ | 0 | .004 | .065 | .164 | .223 | .239 | .309 | .382 | .455 | .527 | .655 | .760 | .869 | .993 |
|---------------------|------------------------------------|------|------|-------|-------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|
| β, deg | $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | |
| P_{eL} | .276 | | | .176 | .148 | | .085 | .067 | .013 | -.022 | -.037 | -.004 | .035 | .008 | -.010 |
| 80 | .276 | | | .179 | .130 | | .070 | -.042 | .001 | | -.037 | | | | |
| 60 | | | | .241 | .206 | | .156 | .016 | -.076 | -.040 | | -.004 | | | |
| 40 | .376 | | | .254 | .241 | | .065 | -.080 | | -.042 | | | .035 | | |
| 32 | | | | .323 | | | | | | | | | | | |
| 20 | | .476 | | .379 | | .298 | .194 | .112 | -.074 | -.125 | | .024 | | .008 | |
| 11 | | | | | | | .259 | | | | | | | | |
| 0 | | | .545 | | | | .308 | .117 | -.069 | | | | .037 | | -.010 |
| | $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | |
| P_{eL} | .337 | | | .243 | .224 | | .154 | .134 | .070 | .017 | -.027 | -.032 | -.005 | -.002 | -.019 |
| 80 | .337 | | | .258 | .227 | | .157 | .046 | .049 | | -.027 | | | | |
| 60 | | | | .329 | .326 | | .244 | .078 | -.004 | -.018 | | -.032 | | | |
| 40 | .348 | | | .369 | .317 | | .117 | -.042 | | | -.038 | | -.005 | | |
| 32 | | | | .382 | | | | | | | | | | | |
| 20 | | .414 | | .405 | | .327 | .257 | .122 | -.062 | -.086 | | -.027 | | -.002 | |
| 11 | | | | | | | .282 | | | | | | | | |
| 0 | | | .533 | | | | .305 | .114 | -.069 | | | | .021 | | -.019 |
| -11 | | | | | | | .236 | | | | | | | | |
| -20 | | .510 | | .359 | | .266 | .093 | .085 | -.096 | -.148 | | .020 | | -.001 | |
| -32 | | | | .268 | | | | | | | | | | | |
| -40 | .409 | | | .074 | .154 | | | .006 | -.123 | | -.043 | | .032 | | |
| -60 | | | | .146 | .113 | | .063 | -.051 | -.132 | -.074 | | -.026 | | | |
| -80 | .205 | | | .098 | .041 | | -.012 | -.110 | -.060 | | -.099 | | | | |
| - P_{eL} | .205 | | | .106 | .080 | | .024 | -.007 | -.047 | -.078 | -.099 | -.026 | .032 | -.001 | -.020 |
| | $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | |
| P_{eL} | .392 | | | .312 | .310 | | .231 | .204 | .130 | .064 | -.011 | -.040 | -.098 | -.031 | -.048 |
| 80 | .392 | | | .331 | .329 | | .253 | .142 | .107 | | -.011 | | | | |
| 60 | | | | .405 | .459 | | .328 | .154 | .062 | -.002 | | -.040 | | | |
| 40 | .336 | | | .438 | .393 | | .169 | .004 | | -.053 | | | -.098 | | |
| 32 | | | | .424 | | | | | | | | | | | |
| 20 | | .349 | | .426 | | .350 | .300 | .148 | -.064 | -.085 | | -.123 | | -.031 | |
| 11 | | | | | | | .304 | | | | | | | | |
| 0 | | | .492 | | | | .308 | .054 | -.092 | | | | -.028 | | -.048 |
| -11 | | | | | | | .232 | | | | | | | | |
| -20 | | .525 | | .365 | | .255 | -.039 | .011 | -.119 | -.169 | | -.043 | | -.045 | |
| -32 | | | | .253 | | | | | | | | | | | |
| -40 | .433 | | | -.113 | -.051 | | | -.048 | -.166 | | -.064 | | -.023 | | |
| -60 | | | | .060 | .024 | | -.020 | -.119 | -.155 | -.118 | | -.018 | | | |
| -80 | .091 | | | .024 | -.046 | | -.099 | -.145 | -.135 | | -.168 | | | | |
| - P_{eL} | .091 | | | .045 | .015 | | -.039 | -.091 | -.118 | -.153 | -.168 | -.018 | -.023 | -.045 | -.055 |

TABLE XVII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH FORWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

| x/l deg | -0.015 | 0 | $.004$ | $.065$ | $.164$ | $.223$ | $.239$ | $.309$ | $.382$ | $.455$ | $.527$ | $.655$ | $.760$ | $.869$ | $.993$ |
|------------------------------------|----------|------|--------|--------|--------|--------|--------|--------|--------|--------|--------|--------|--------|--------|--------|
| $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | | |
| F.L. | .237 | | | .140 | | | .066 | .045 | -.007 | -.050 | -.087 | -.024 | .007 | -.012 | -.018 |
| 80 | .237 | | | .153 | .134 | | .074 | -.011 | -.024 | | -.087 | | | | |
| 60 | | | | .200 | .164 | | .136 | -.004 | -.060 | -.076 | | -.024 | | | |
| 40 | .257 | | | .223 | .179 | | | .024 | -.101 | | -.076 | | .007 | | |
| 32 | | | | .259 | | | | | | | | | | | |
| 20 | | .341 | | .270 | | .192 | .118 | .040 | -.114 | -.122 | | -.001 | | -.012 | |
| 11 | | | | | | | .156 | | | | | | | | |
| 0 | | | .409 | | | | .198 | .034 | -.114 | | | | .009 | | -.015 |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | | |
| F.L. | .291 | | | .185 | | | .122 | .092 | .036 | -.022 | -.097 | -.104 | -.023 | -.034 | -.042 |
| 80 | .291 | | | .203 | .197 | | .137 | .051 | .008 | | -.097 | | | | |
| 60 | | | | .255 | .253 | | .187 | .047 | -.032 | -.087 | | -.104 | | | |
| 40 | .228 | | | .292 | .230 | | | .053 | -.075 | | -.104 | | -.023 | | |
| 32 | | | | .317 | | | | | | | | | | | |
| 20 | | .274 | | .287 | | .210 | .161 | .037 | -.112 | -.126 | | -.047 | | -.034 | |
| 11 | | | | | | | .174 | | | | | | | | |
| 0 | | | .379 | | | | .191 | .022 | -.122 | | | | -.033 | | -.042 |
| -11 | | | | | | | .135 | | | | | | | | |
| -20 | | .375 | | .248 | | .160 | .019 | .010 | -.141 | -.164 | | -.014 | | -.057 | |
| -32 | | | | .194 | | | | | | | | | | | |
| -40 | .284 | | | .089 | .098 | | | -.028 | -.145 | | -.062 | | -.023 | | |
| -60 | | | | .122 | .083 | | .054 | -.064 | -.115 | -.094 | | -.010 | | | |
| -80 | .161 | | | .092 | .059 | | -.001 | -.079 | -.060 | | -.087 | | | | |
| -F.L. | .161 | | | .089 | | | .016 | -.006 | -.046 | -.076 | -.087 | -.010 | -.023 | -.057 | -.038 |
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | | |
| F.L. | .344 | | | .242 | | | .189 | .152 | .082 | .010 | -.079 | -.117 | -.048 | -.064 | -.067 |
| 80 | .344 | | | .252 | .273 | | .212 | .120 | .050 | | -.079 | | | | |
| 60 | | | | .300 | .337 | | .257 | .102 | .001 | -.075 | | -.117 | | | |
| 40 | .199 | | | .337 | .283 | | | .082 | -.050 | | -.140 | | -.048 | | |
| 32 | | | | .329 | | | | | | | | | | | |
| 20 | | .202 | | .302 | | .218 | .185 | .055 | -.120 | -.151 | | -.116 | | -.064 | |
| 11 | | | | | | | .186 | | | | | | | | |
| 0 | | | .330 | | | | .186 | -.032 | -.149 | | | | -.093 | | -.067 |
| -11 | | | | | | | .129 | | | | | | | | |
| -20 | | .372 | | .238 | | .140 | -.176 | -.082 | -.174 | -.195 | | -.112 | | -.070 | |
| -32 | | | | .146 | | | | | | | | | | | |
| -40 | .304 | | | -.091 | -.082 | | | -.093 | -.193 | | -.095 | | -.148 | | |
| -60 | | | | .046 | .002 | | -.027 | -.126 | -.164 | -.150 | | -.110 | | | |
| -80 | .047 | | | .029 | -.011 | | -.066 | -.128 | -.114 | | -.118 | | | | |
| -F.L. | .047 | | | .036 | | | -.033 | -.059 | -.088 | -.109 | -.118 | -.110 | -.148 | -.070 | -.078 |

TABLE XVIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED SMALL FLARE-WINGED CANOPY

(a) $M=1.41$

| x/l | .011 | 0 | .004 | .052 | .102 | .142 | .149 | .193 | .249 | .324 | .467 | .668 | .804 | .985 |
|------------------------|------------------------------------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|------|
| θ , deg | $\alpha=0.4^\circ; \beta=0^\circ$ | | | | | | | | | | | | | |
| $C_{p, \text{top}}$ | .296 | .285 | .152 | .121 | .071 | | -.005 | -.031 | -.060 | -.111 | -.060 | -.066 | -.022 | |
| 80 | | | .152 | | | | -.038 | | -.068 | -.111 | | | | |
| 60 | .296 | | .159 | .120 | .073 | | | -.108 | | -.104 | -.060 | | | |
| 48 | | .356 | | | | | | | | | | | | |
| 43 | .388 | | .242 | .166 | .105 | | | | -.153 | -.054 | | -.066 | | |
| 40 | | .447 | | | | | | | | | | | | |
| 32 | | | .326 | | | | | | | | | | | |
| 29 | | | | | .148 | | | | | | | | | |
| 26 | | | | .271 | | | | | | | | | | |
| 20 | | .516 | .338 | | .216 | .100 | | -.072 | -.202 | -.070 | -.040 | | | |
| 0 | | | .529 | .354 | .308 | .225 | .089 | -.049 | -.224 | -.063 | -.040 | -.044 | -.022 | |
| $C_{p, \text{bottom}}$ | $\alpha=0.4^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | |
| 80 | .333 | .337 | .245 | .211 | .150 | | .058 | .032 | .017 | -.147 | -.093 | -.098 | -.042 | |
| 60 | | | .245 | | | | .050 | | -.028 | -.147 | | | | |
| 48 | .333 | | .265 | .224 | .173 | | | -.031 | | -.173 | -.093 | | | |
| 43 | | .408 | | | | | | | | | | | | |
| 40 | .373 | | .336 | .265 | .201 | | | | -.126 | -.197 | | -.096 | | |
| 32 | | .450 | | | | | | | | | | | | |
| 29 | | | .368 | | | | | | | | | | | |
| 26 | | | | .238 | | | | | | | | | | |
| 20 | | .488 | .351 | | .263 | .180 | | -.056 | -.197 | -.135 | -.068 | | | |
| 0 | | .514 | .346 | .305 | .227 | .049 | -.126 | -.239 | -.099 | -.064 | -.068 | -.042 | | |
| -20 | | .515 | .344 | | .150 | -.034 | | -.122 | -.278 | -.086 | -.063 | | | |
| -26 | | | | .191 | | | | | | | | | | |
| -29 | | | | | -.006 | | | | | | | | | |
| -42 | | | .262 | | | | | | | | | | | |
| -40 | | .467 | | | | | | | | | | | | |
| -43 | .370 | | .096 | .017 | -.031 | | | | -.215 | -.080 | | -.088 | | |
| -48 | | .242 | | | | | | | | | | | | |
| -60 | .229 | | .024 | -.015 | -.054 | | | -.223 | | -.080 | -.071 | | | |
| -80 | | | .046 | | | | -.133 | | -.119 | -.062 | | | | |
| -94.6 | .229 | .201 | .045 | .022 | -.012 | | -.062 | -.100 | -.121 | -.082 | -.071 | -.068 | -.037 | |

TABLE VIII.- EFFECTIVE COEFFICIENTS FOR CONFIGURATIONS WITH REARWARD-LOCATED SMALL FLAT-WINGED SHIELD CANALS - Continued

| (a) $M=141$ | | | | | | | | | | | | | | |
|------------------------------------|------------------|-----------------|------------------|-----------------|------------------|-----------------|------------------|-----------------|------------------|-----------------|------------------|-----------------|------------------|-----------------|
| α/β | $\alpha=0^\circ$ | $\beta=0^\circ$ | $\alpha=0^\circ$ | $\beta=0^\circ$ | $\alpha=0^\circ$ | $\beta=0^\circ$ | $\alpha=0^\circ$ | $\beta=0^\circ$ | $\alpha=0^\circ$ | $\beta=0^\circ$ | $\alpha=0^\circ$ | $\beta=0^\circ$ | $\alpha=0^\circ$ | $\beta=0^\circ$ |
| $\alpha=0.4^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | |
| $P_{0.1}$ | .355 | .377 | .336 | .307 | .276 | .178 | .106 | .034 | -.140 | -.132 | -.114 | -.089 | | |
| 80 | | | .336 | | | .143 | | .014 | -.140 | | | | | |
| 60 | .355 | | .357 | .228 | .203 | | .044 | | -.157 | -.132 | | | | |
| 40 | | .116 | | | | | | | | | | | | |
| 20 | .336 | | .392 | .338 | .276 | | | -.115 | -.267 | | -.144 | | | |
| 10 | | .163 | | | | | | | | | | | | |
| 32 | | | .391 | | | | | | | | | | | |
| 29 | | | | | .287 | | | | | | | | | |
| 26 | | | | | .342 | | | | | | | | | |
| 20 | | .431 | .360 | .290 | .225 | -.051 | -.218 | -.337 | -.214 | | | | | |
| 0 | | .440 | .378 | .298 | .223 | .024 | -.123 | -.288 | -.172 | -.110 | -.111 | -.089 | | |
| -20 | .195 | | .214 | .096 | -.152 | | -.120 | -.377 | -.190 | -.127 | | | | |
| -46 | | | .121 | | | | | | | | | | | |
| -69 | | | | | -.235 | | | | | | | | | |
| -82 | | | .290 | | | | | | | | | | | |
| -10 | | .354 | | | | | | | | | | | | |
| -13 | .345 | | -.092 | -.155 | -.211 | | | -.313 | -.167 | | -.128 | | | |
| -18 | | .086 | | | | | | | | | | | | |
| -40 | .128 | | -.128 | -.183 | -.197 | | -.360 | | -.202 | -.216 | | | | |
| -60 | | | -.068 | | | -.212 | | -.191 | -.114 | | | | | |
| -70 | .128 | .098 | -.062 | -.073 | -.101 | -.166 | -.191 | -.217 | -.111 | -.238 | -.128 | -.074 | | |
| $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | | | |
| $P_{0.1}$ | .227 | .220 | .125 | .104 | .060 | -.009 | -.030 | -.026 | -.143 | -.140 | -.070 | -.014 | | |
| 80 | | | .125 | | | -.026 | | -.036 | -.143 | | | | | |
| 60 | .227 | | .130 | .103 | .062 | | -.107 | | | -.140 | | | | |
| 40 | | .260 | | | | | | | | | | | | |
| 20 | .290 | | .195 | .123 | .077 | | | -.179 | -.123 | | -.070 | | | |
| 10 | | .340 | | | | | | | | | | | | |
| 32 | | | .216 | | | | | | | | | | | |
| 29 | | | | | .161 | | | | | | | | | |
| 26 | | | | | .124 | | | | | | | | | |
| 20 | .392 | | .252 | | .112 | .045 | -.095 | -.220 | -.306 | -.054 | | | | |
| 0 | | .165 | .264 | .214 | .139 | .032 | -.119 | -.226 | -.077 | -.090 | -.039 | -.014 | | |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | | | |
| $P_{0.1}$ | .240 | .260 | .203 | .185 | .132 | .037 | .009 | -.045 | -.1805 | -.253 | -.093 | -.027 | | |
| 80 | | | .203 | | | .045 | | -.072 | -.1805 | | | | | |
| 60 | .215 | | .223 | .146 | .143 | | -.051 | | -.253 | | | | | |
| 40 | | .314 | | | | | | | | | | | | |
| 20 | .266 | | .254 | .199 | .151 | | | -.179 | -.262 | | -.093 | | | |
| 10 | | .332 | | | | | | | | | | | | |
| 32 | | | .278 | | | | | | | | | | | |
| 29 | | | | | .167 | | | | | | | | | |
| 26 | | | | | .233 | | | | | | | | | |
| 20 | .370 | | .250 | | .152 | .113 | -.065 | -.214 | -.152 | -.087 | | | | |
| 0 | | .195 | .257 | .211 | .140 | -.052 | -.167 | -.271 | -.111 | -.128 | -.067 | -.027 | | |
| -20 | .395 | | .227 | | .074 | -.075 | -.135 | -.256 | -.098 | -.091 | | | | |
| -46 | | | .229 | | | | | | | | | | | |
| -69 | | | | | -.033 | | | | | | | | | |
| -82 | | | .187 | | | | | | | | | | | |
| -10 | | .313 | | | | | | | | | | | | |
| -13 | .278 | | .065 | -.000 | -.036 | | | -.204 | -.171 | | -.095 | | | |
| -18 | | .195 | | | | | | | | | | | | |
| -40 | .176 | | .052 | -.006 | -.039 | | -.184 | | -.094 | | | | | |
| -60 | | | .036 | | | -.106 | | -.129 | -.058 | | | | | |
| -70 | .176 | .155 | .030 | .015 | -.023 | -.088 | -.103 | -.121 | -.198 | -.091 | -.095 | -.023 | | |

TABLE XVII - PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REAR-ARC-LOCATED SMALL FLAG-NITROGEN CLOUD - Cont'd

[illegible]

TABLE XVIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED SMALL FLARE-TRIGGERED CANOPY - Continued

| (b) $M=2.01$ | | | | | | | | | | | | |
|------------------------------------|------------------|------------------|------------------|-------------------|-------------------|-------------------|-------------------|-------------------|-------------------|-------------------|-------------------|-------------------|
| α/β | $\alpha=0^\circ$ | $\alpha=4^\circ$ | $\alpha=8^\circ$ | $\alpha=12^\circ$ | $\alpha=16^\circ$ | $\alpha=20^\circ$ | $\alpha=24^\circ$ | $\alpha=28^\circ$ | $\alpha=32^\circ$ | $\alpha=36^\circ$ | $\alpha=40^\circ$ | $\alpha=44^\circ$ |
| $\alpha=0^\circ; \beta=-8^\circ$ | | | | | | | | | | | | |
| P.L. | .301 | .334 | .286 | .280 | .235 | .165 | .136 | .105 | .039 | -.076 | -.091 | -.083 |
| 80 | | | .286 | | | .262 | | .088 | .039 | | | |
| 60 | .301 | | .323 | .302 | .277 | | .103 | | .116 | -.076 | | |
| 40 | | .368 | | | | | | | | | | |
| 20 | .306 | | .371 | .359 | .297 | | | -.016 | -.116 | | -.091 | |
| 0 | | .334 | | | | | | | | | | |
| -20 | | | .399 | | | | | | | | | |
| -40 | | | | .309 | | | | | | | | |
| -60 | | | | .371 | | | | | | | | |
| -80 | .347 | .393 | | .326 | .263 | | -.012 | -.094 | -.166 | -.105 | | |
| -100 | .347 | .380 | .341 | .306 | .235 | -.031 | -.143 | -.169 | -.097 | -.076 | -.083 | |
| -120 | .442 | .341 | .293 | .230 | | -.082 | -.184 | -.218 | | -.102 | | |
| -140 | | | .207 | | | | | | | | | |
| -160 | | | | -.093 | | | | | | | | |
| -180 | | | .260 | | | | | | | | | |
| -200 | | .343 | | | | | | | | | | |
| -220 | .252 | .004 | -.030 | -.071 | | | -.185 | -.109 | | -.103 | | |
| -240 | | .094 | | | | | | | | | | |
| -260 | .094 | -.035 | -.102 | -.128 | | -.193 | | -.112 | -.142 | | | |
| -280 | | -.080 | | | | -.130 | | -.169 | -.091 | | | |
| -300 | .094 | .072 | -.020 | -.036 | -.049 | -.087 | -.116 | -.127 | -.091 | -.142 | -.103 | -.076 |
| $\alpha=6.5^\circ; \beta=0^\circ$ | | | | | | | | | | | | |
| P.L. | .161 | .164 | .108 | .094 | .062 | .018 | -.007 | -.026 | -.109 | -.116 | -.052 | -.010 |
| 80 | | | .108 | | | -.003 | | -.037 | -.109 | | | |
| 60 | .161 | | .133 | .095 | .077 | | -.053 | | -.052 | -.116 | | |
| 40 | | .211 | | | | | | | | | | |
| 20 | .216 | | .194 | .140 | .093 | | | -.101 | -.102 | | -.052 | |
| 0 | | .267 | | | | | | | | | | |
| -20 | | | .256 | | | | | | | | | |
| -40 | | | | .110 | | | | | | | | |
| -60 | | | | .200 | | | | | | | | |
| -80 | .325 | .268 | | .153 | .079 | .002 | -.137 | -.062 | -.038 | | | |
| -100 | .362 | .280 | .229 | .197 | .035 | -.061 | -.151 | -.049 | -.033 | -.028 | -.010 | |
| $\alpha=6.5^\circ; \beta=-4^\circ$ | | | | | | | | | | | | |
| P.L. | .247 | .211 | .168 | .162 | .126 | .073 | .062 | .019 | -.094 | -.136 | -.072 | -.023 |
| 80 | | | .162 | | | .060 | | .002 | -.094 | | | |
| 60 | .247 | | .201 | .175 | .153 | | .009 | | -.082 | -.116 | | |
| 40 | | .213 | | | | | | | | | | |
| 20 | .210 | | .251 | .217 | .166 | | | -.083 | -.164 | | -.072 | |
| 0 | | .247 | | | | | | | | | | |
| -20 | | | .222 | | | | | | | | | |
| -40 | | | | .160 | | | | | | | | |
| -60 | | | | .240 | | | | | | | | |
| -80 | .204 | .276 | | .199 | .141 | -.005 | -.110 | -.152 | -.082 | | | |
| -100 | .311 | .277 | .228 | .197 | .011 | -.076 | -.163 | -.083 | -.107 | -.058 | -.023 | |
| -120 | .336 | .255 | | .173 | .005 | -.013 | -.167 | -.066 | -.071 | | | |
| -140 | | | .151 | | | | | | | | | |
| -160 | | | | .002 | | | | | | | | |
| -180 | | .212 | | | | | | | | | | |
| -200 | | .261 | | | | | | | | | | |
| -220 | .205 | .100 | .042 | .001 | | | -.133 | -.067 | | -.058 | | |
| -240 | | .119 | | | | | | | | | | |
| -260 | .110 | .047 | .008 | -.006 | | -.123 | | -.044 | -.075 | | | |
| -280 | | .042 | | | | -.065 | | -.071 | -.094 | | | |
| -300 | .130 | .108 | .042 | .005 | -.033 | -.053 | -.065 | -.094 | -.075 | -.058 | -.023 | |

TABLE XVIII.- PRESSURE COEFFICIENTS FOR CONFIGURATION WITH REARWARD-LOCATED SMALL FLAT-WINDSHIELD CANOPY - Concluded

(b) $M=2.01$

| x/l ϕ, deg | -.011 | 0 | .004 | .052 | .102 | .142 | .149 | .193 | .249 | .324 | .467 | .662 | .804 | .985 |
|------------------------------------|-------|------|------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|
| $\alpha=6.5^\circ; \beta=-8^\circ$ | | | | | | | | | | | | | | |
| P.L. | .200 | | .247 | .224 | .228 | .188 | | .126 | .091 | .066 | -.066 | -.124 | -.107 | -.046 |
| 80 | | | | .224 | | | | .128 | | .043 | -.066 | | | |
| 60 | .200 | | | .242 | .232 | .214 | | .058 | | | -.125 | -.124 | | |
| 48 | | | .258 | | | | | | | | | | | |
| 43 | .181 | | | .256 | .264 | .215 | | | | -.063 | -.177 | | -.107 | |
| 40 | | | .200 | | | | | | | | | | | |
| 32 | | | | .271 | | | | | | | | | | |
| 29 | | | | | | .207 | | | | | | | | |
| 26 | | | | | .255 | | | | | | | | | |
| 20 | | .223 | | .275 | | .218 | .172 | | .002 | -.139 | -.187 | -.106 | | |
| 0 | | | .286 | .274 | .231 | | .199 | .055 | -.096 | -.190 | -.170 | -.145 | -.078 | -.046 |
| -20 | | .307 | | .234 | | .101 | -.031 | | -.047 | -.212 | -.164 | -.176 | | |
| -26 | | | | | .120 | | | | | | | | | |
| -29 | | | | | | -.143 | | | | | | | | |
| -32 | | | | .171 | | | | | | | | | | |
| -40 | | | .241 | | | | | | | | | | | |
| -43 | .165 | | | -.042 | -.079 | -.109 | | | | -.194 | -.181 | | -.095 | |
| -48 | | | .050 | | | | | | | | | | | |
| -60 | .047 | | | -.057 | -.105 | -.119 | | | -.187 | | -.113 | -.113 | | |
| -80 | | | | -.035 | | | | -.119 | | -.114 | -.079 | | | |
| -P.L. | .047 | | .037 | -.035 | -.043 | -.055 | | -.088 | -.107 | -.112 | -.079 | -.113 | -.095 | -.046 |

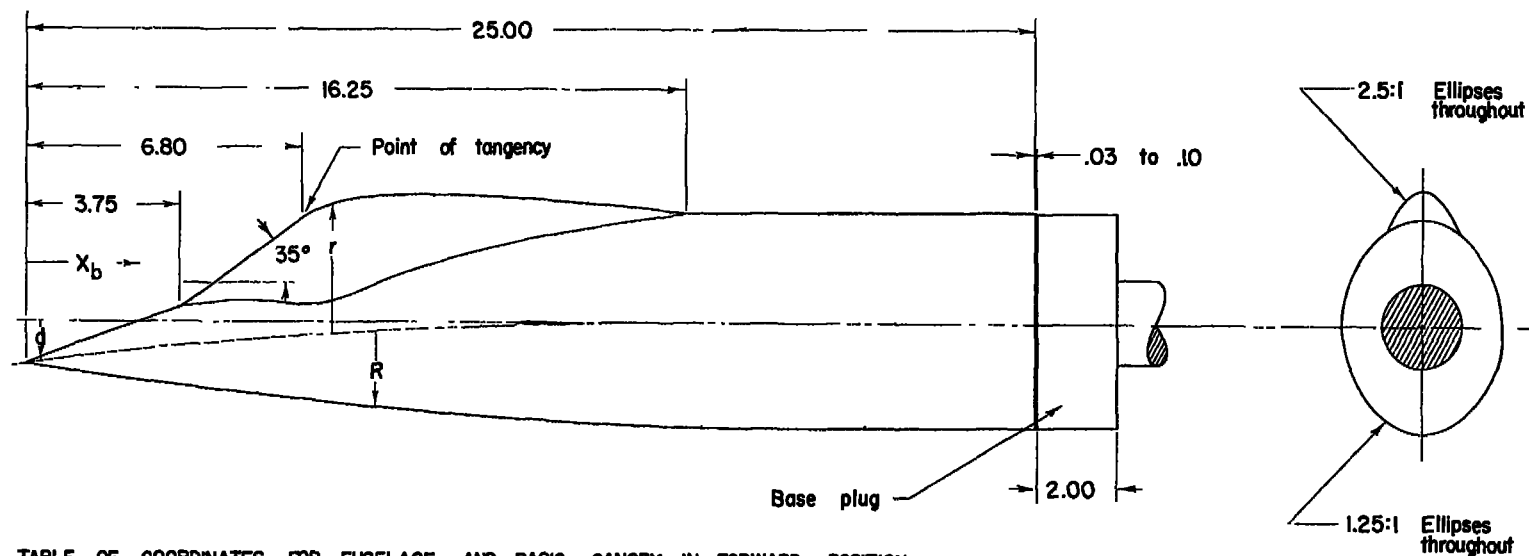


TABLE OF COORDINATES FOR FUSELAGE AND BASIC CANOPY IN FORWARD POSITION

| Fuselage Station X_b | Droop d | Fuselage Major radius R | Canopy Major radius r | X_b | d | R | r |
|---------------------------|--------------|------------------------------|----------------------------|--------|------|-------|-------|
| .000 | 1.000 | .000 | | 10.000 | .183 | 2.049 | 3.148 |
| 1.500 | .837 | .418 | | 11.250 | .127 | 2.187 | 3.077 |
| 2.500 | .735 | .674 | | 12.500 | .081 | 2.300 | 2.961 |
| 3.750 | .617 | .969 | .969 | 13.750 | .046 | 2.388 | 2.811 |
| 5.000 | .510 | 1.237 | 1.737 | 15.000 | .020 | 2.450 | 2.655 |
| 6.250 | .413 | 1.479 | 2.516 | 16.250 | .005 | 2.480 | 2.480 |
| 7.500 | .326 | 1.695 | 3.096 | 17.500 | .000 | 2.500 | |
| 8.750 | .250 | 1.885 | 3.175 | 25.000 | .000 | 2.500 | |

Figure 1.- Details of canopy-fuselage model showing round-windshield canopy in the forward location. All dimensions are in inches.

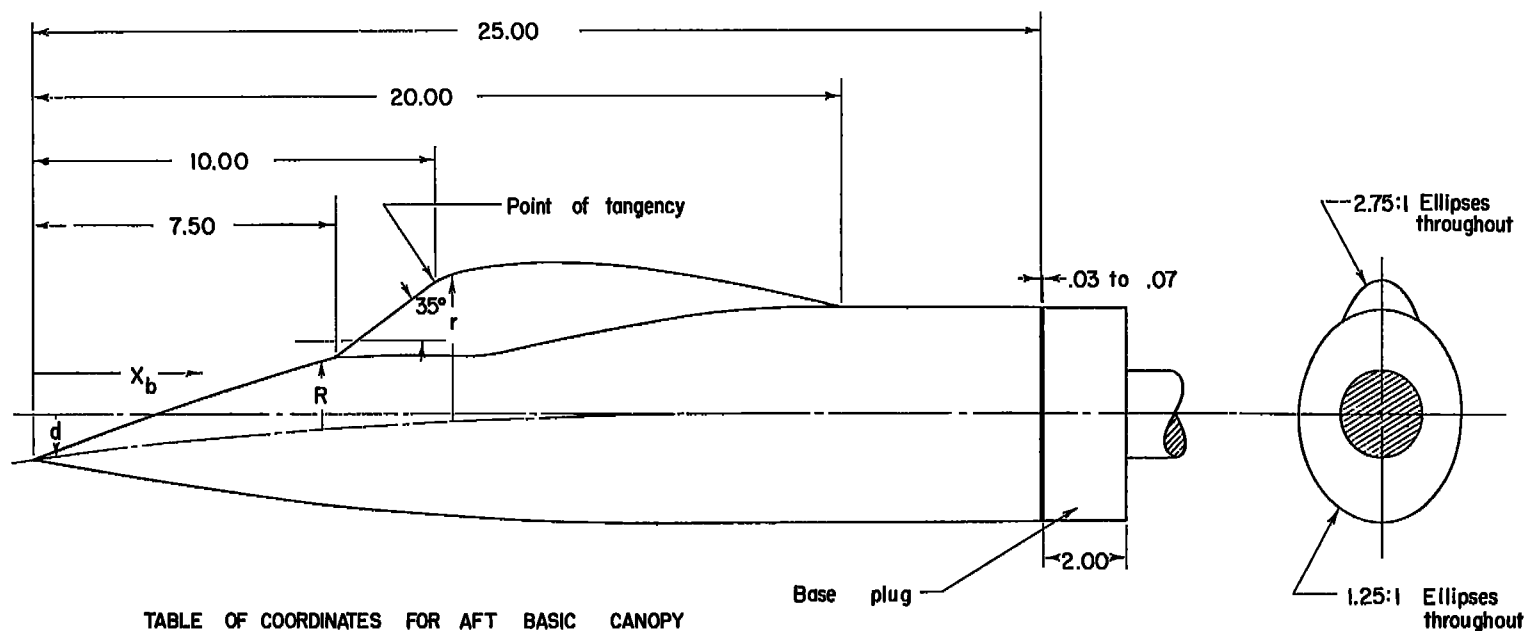


TABLE OF COORDINATES FOR AFT BASIC CANOPY

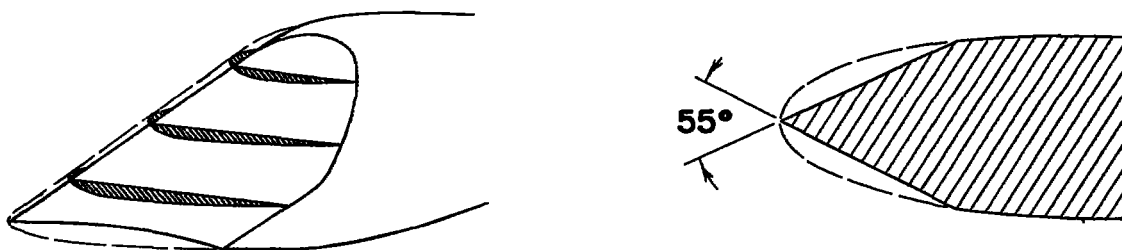
| Fuselage Station, X_b | Droop d | Fuselage Major radius R | Canopy Major radius r | X_b | d | R | r |
|-------------------------|-----------|---------------------------|-------------------------|--------|------|-------|-------|
| .000 | 1.000 | .000 | | 12.500 | .081 | 2.300 | 3.609 |
| 1.500 | .837 | .418 | | 13.750 | .046 | 2.388 | 3.511 |
| 2.500 | .735 | .674 | | 15.000 | .020 | 2.450 | 3.370 |
| 3.750 | .617 | .969 | | 16.250 | .005 | 2.480 | 3.180 |
| 5.000 | .510 | 1.237 | | 17.500 | .000 | 2.500 | 2.970 |
| 6.250 | .413 | 1.479 | | 18.750 | .000 | 2.500 | 2.750 |
| 7.500 | .326 | 1.695 | 1.695 | 20.000 | .000 | 2.500 | 2.500 |
| 8.750 | .250 | 1.885 | 2.494 | 25.000 | .000 | 2.500 | |
| 10.000 | .183 | 2.049 | 3.302 | | | | |
| 11.250 | .127 | 2.187 | 3.602 | | | | |

Figure 2.- Details of canopy-fuselage model showing round-windshield canopy in the rearward location. All dimensions are in inches.



Typical section in X-Y plane

(a) Method of development of flat-faced canopies from basic or round-faced canopies.



Typical section in X-Y plane

(b) Method of development of vee-faced canopies from basic or round-faced canopies.

Figure 3.- Method of development of flat and vee-windshield canopies from the basic or round-windshield canopy. All dimensions are in inches.

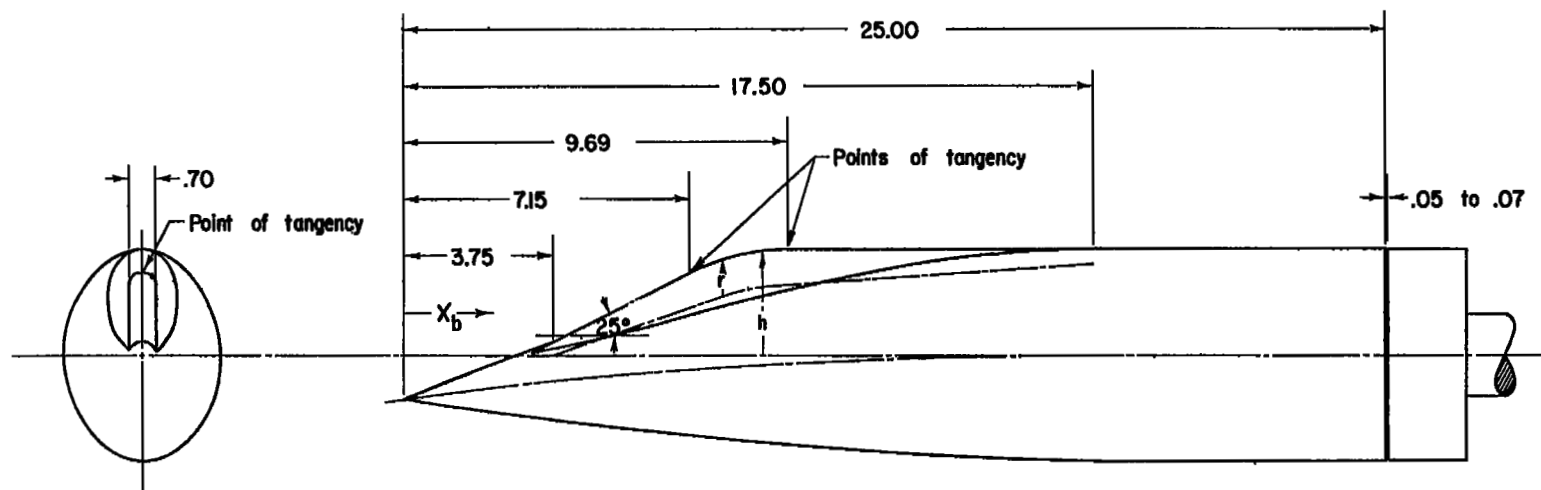


TABLE OF DIMENSIONS FOR SMALL CANOPY IN FORWARD POSITION

| Fuselage Station X_b | Canopy Profile h | Canopy Radius r | Fuselage Station X_b | Canopy Profile h | Canopy Radius r |
|---------------------------|-----------------------|----------------------|---------------------------|-----------------------|----------------------|
| 3.750 | .325 | .352 | 11.250 | 2.500 | .733 |
| 5.000 | .935 | .529 | 12.500 | 2.500 | .857 |
| 6.250 | 1.518 | .706 | 13.750 | 2.500 | .581 |
| 7.500 | 2.088 | .870 | 15.000 | 2.500 | .504 |
| 8.750 | 2.427 | .849 | 16.250 | 2.500 | .428 |
| 10.000 | 2.500 | .809 | 17.500 | 2.500 | .352 |

Note:

Dimensions from table describe a round-faced canopy. Flat-faced configuration results from addition of the flat face and fitting to fair to basic shape in a manner similar to that shown in figure 3

Figure 4.- Details of canopy-fuselage model showing small forward-located flat-windshield canopy.
All dimensions are in inches.

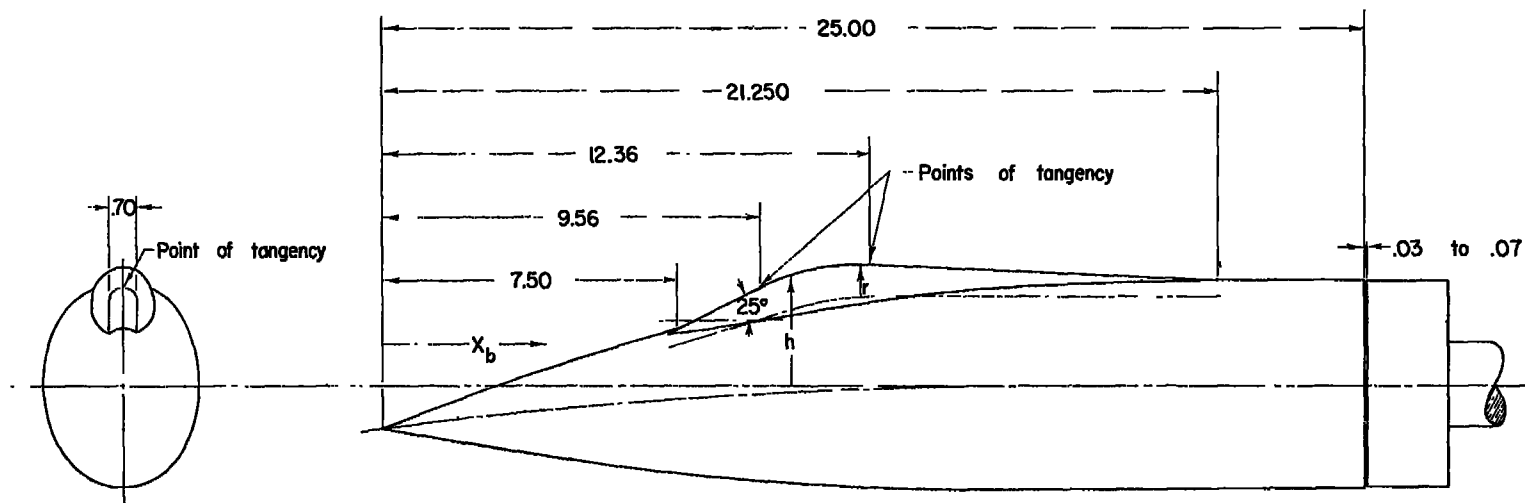


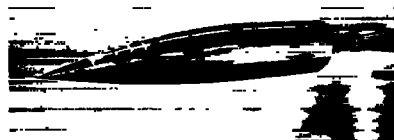
TABLE OF DIMENSIONS FOR SMALL CANOPY IN AFT POSITION

| Fuselage Station X_b | Canopy Profile h | Canopy Radius r | Fuselage Station X_b | Canopy Profile h | Canopy Radius r |
|---------------------------|-----------------------|----------------------|---------------------------|-----------------------|----------------------|
| 7.500 | 1.395 | .390 | 15.000 | 2.773 | .629 |
| 8.750 | 1.952 | .592 | 16.250 | 2.718 | .574 |
| 10.000 | 2.514 | .772 | 17.500 | 2.664 | .520 |
| 11.250 | 2.833 | .758 | 19.375 | 2.582 | .438 |
| 12.500 | 2.882 | .738 | 21.250 | 2.500 | .356 |
| 13.750 | 2.827 | .683 | | | |

Note:

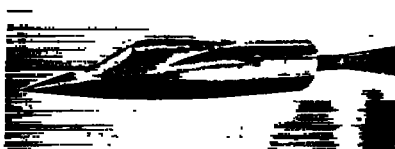
Dimensions from table describe a round-faced canopy. Flat-faced configuration results from addition of the flat face and filling to fair to the basic shape in a manner similar to that shown in figure 3

Figure 5.- Details of canopy-fuselage model showing small rearward-located flat-windshield canopy.
All dimensions are in inches.

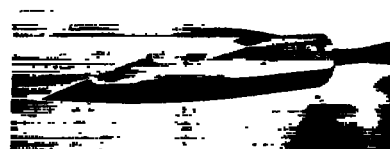


(a) Body alone.

Forward-located canopies



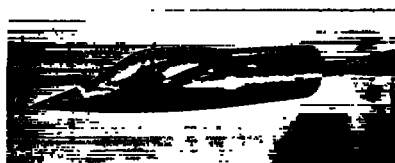
Rearward-located canopies



(b) Large flat-windshield configurations.



(c) Large vee-windshield configurations.



(d) Large round-windshield configurations.



(e) Small flat-windshield configurations.

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Figure 6.- Photographs of models.

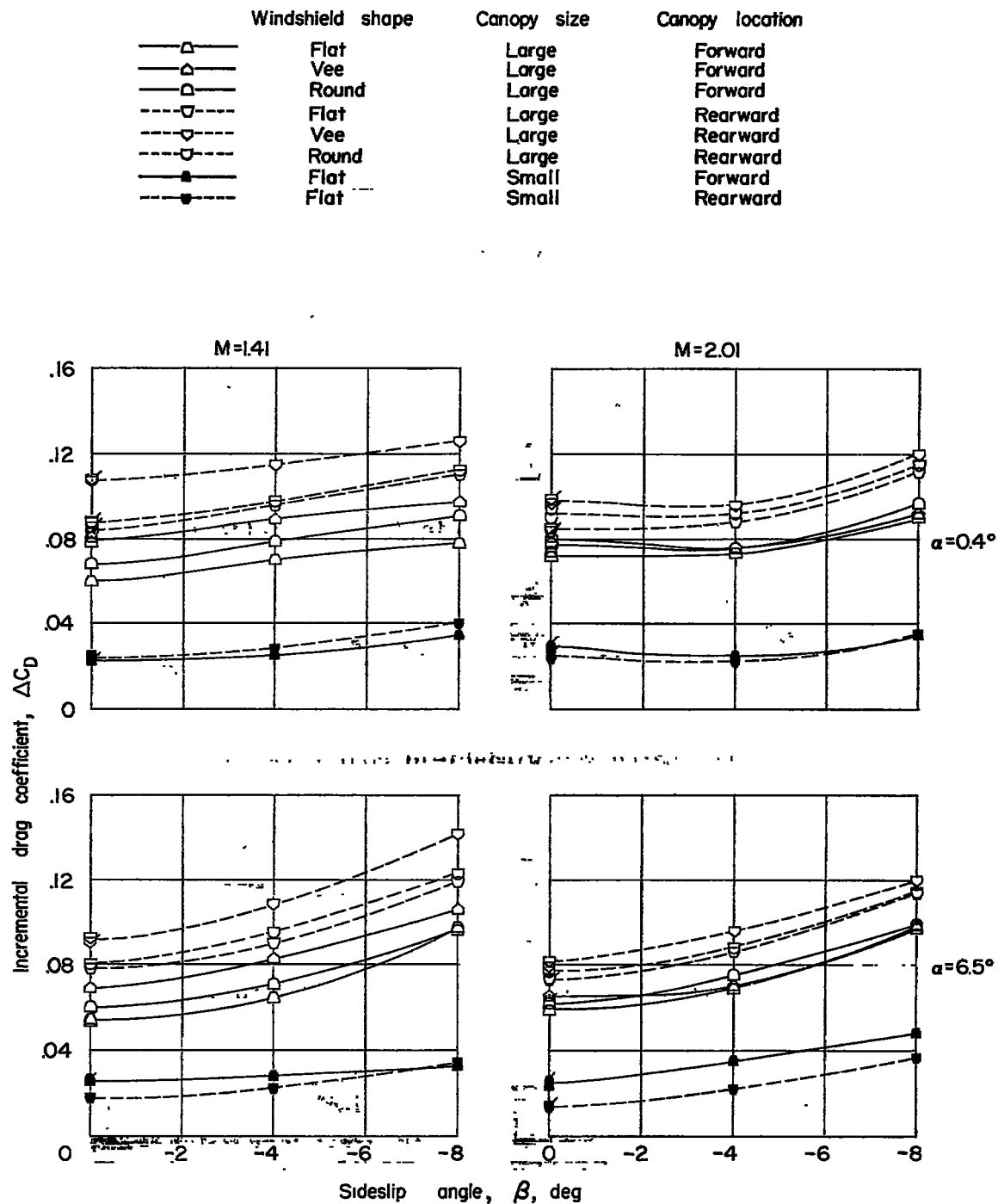


Figure 7.- Incremental drag coefficients for the several canopy configurations at various angles of sideslip for $M = 1.41$ and 2.01 and $\alpha = 0.4^\circ$ and 6.5° . Tailed symbols are check points.

- \triangle Large forward-located canopy
 ∇ Large rearward-located canopy
 \blacktriangle Small forward-located canopy
 \blacktriangledown Small rearward-located canopy
- $M=1.41$
 present tests

— Bodies of revolution; $M=1.40$; ref. 7

K = Location of station of maximum cross-section area , percent of length

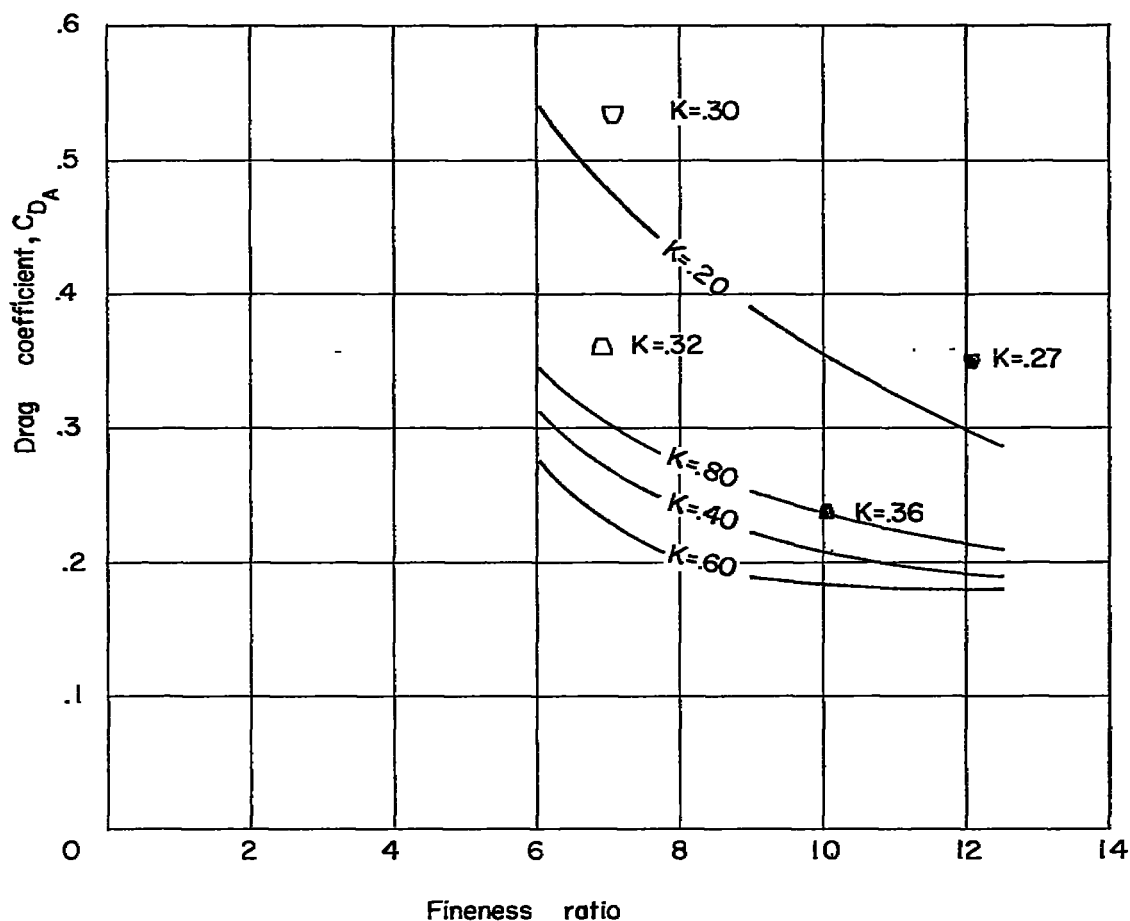
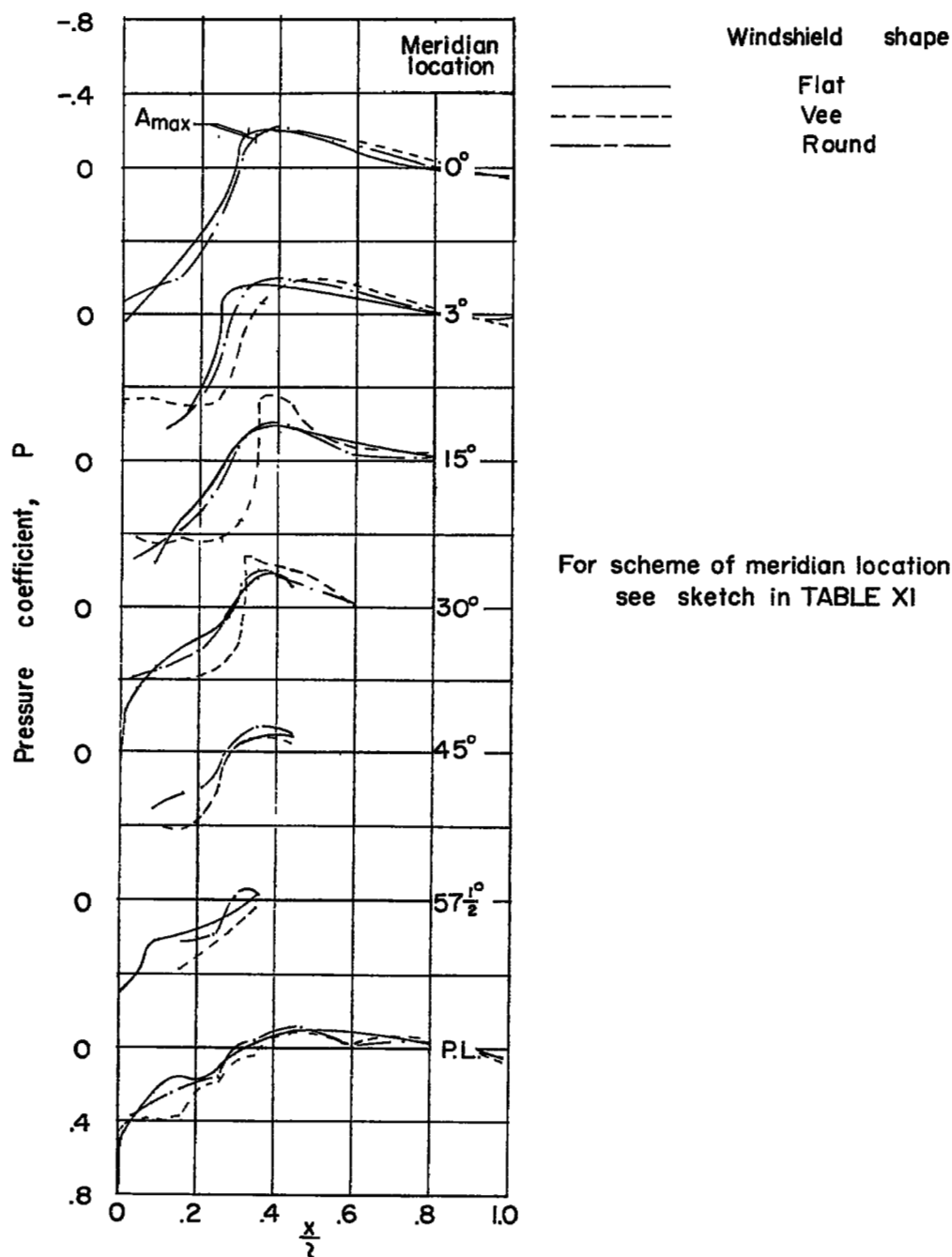
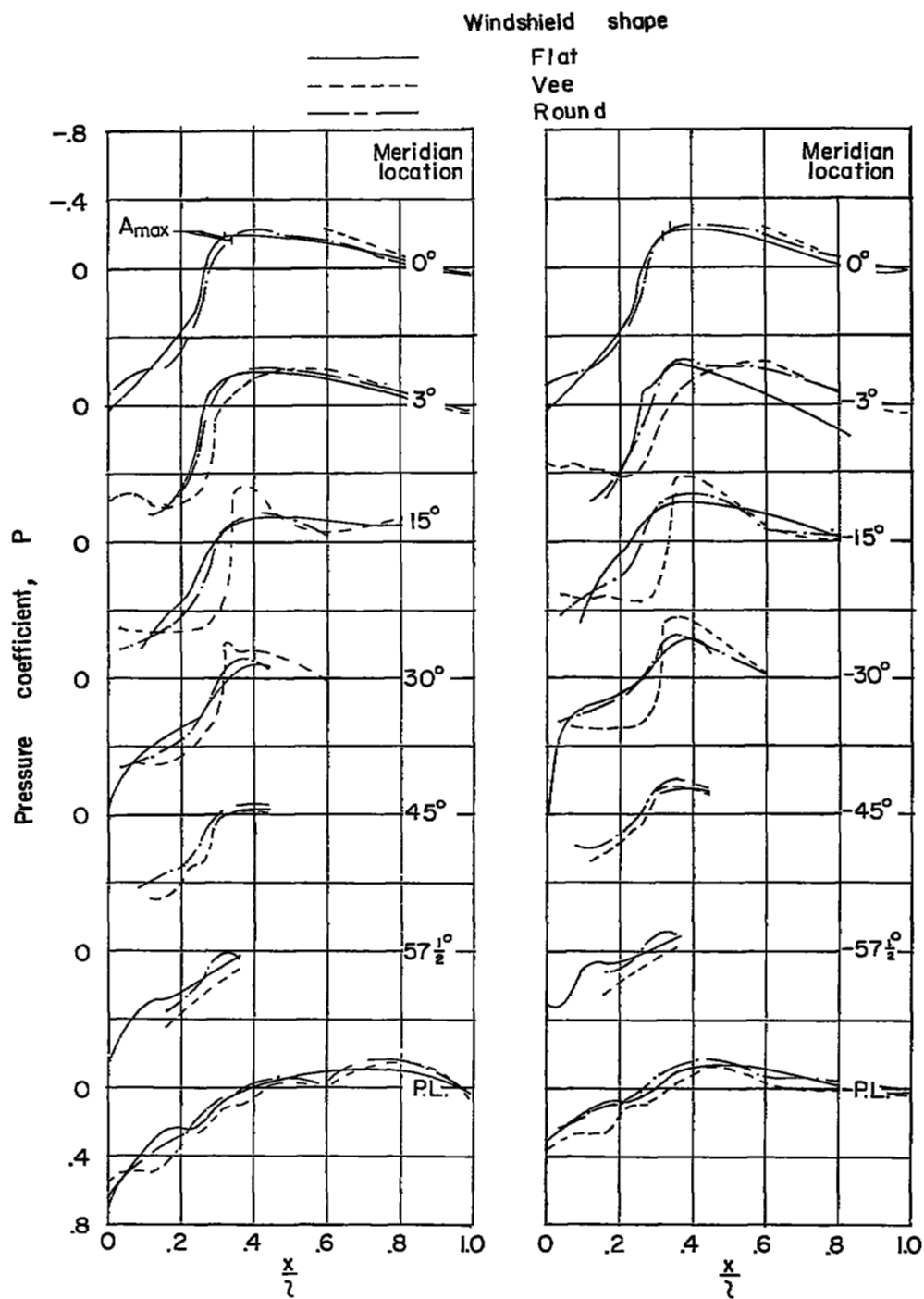


Figure 8.- Incremental drag coefficient C_{DA} (based on canopy maximum cross-section area) for flat-windshield canopies compared with drag coefficients C_{DA} for bodies of revolution having various locations of maximum diameter (ref. 7).



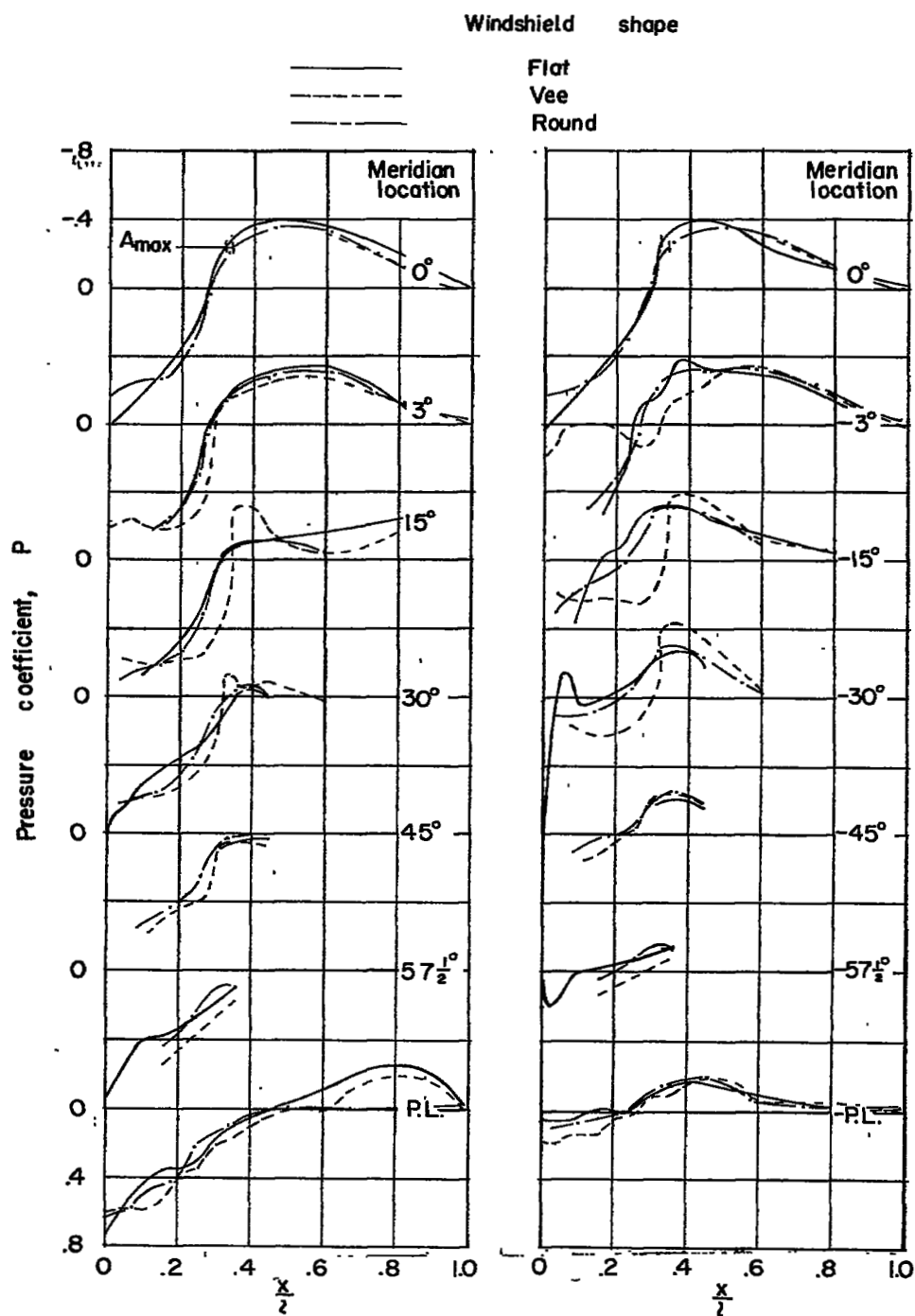
(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

Figure 9.- Effect of windshield shape on pressure-coefficient distributions on large forward-located canopies at $M = 1.41$.



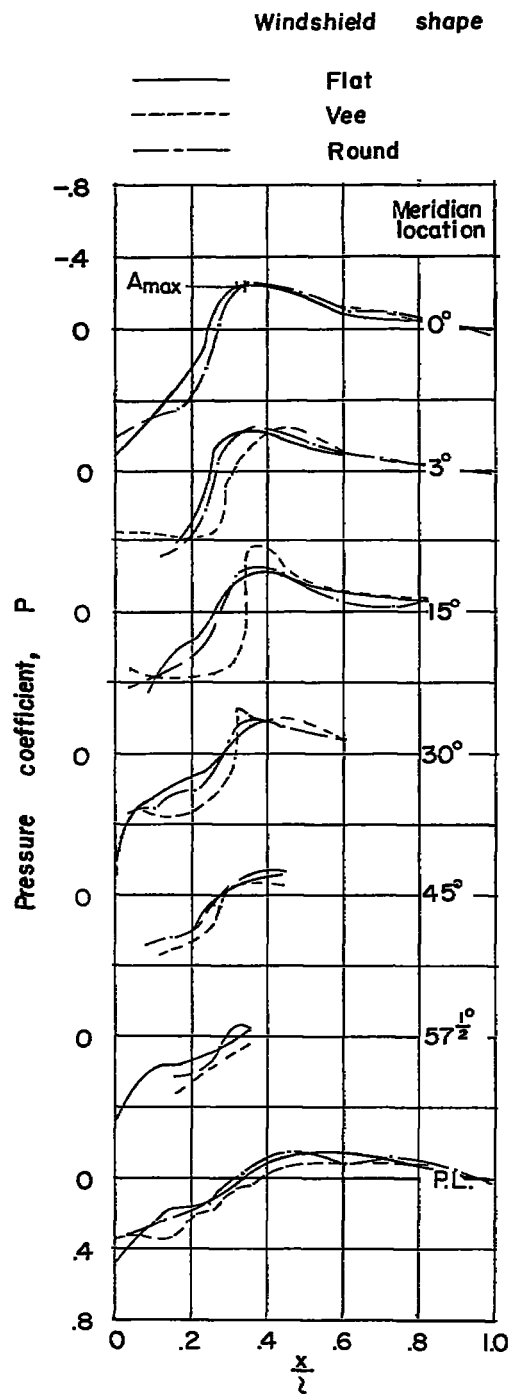
(b) $\alpha = 0.4^\circ$; $\beta = -4^\circ$.

Figure 9.- Continued.



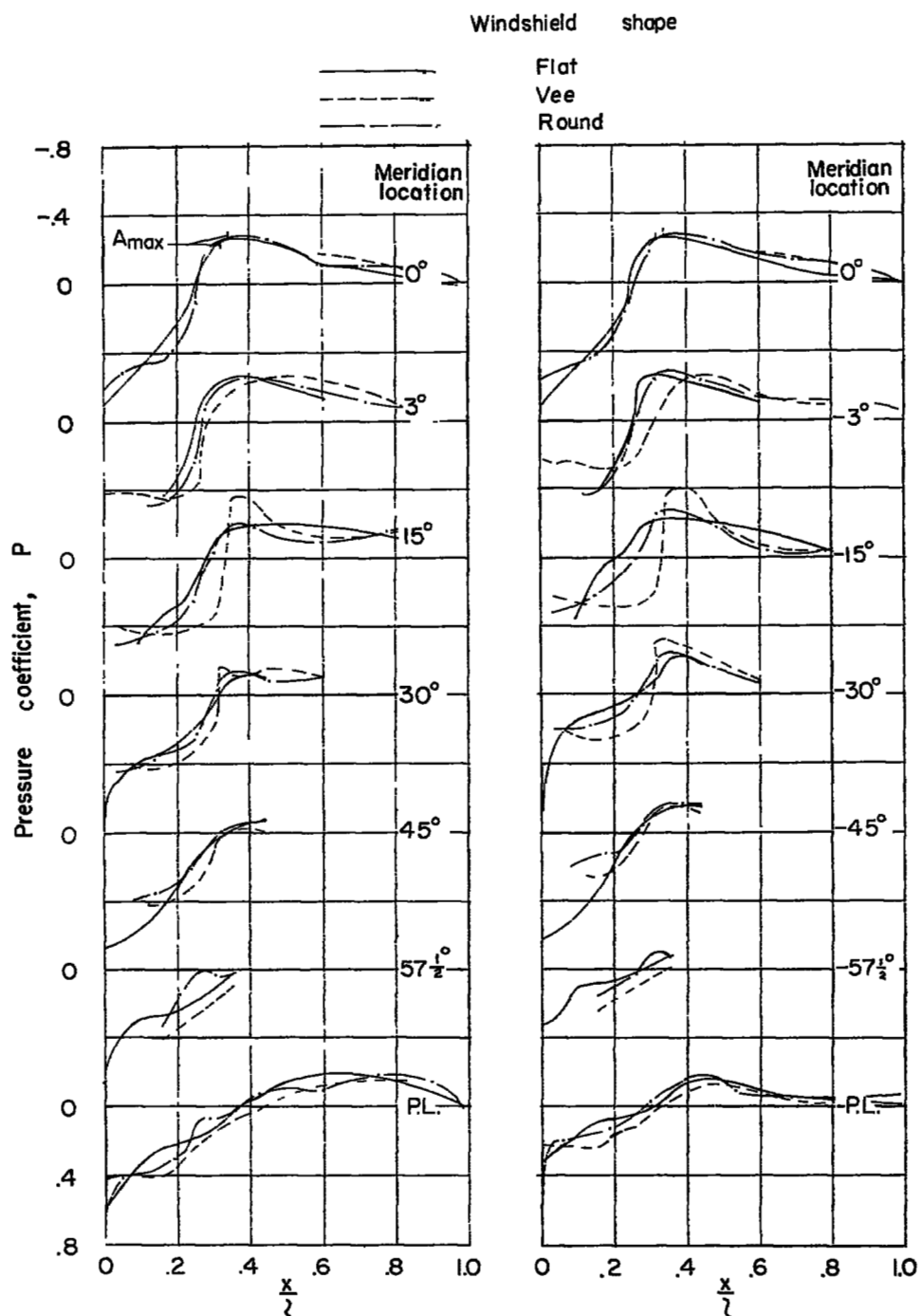
(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$.

Figure 9.- Continued.



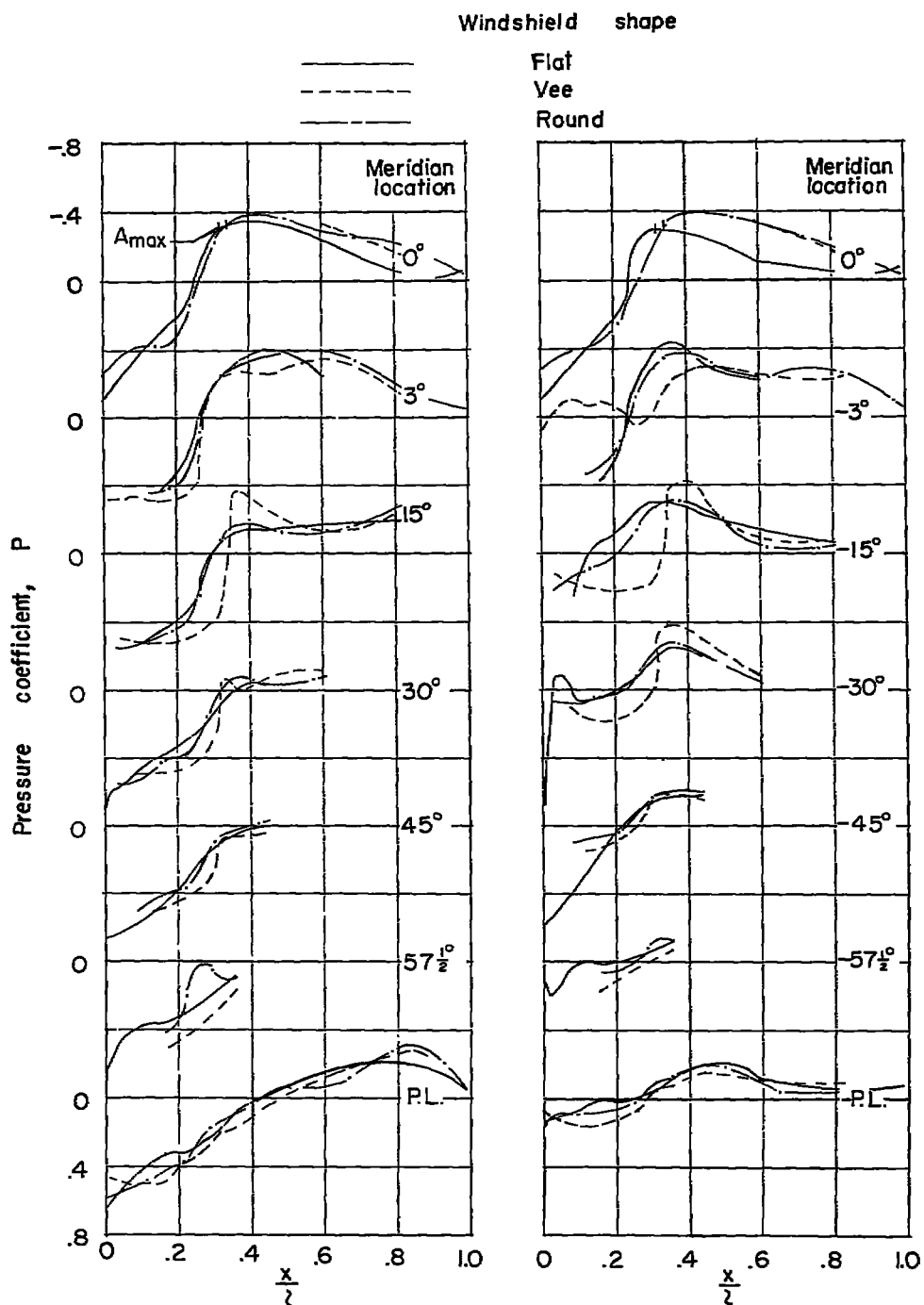
(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

Figure 9.- Continued.



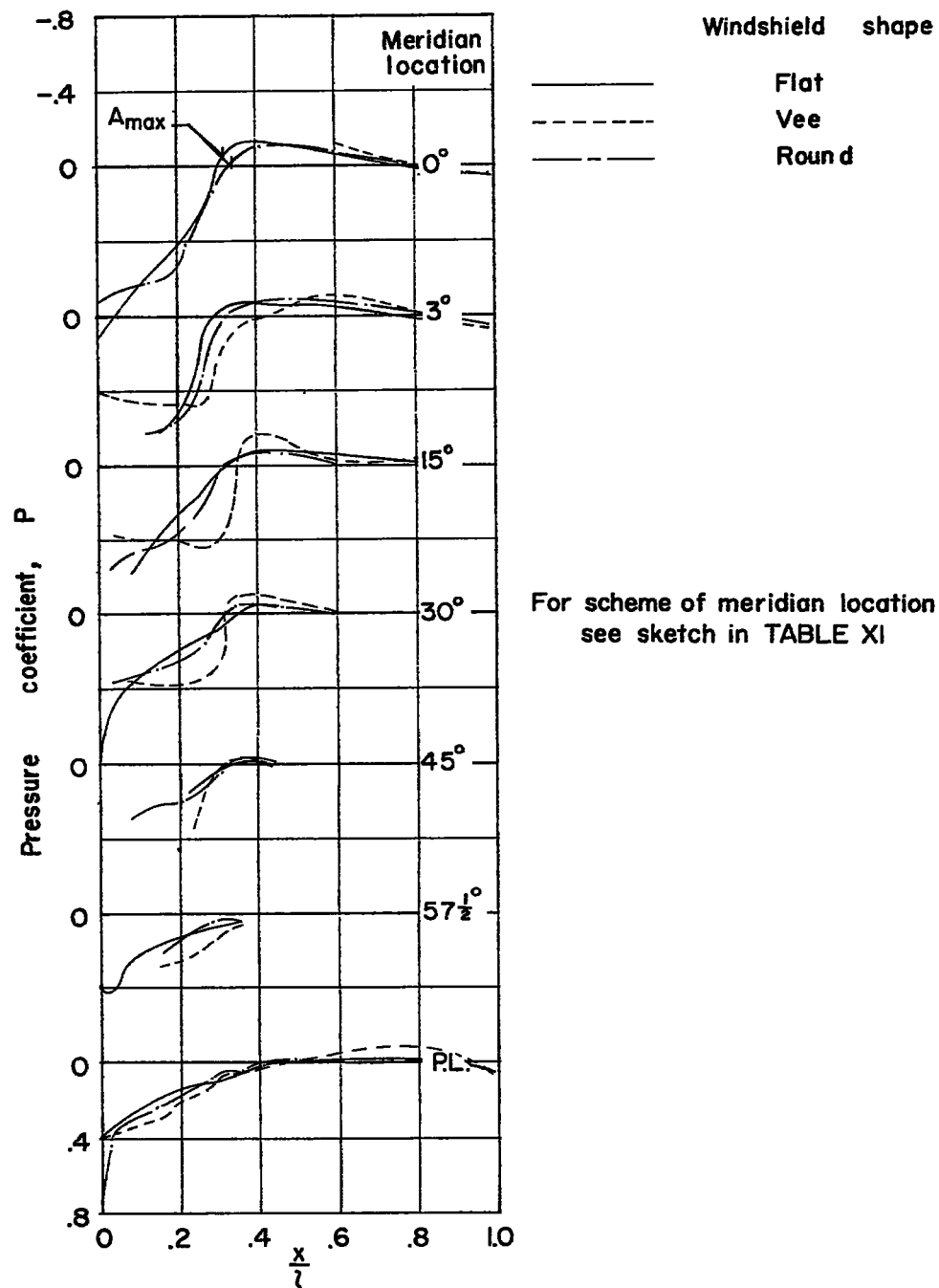
(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.

Figure 9.- Continued.



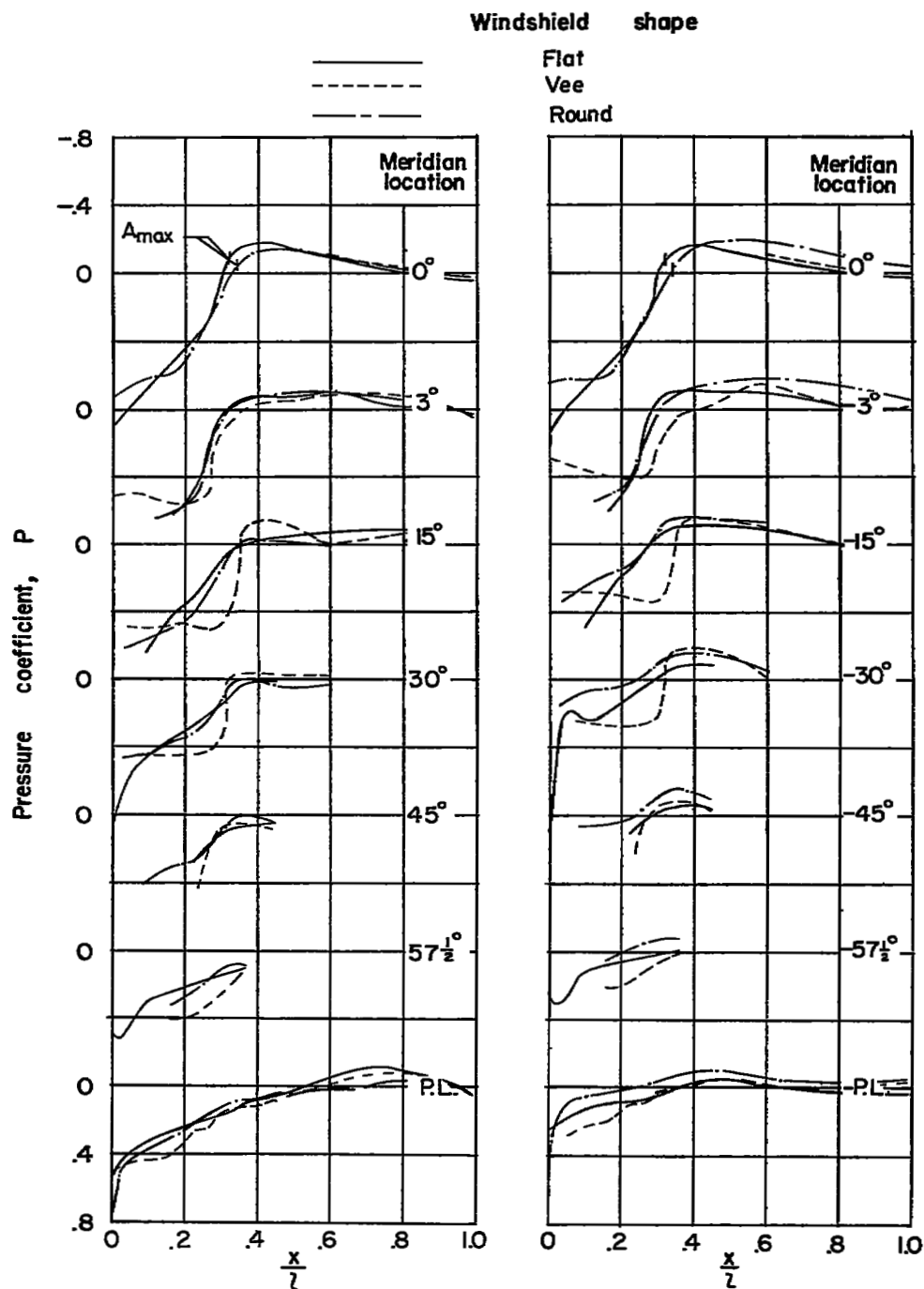
(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 9.- Concluded.



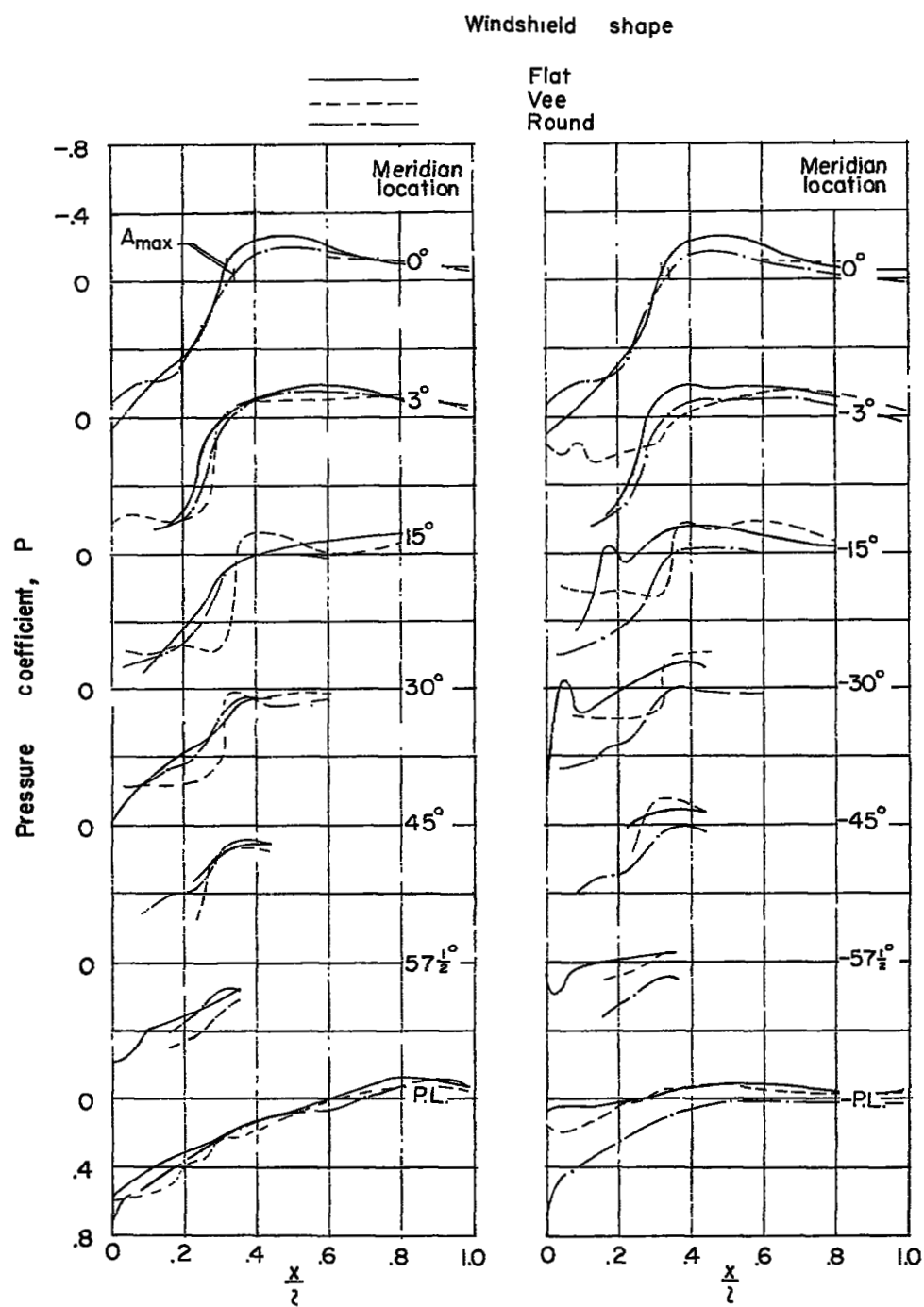
(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

Figure 10.- Effect of windshield shape on pressure-coefficient distributions on large forward-located canopies at $M = 2.01$.



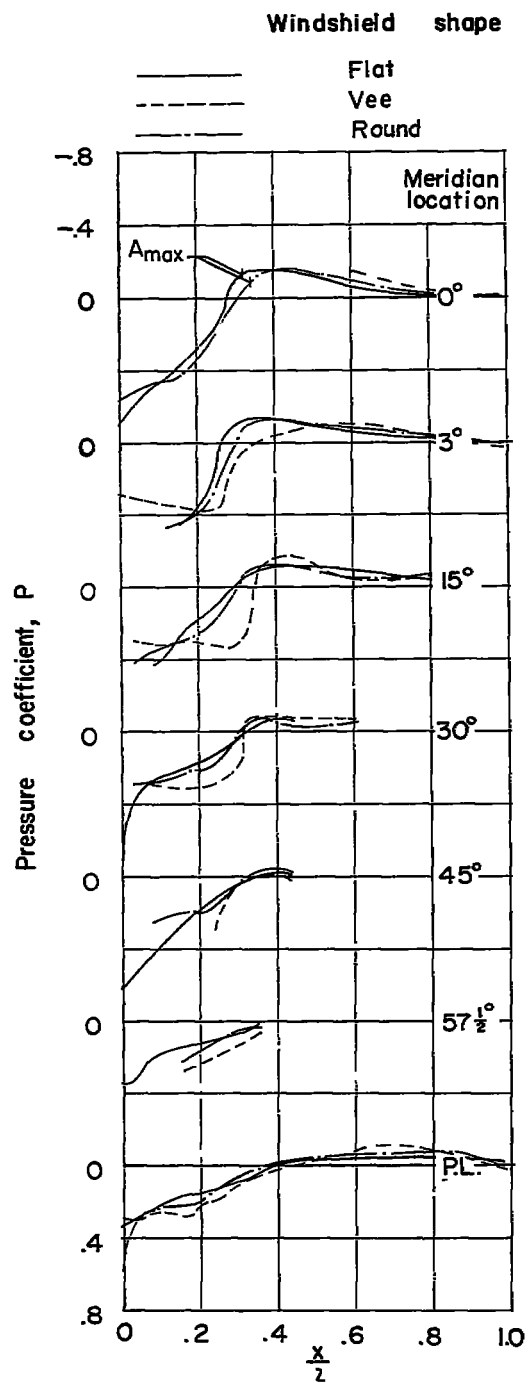
(b) $\alpha = 0.4^\circ$; $\beta = -4^\circ$.

Figure 10.- Continued.



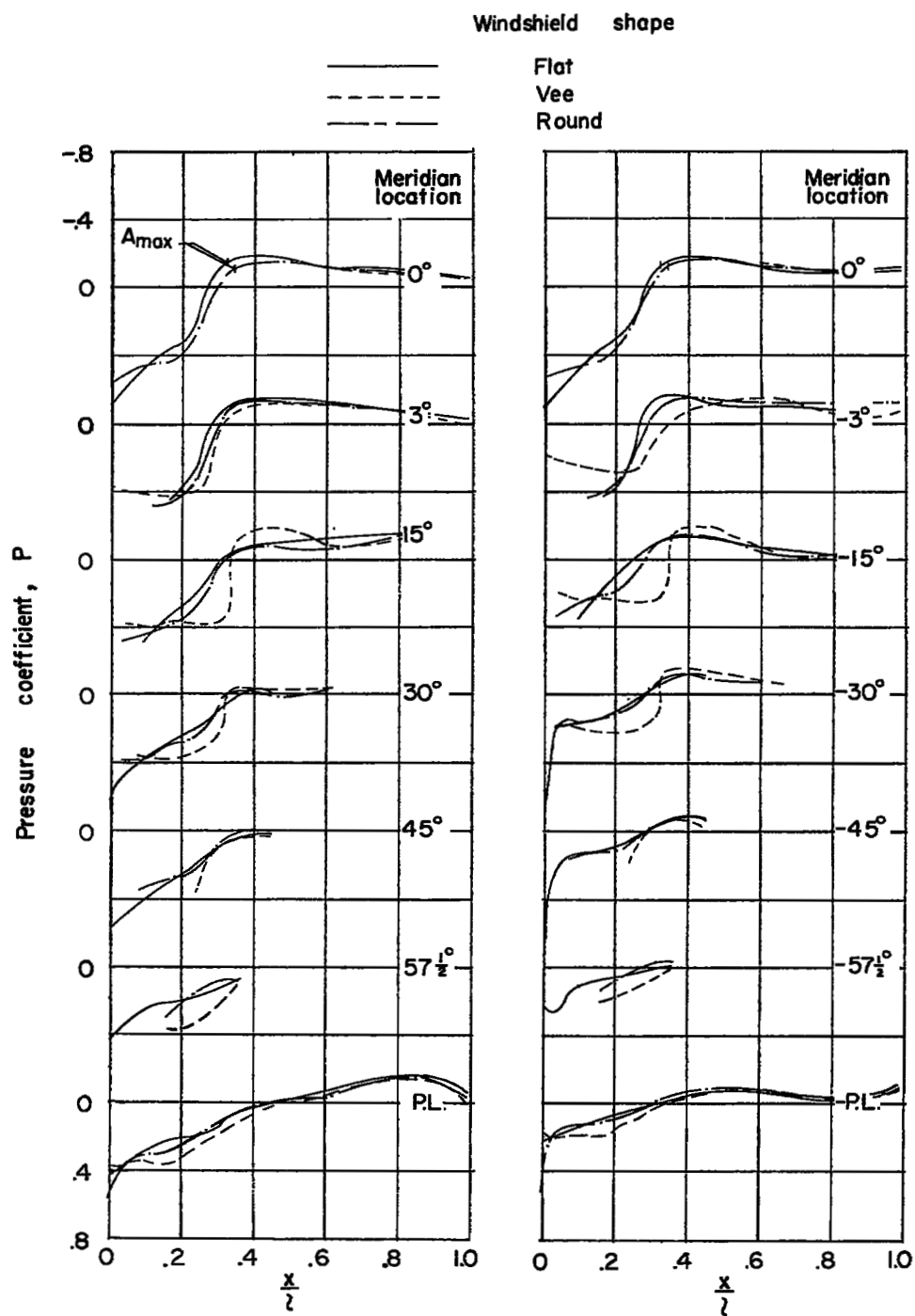
(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$.

Figure 10.- Continued.



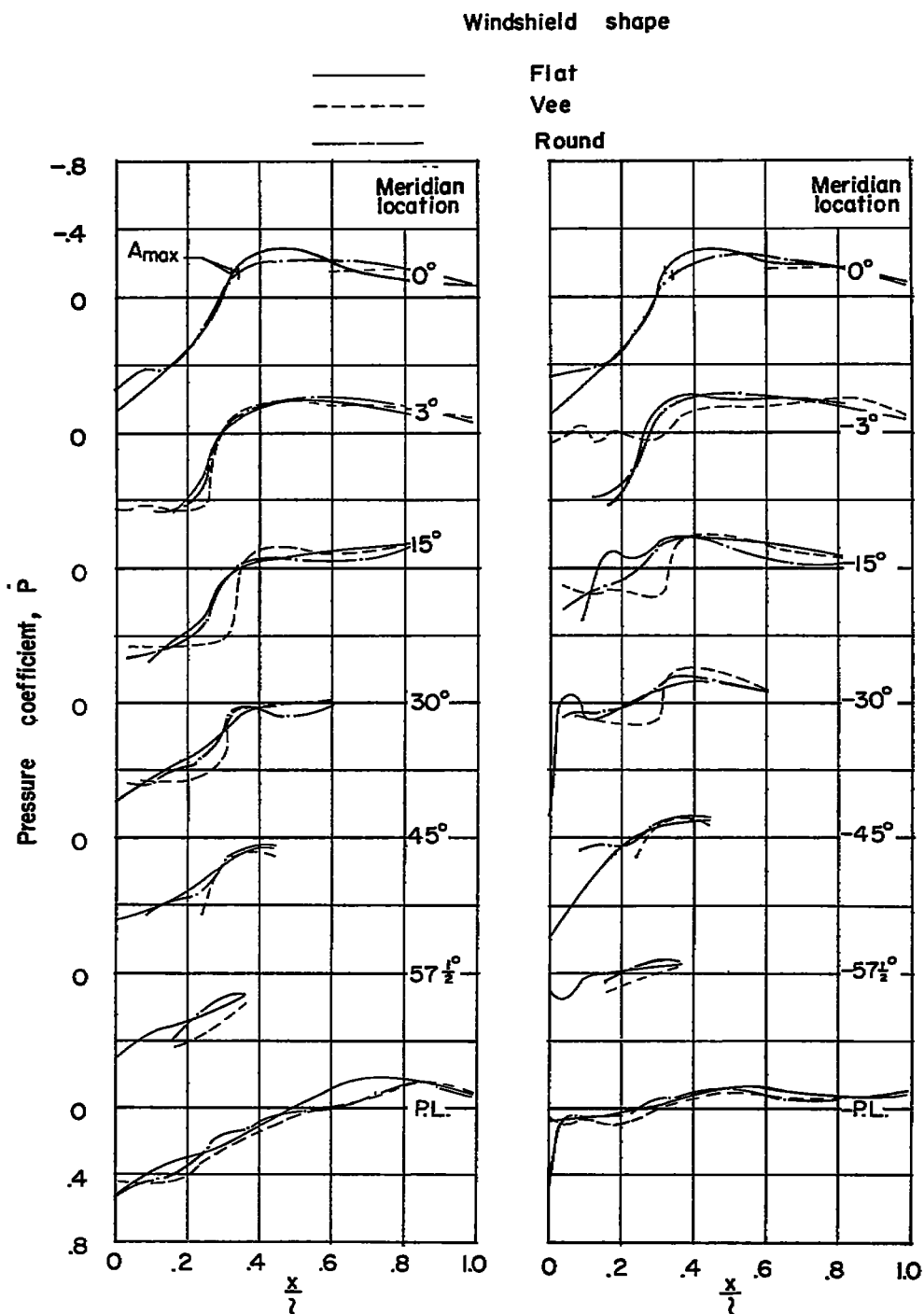
(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

Figure 10.- Continued.



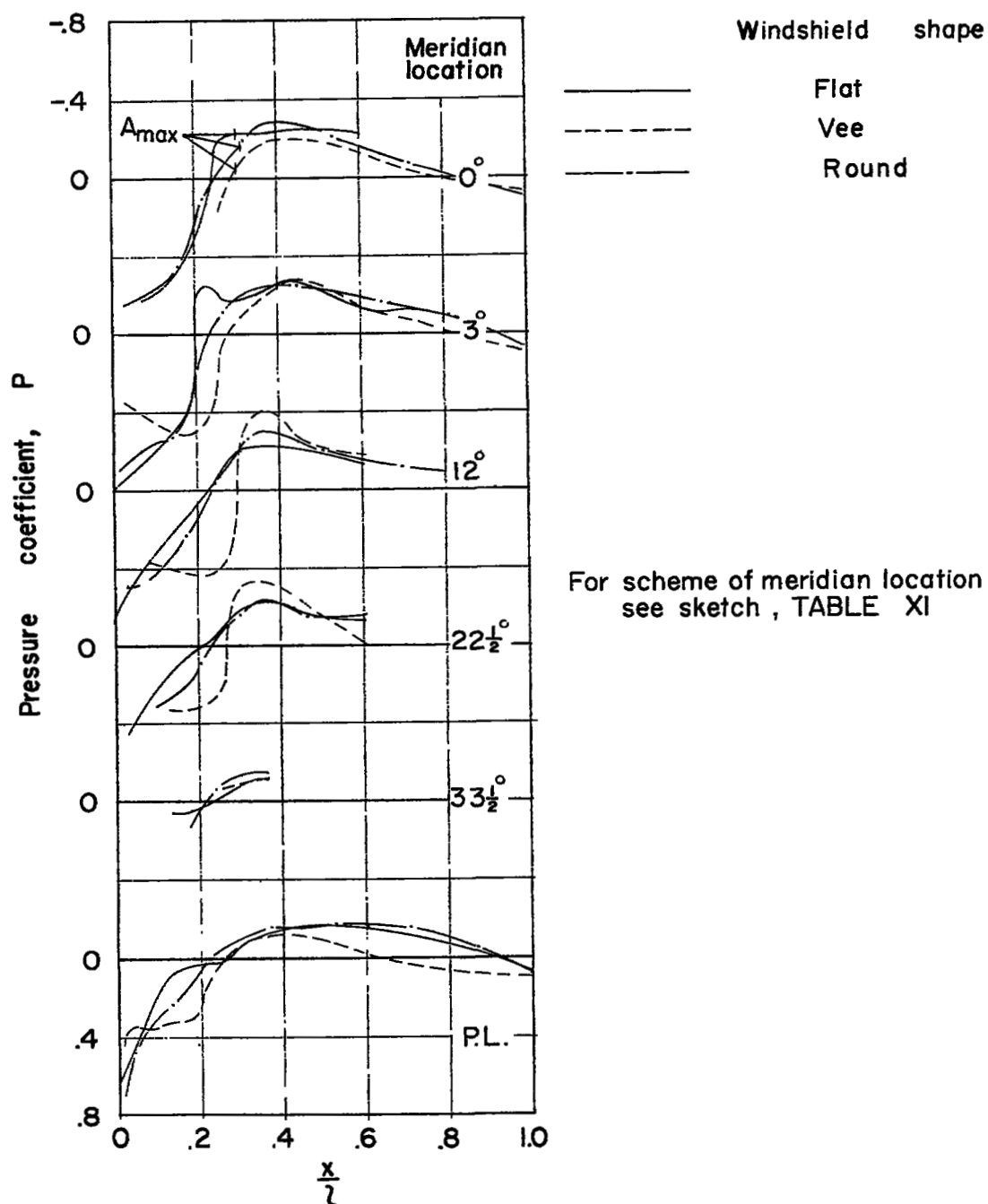
(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.

Figure 10.- Continued.



(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 10.- Concluded.



(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

Figure 11.- Effect of windshield shape on pressure-coefficient distributions on large rearward-located canopies at $M = 1.41$.

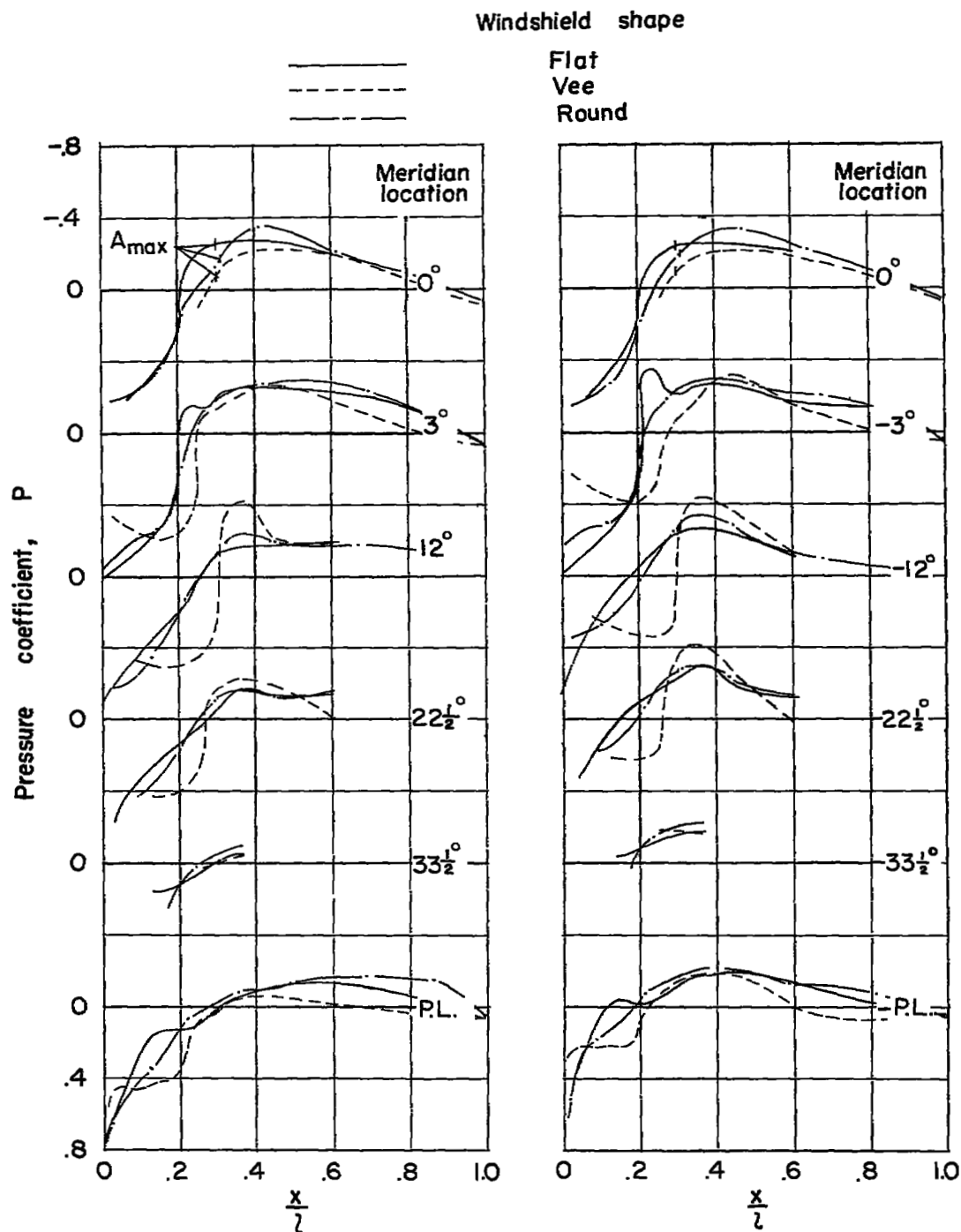
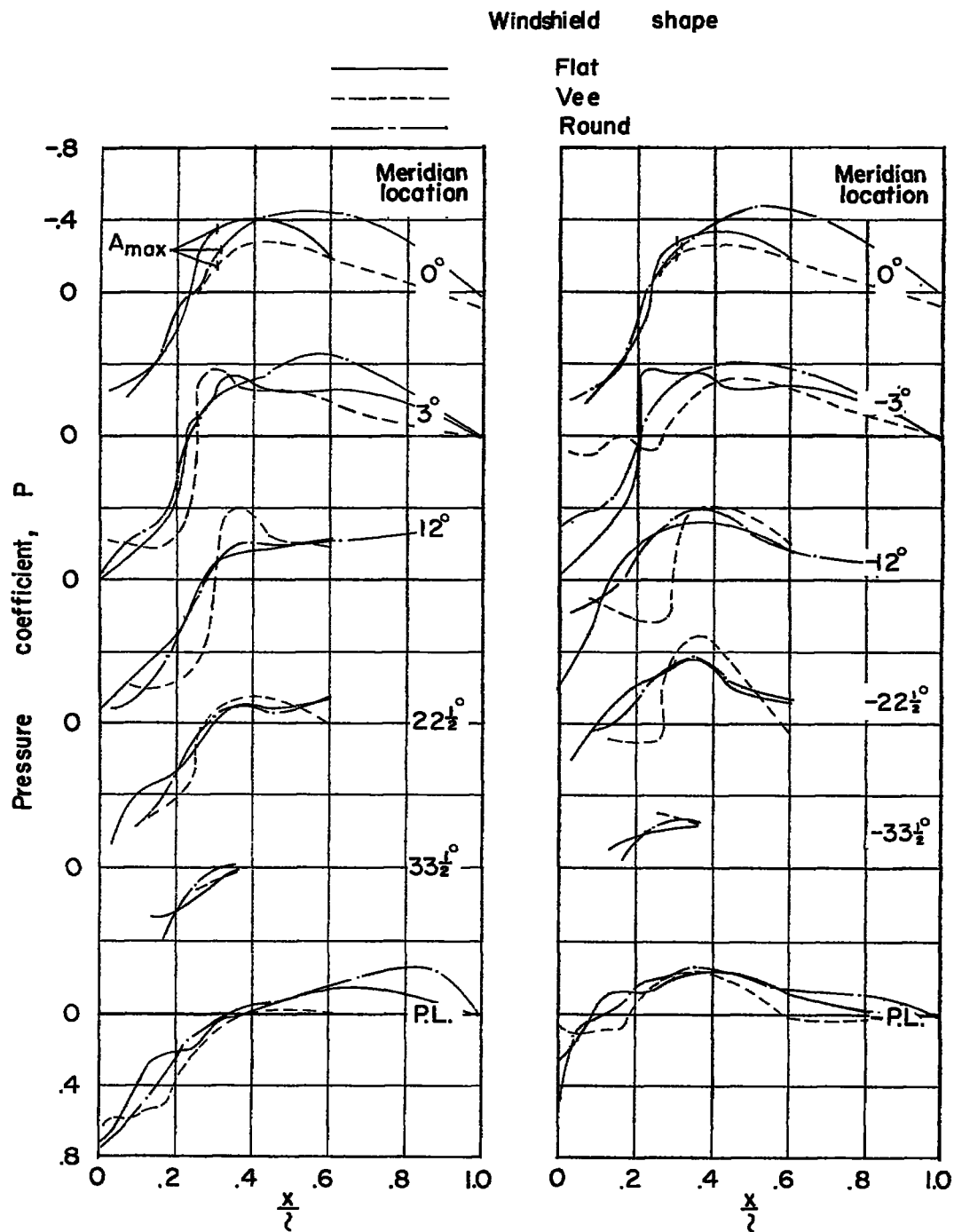
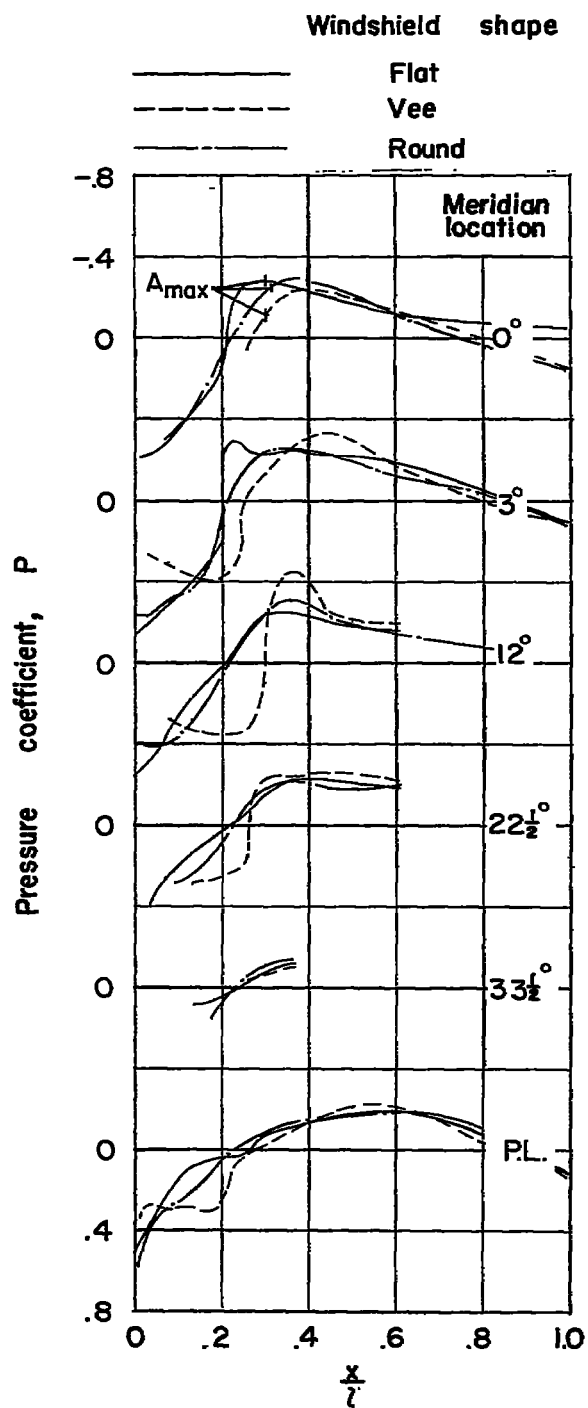


Figure 11.- Continued.



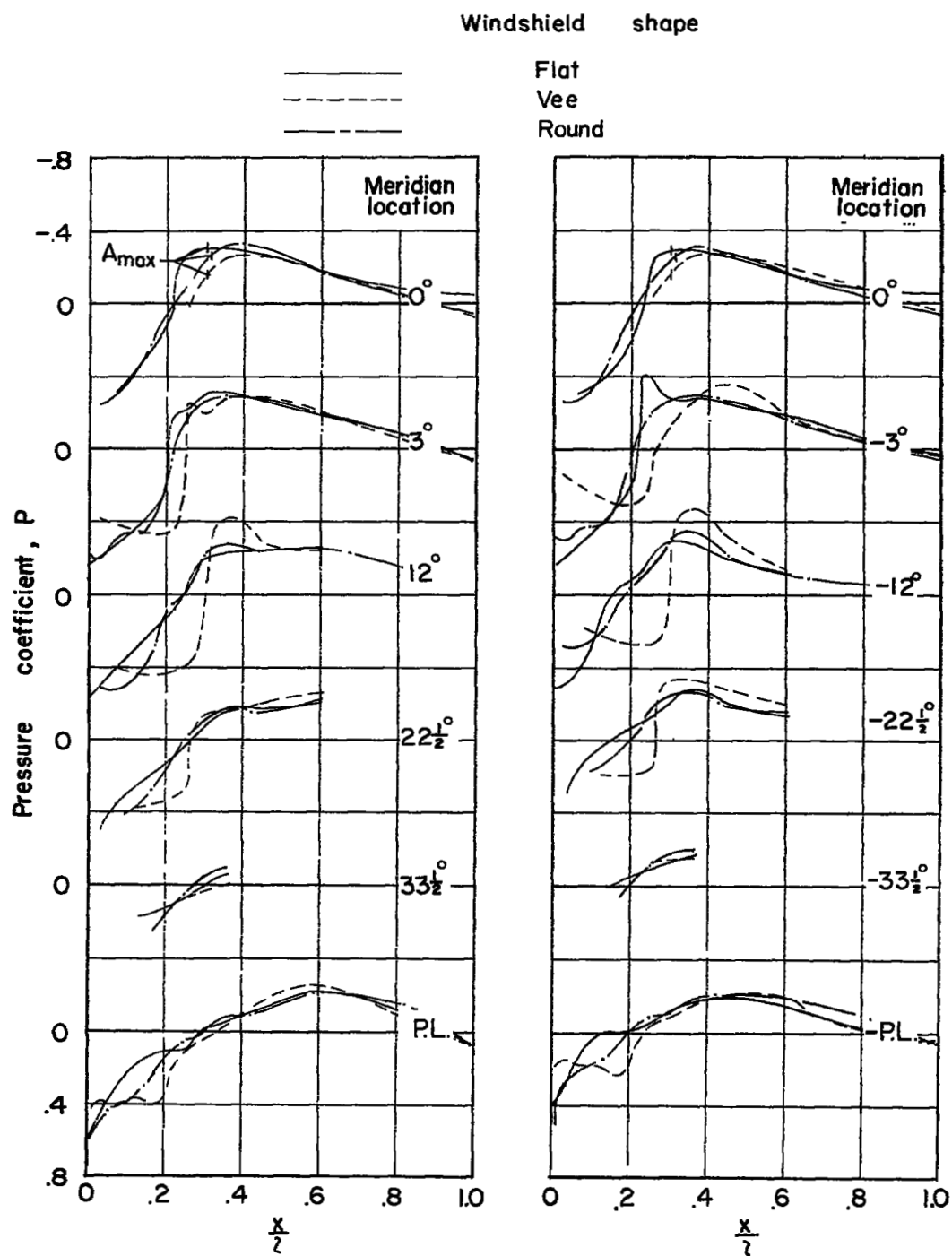
(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$.

Figure 11.- Continued.



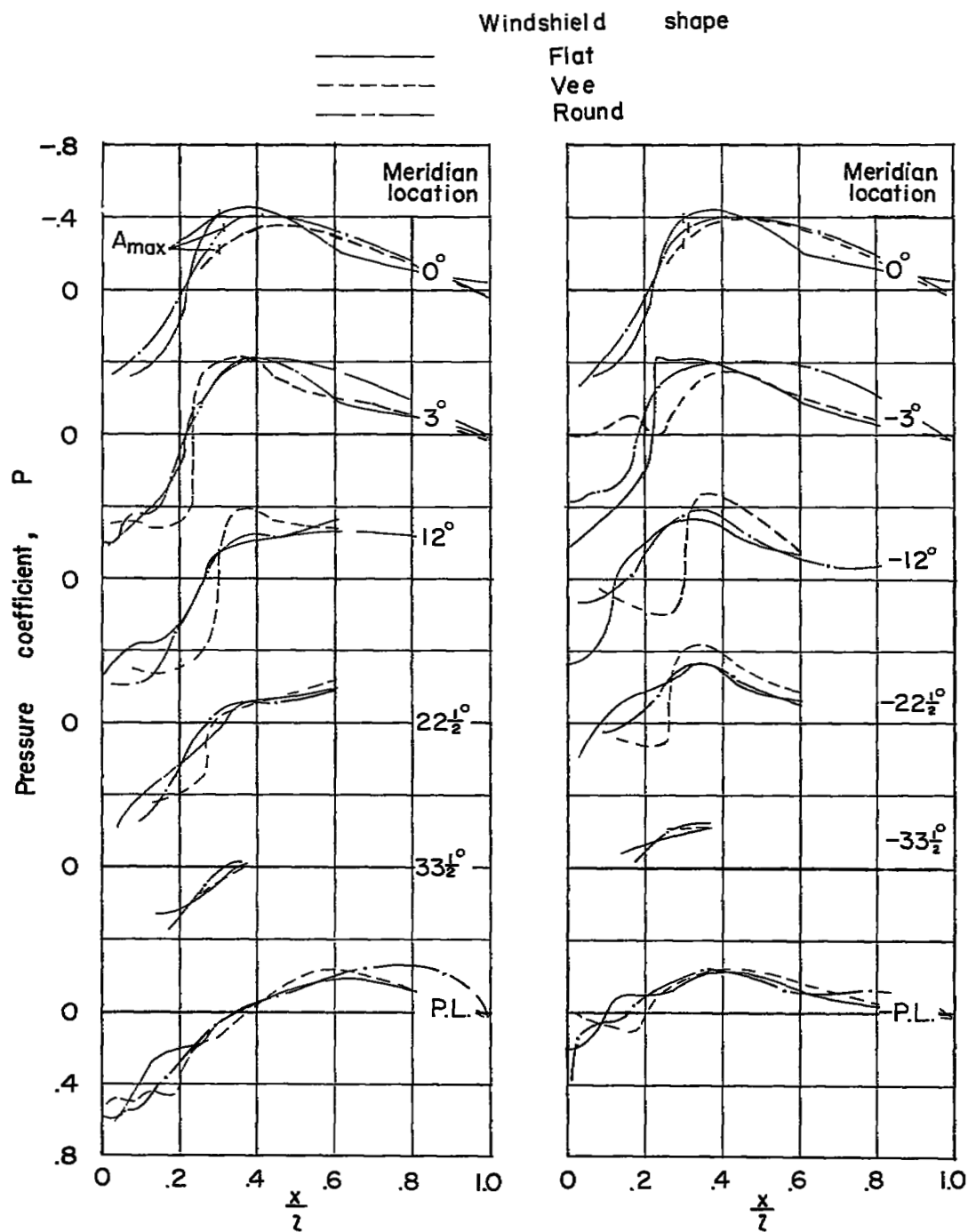
(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

Figure 11.- Continued.



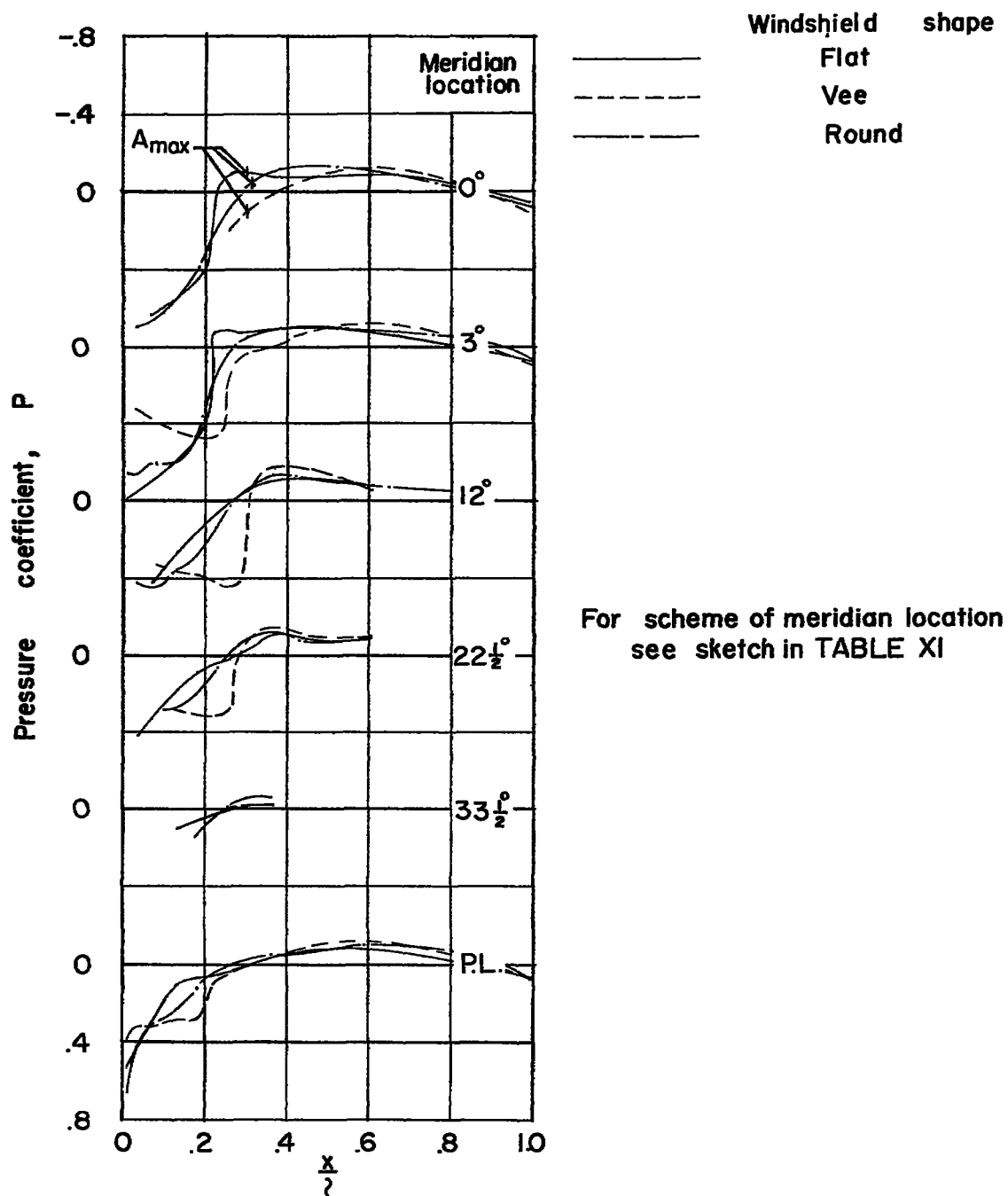
(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.

Figure 11.- Continued.



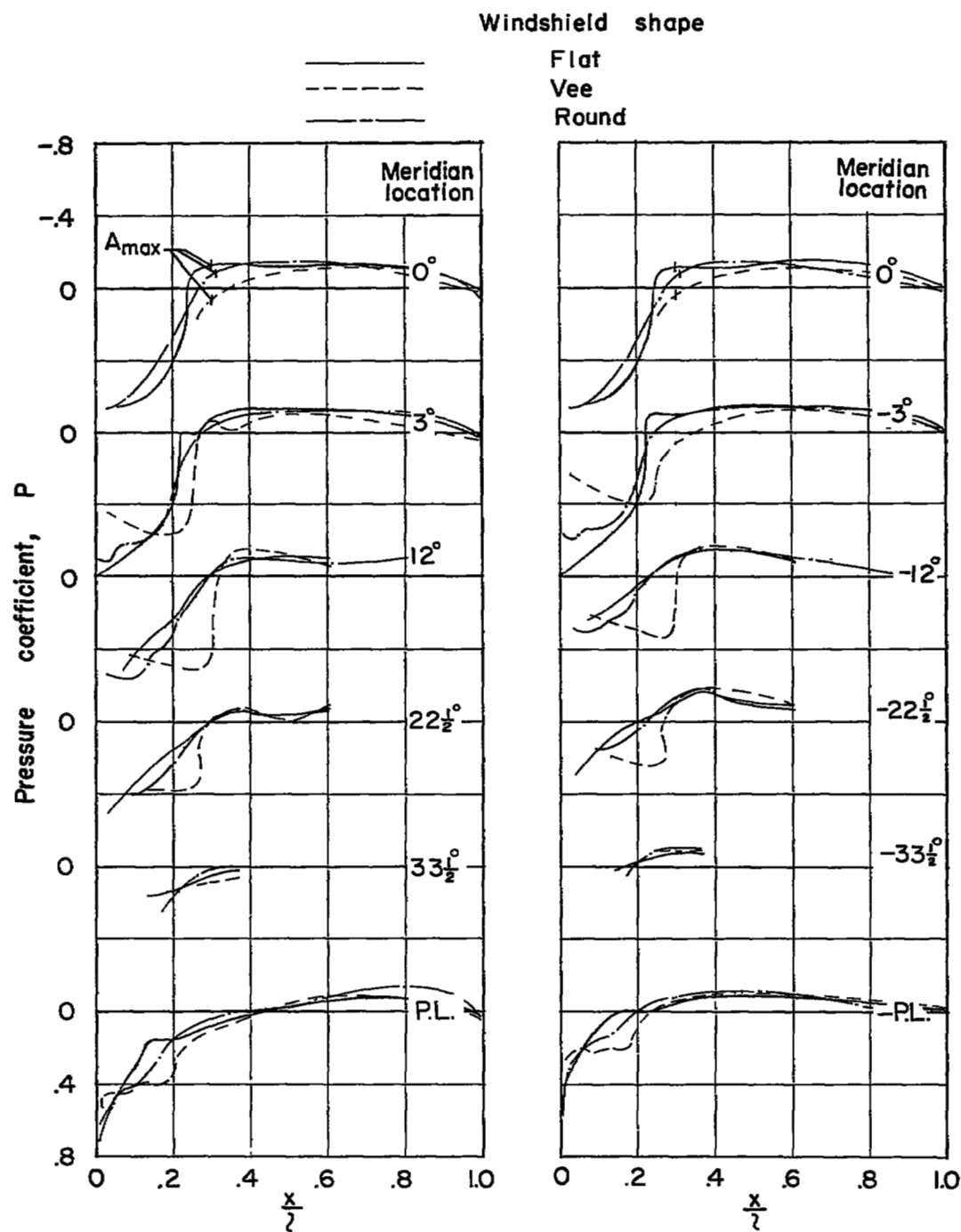
(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 11.- Concluded.



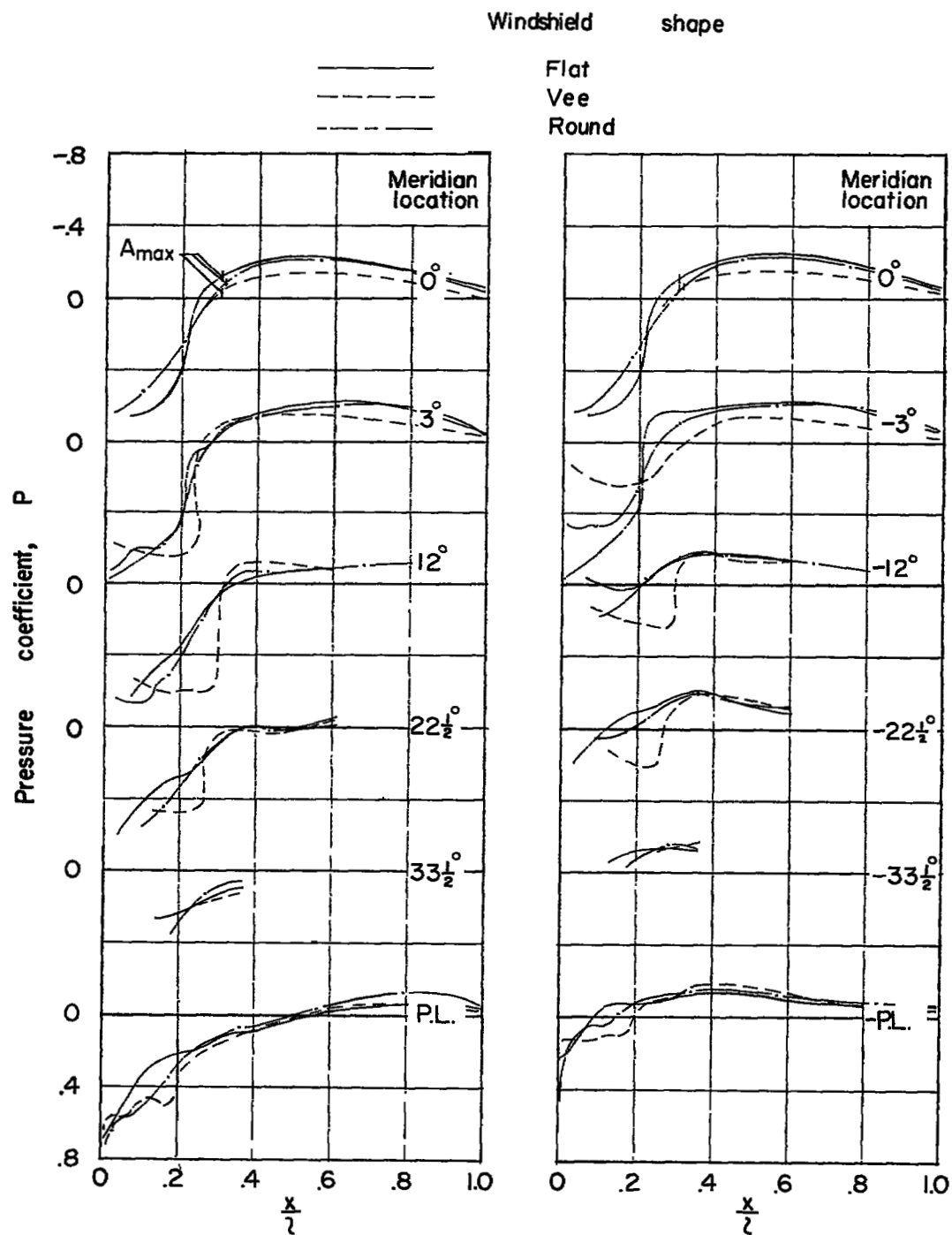
(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

Figure 12.- Effect of windshield shape on pressure-coefficient distributions on large rearward-located canopies at $M = 2.01$.



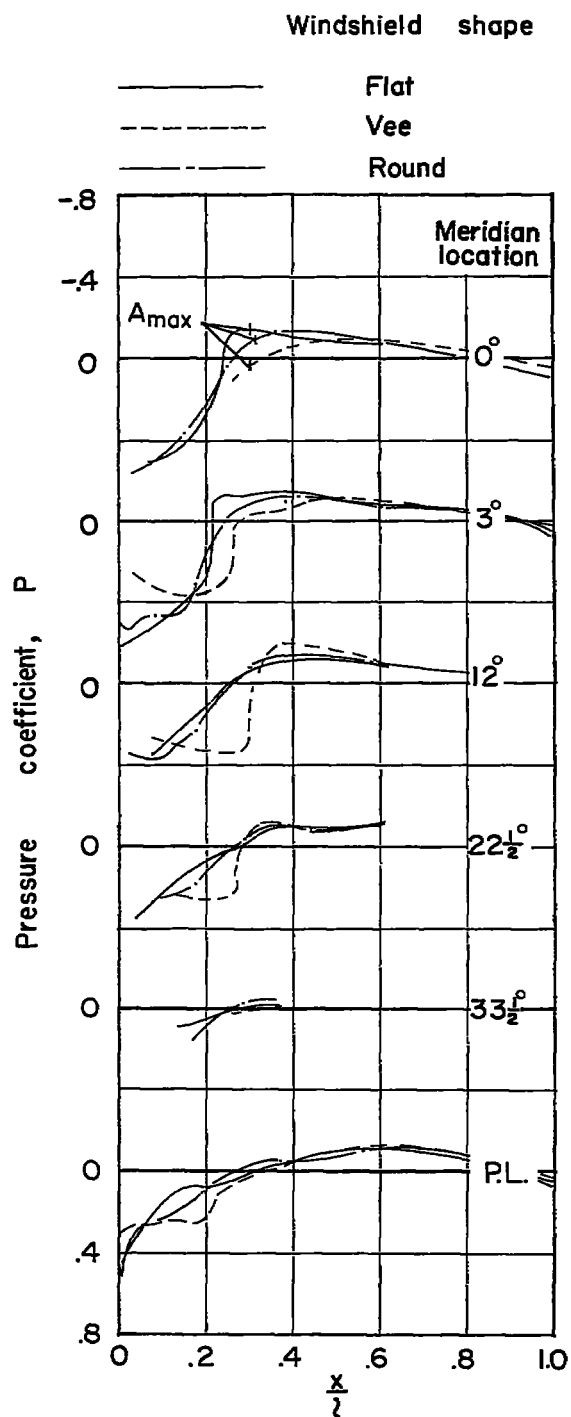
(b) $\alpha = 0.4^\circ$; $\beta = -4^\circ$.

Figure 12.- Continued.



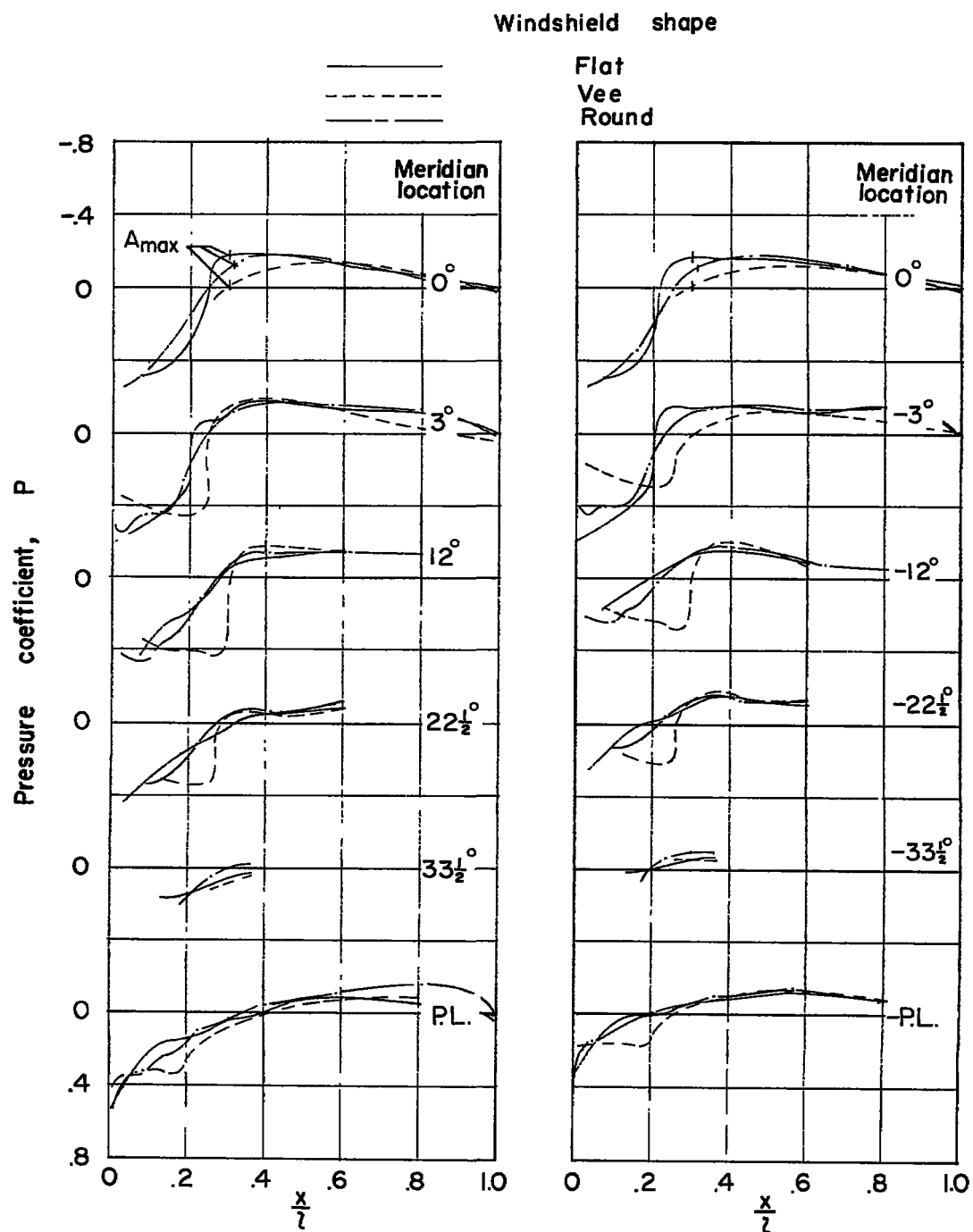
(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$.

Figure 12.- Continued.



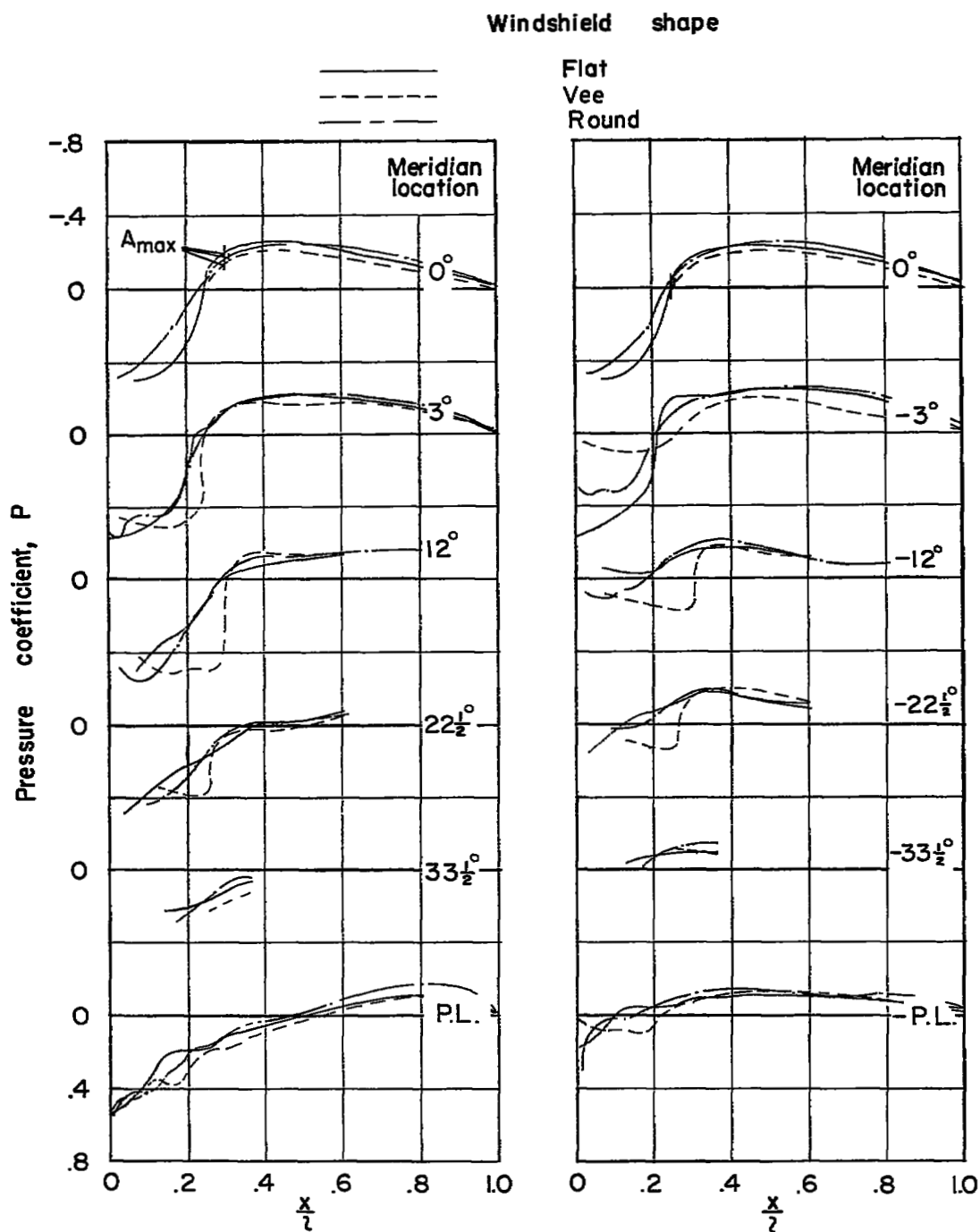
(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

Figure 12.- Continued.



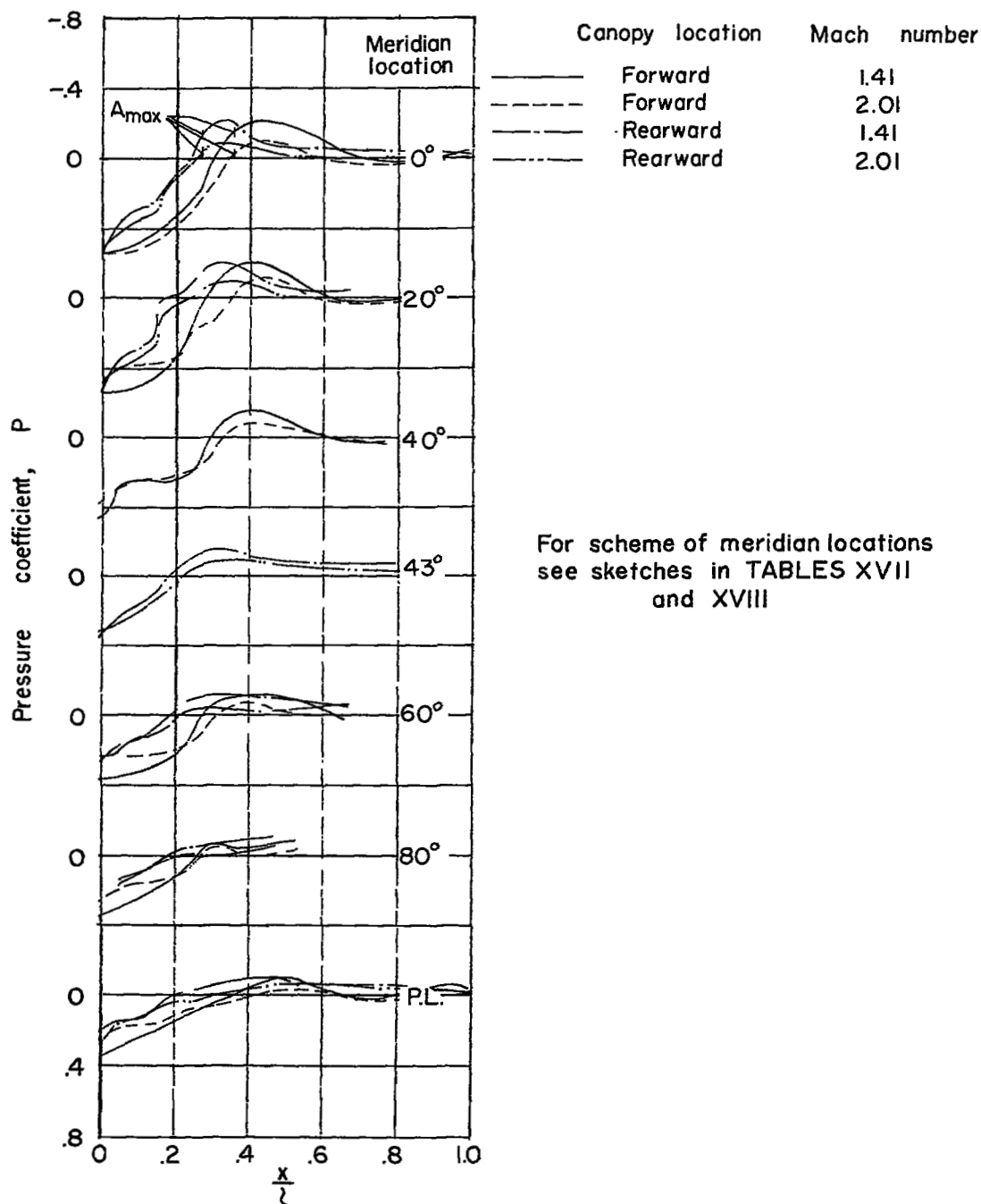
(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.

Figure 12.- Continued.



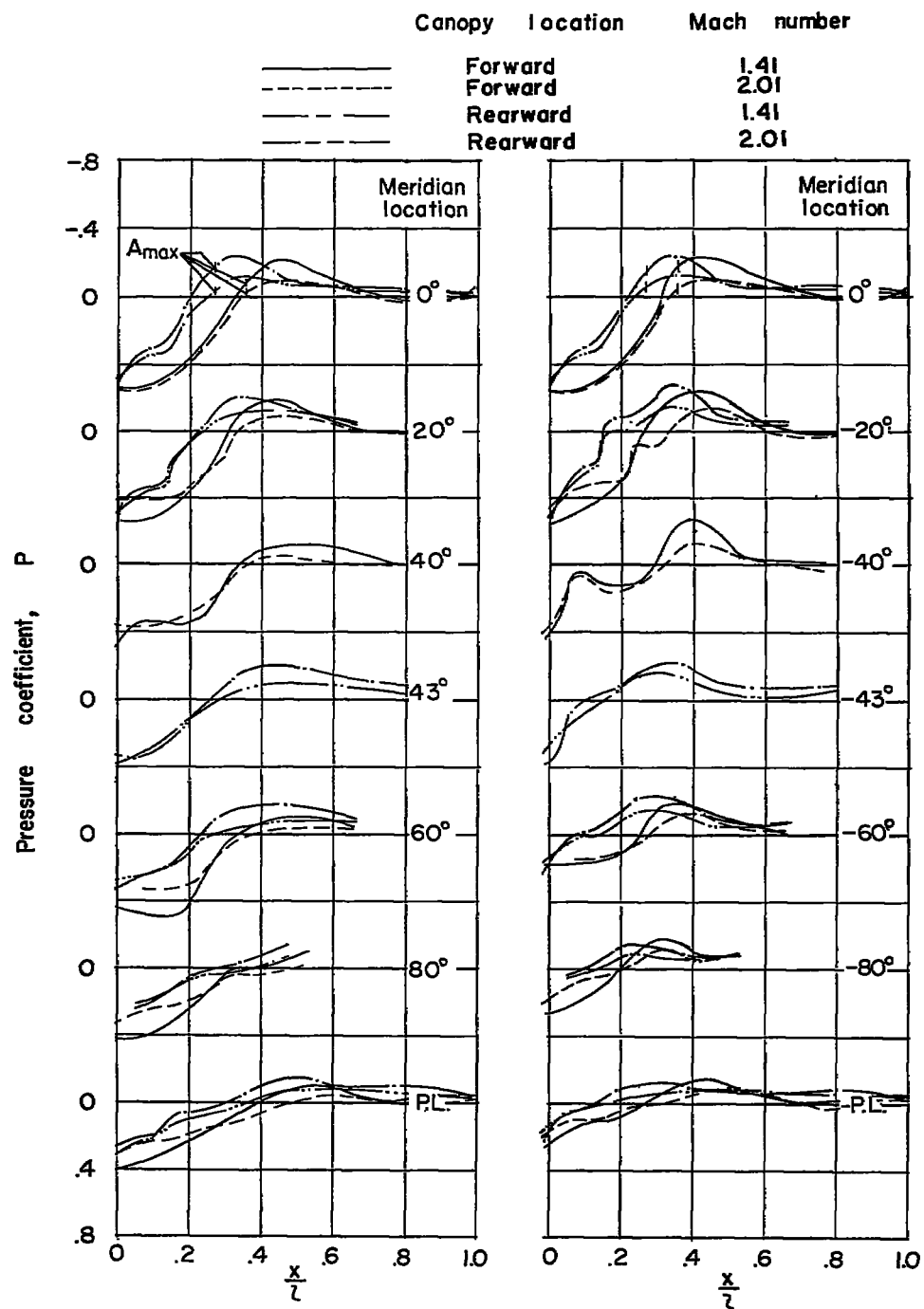
(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 12.- Concluded.



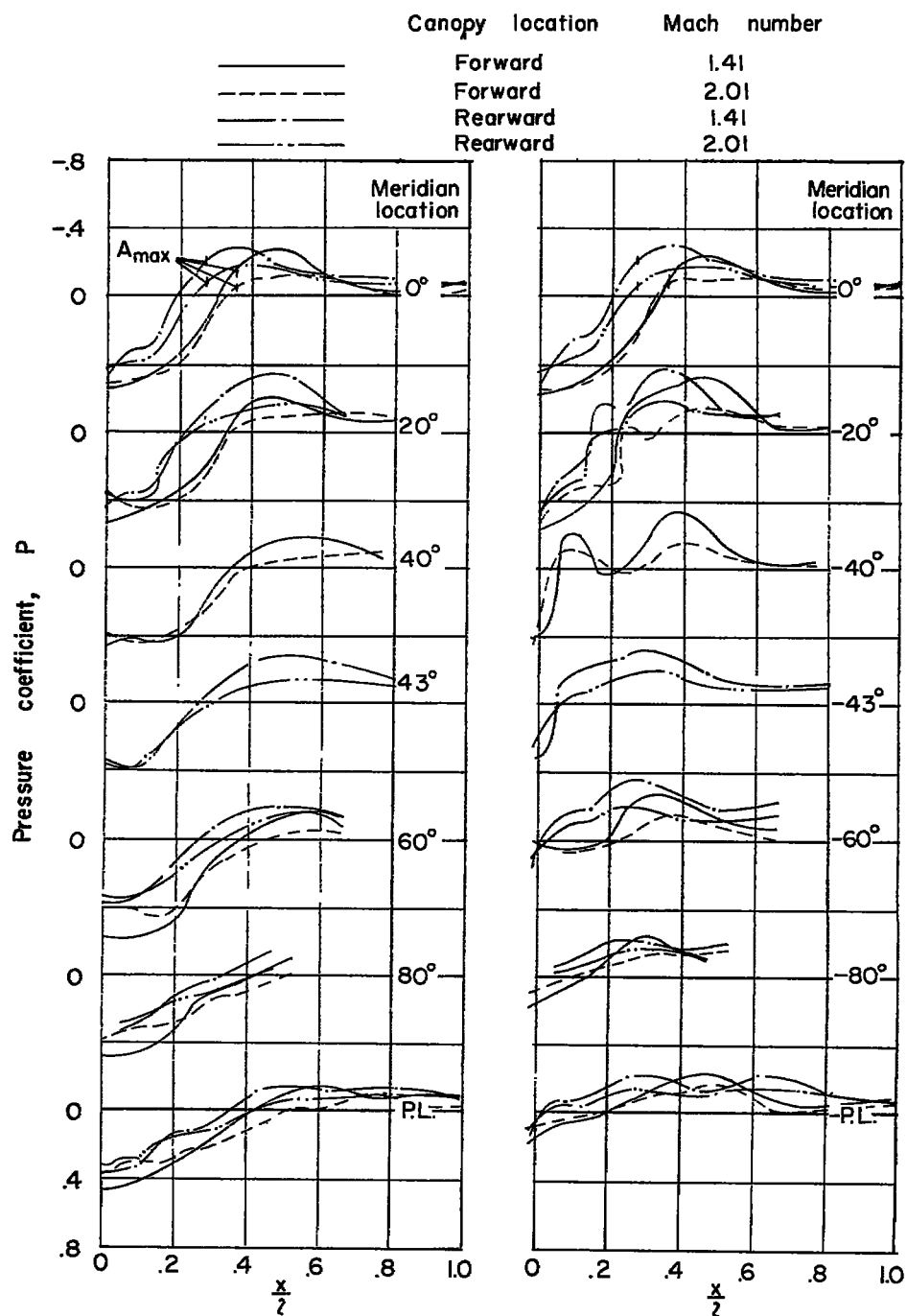
(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$.

Figure 13.- Pressure distributions on small canopies at $M = 1.41$ and 2.01 for various angles of attack and sideslip.



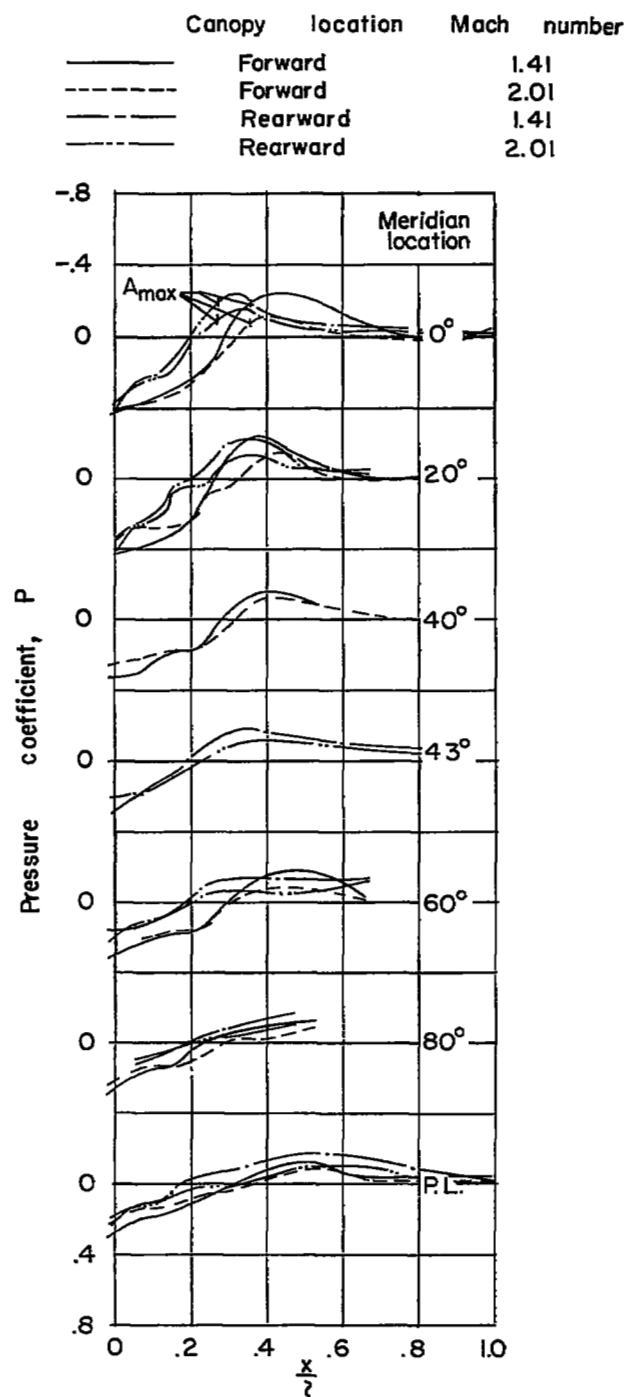
(b) $\alpha = 0.4^\circ$; $\beta = -4^\circ$.

Figure 13.- Continued.



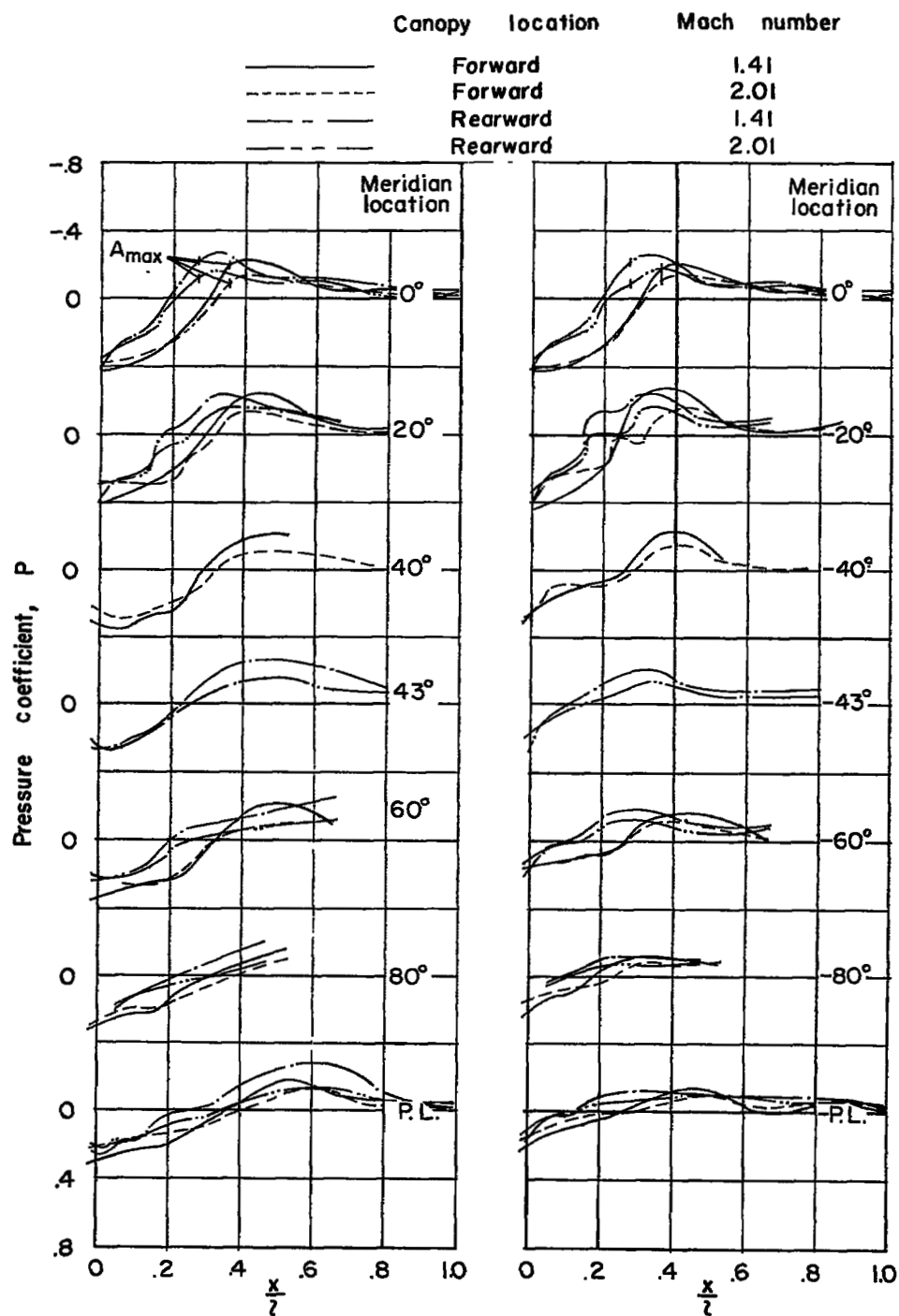
(c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$.

Figure 13.- Continued.



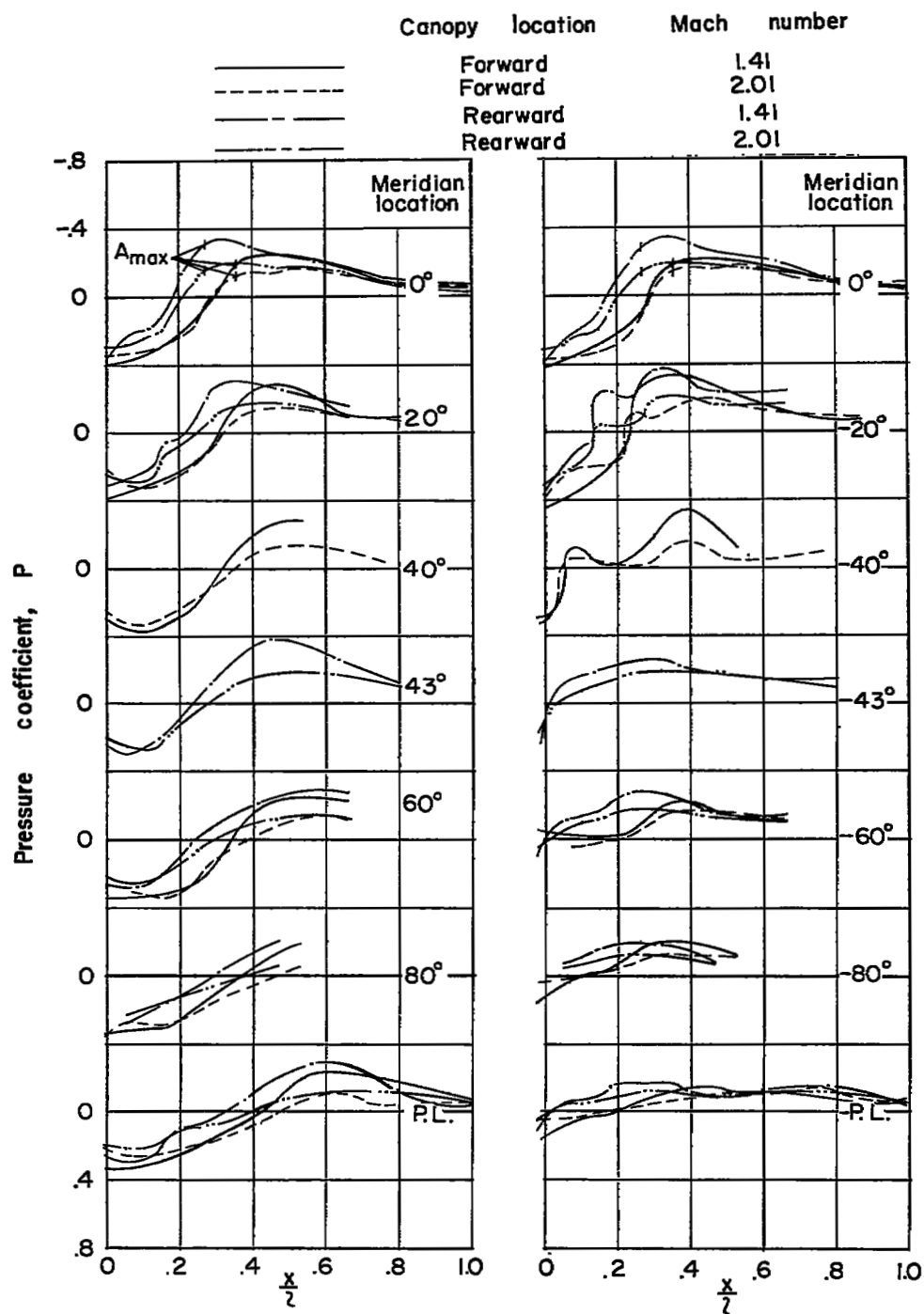
(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.

Figure 13.- Continued.



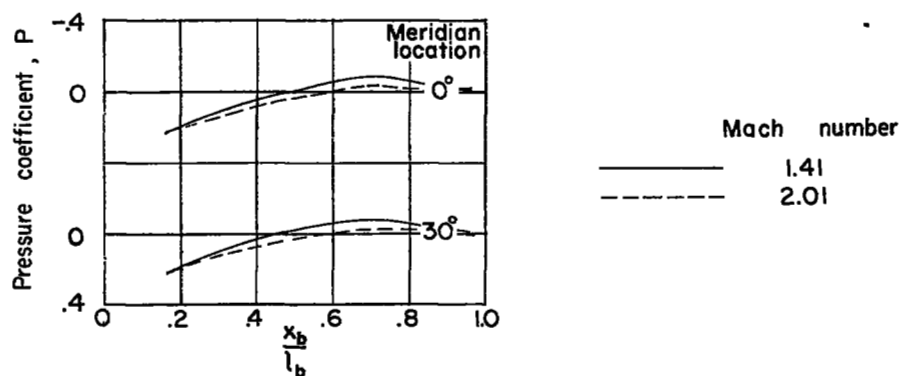
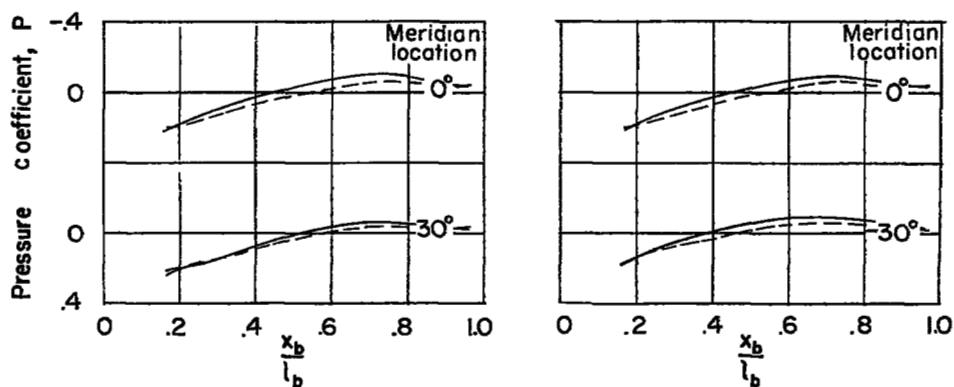
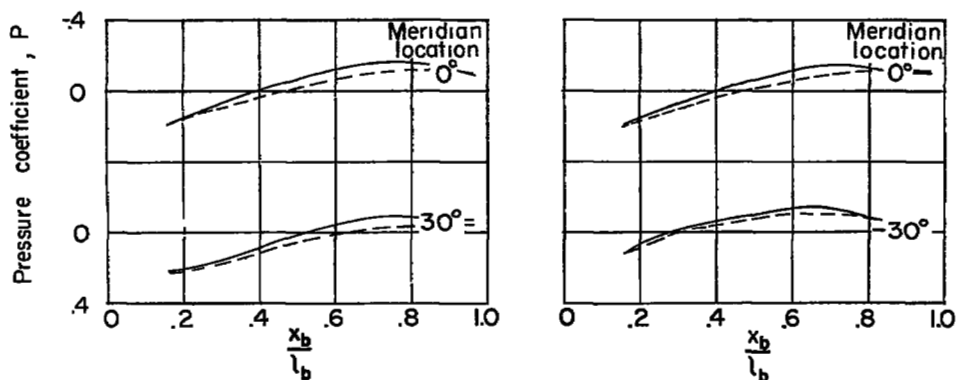
(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.

Figure 13.- Continued.



(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 13.- Concluded.

(a) $\alpha = 0.4^\circ$; $\beta = 0^\circ$ (b) $\alpha = 0.4^\circ$; $\beta = -4^\circ$ (c) $\alpha = 0.4^\circ$; $\beta = -8^\circ$ Figure 14.- Pressure distributions on body alone at $M = 1.41$ and 2.01 for various angles of attack and sideslip.

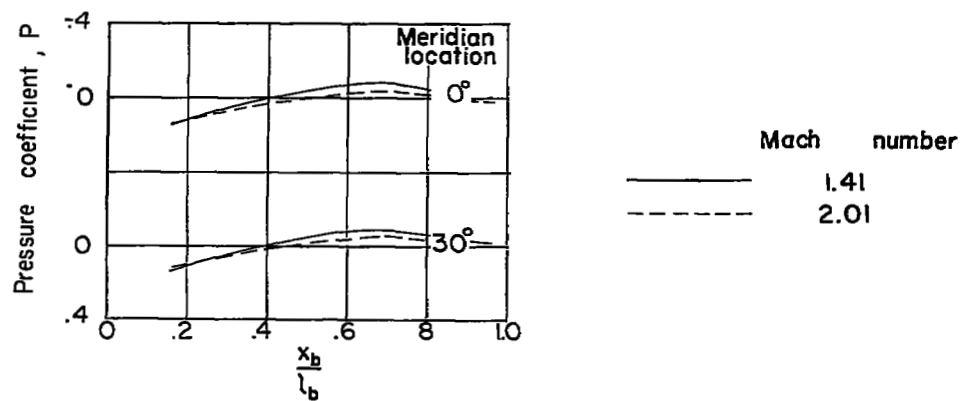
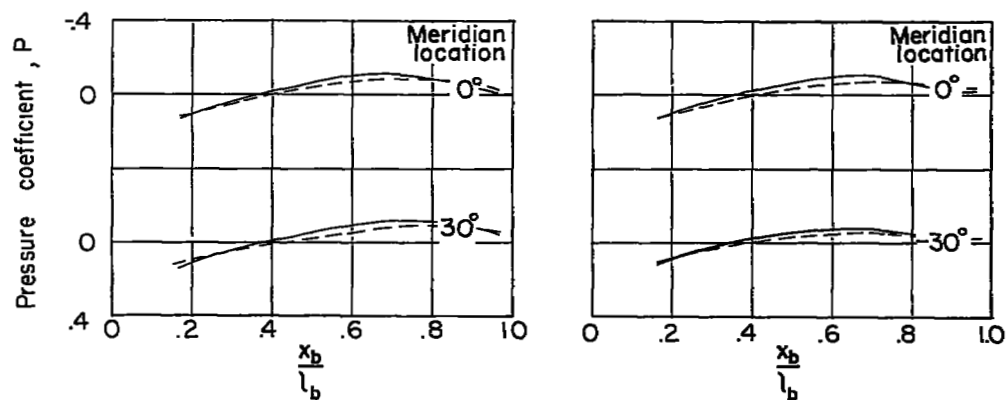
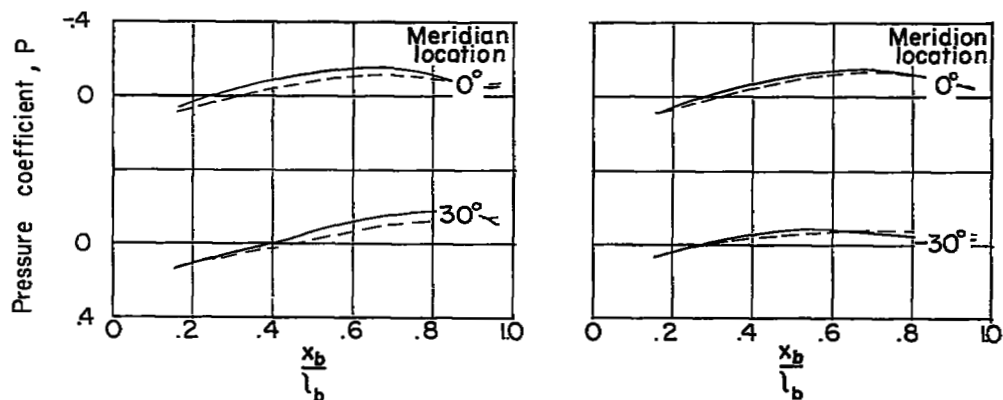
(d) $\alpha = 6.5^\circ$; $\beta = 0^\circ$.(e) $\alpha = 6.5^\circ$; $\beta = -4^\circ$.(f) $\alpha = 6.5^\circ$; $\beta = -8^\circ$.

Figure 14.- Concluded.

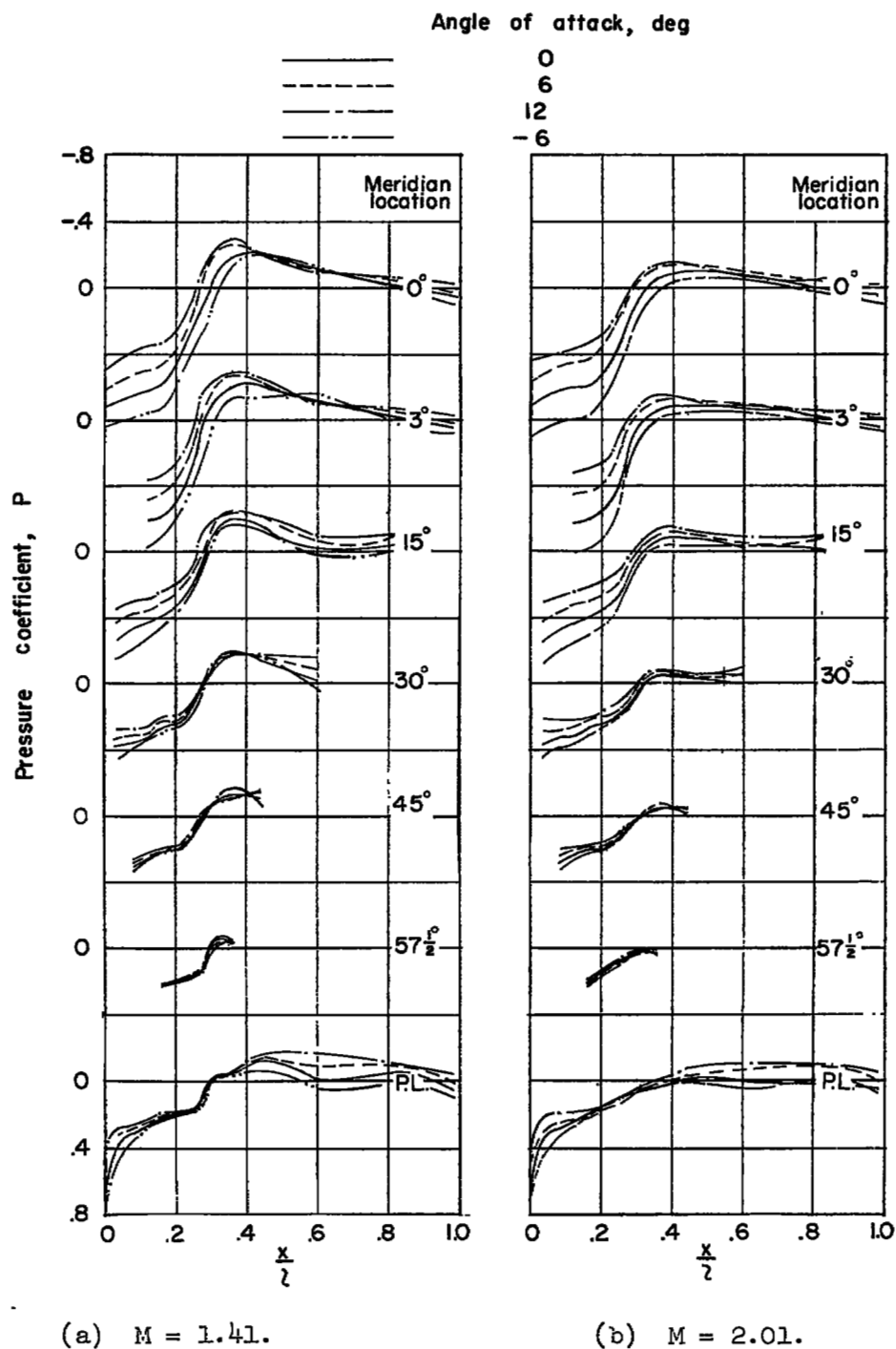


Figure 15.- Pressure distribution on round-windshield canopy in forward location at $M = 1.41$ and 2.01 for various angles of attack and 0.3° sideslip.

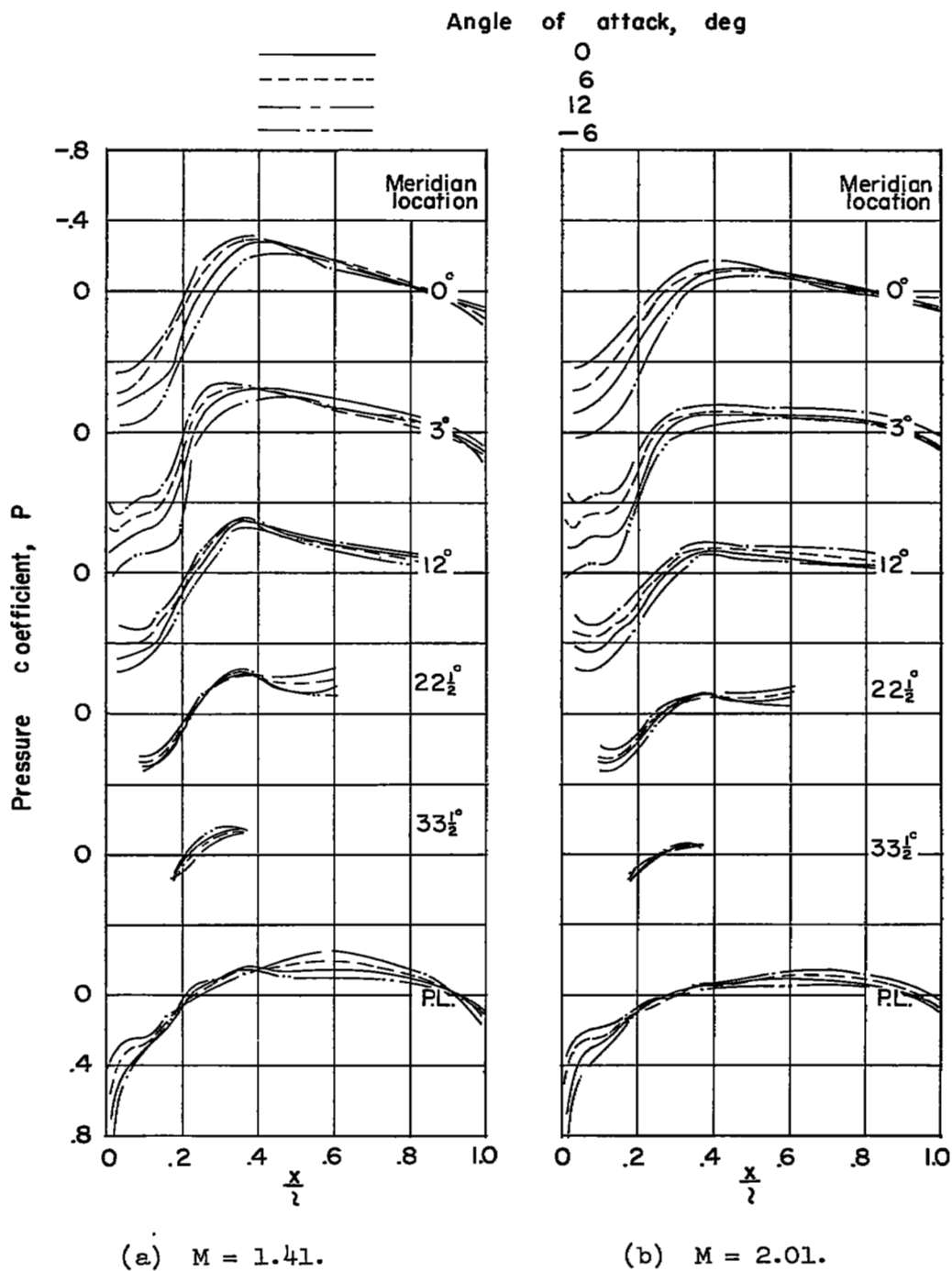


Figure 16.- Pressure distribution on round-windshield canopy in rearward location at $M = 1.41$ and 2.01 for various angles of attack and 0.3° sideslip.

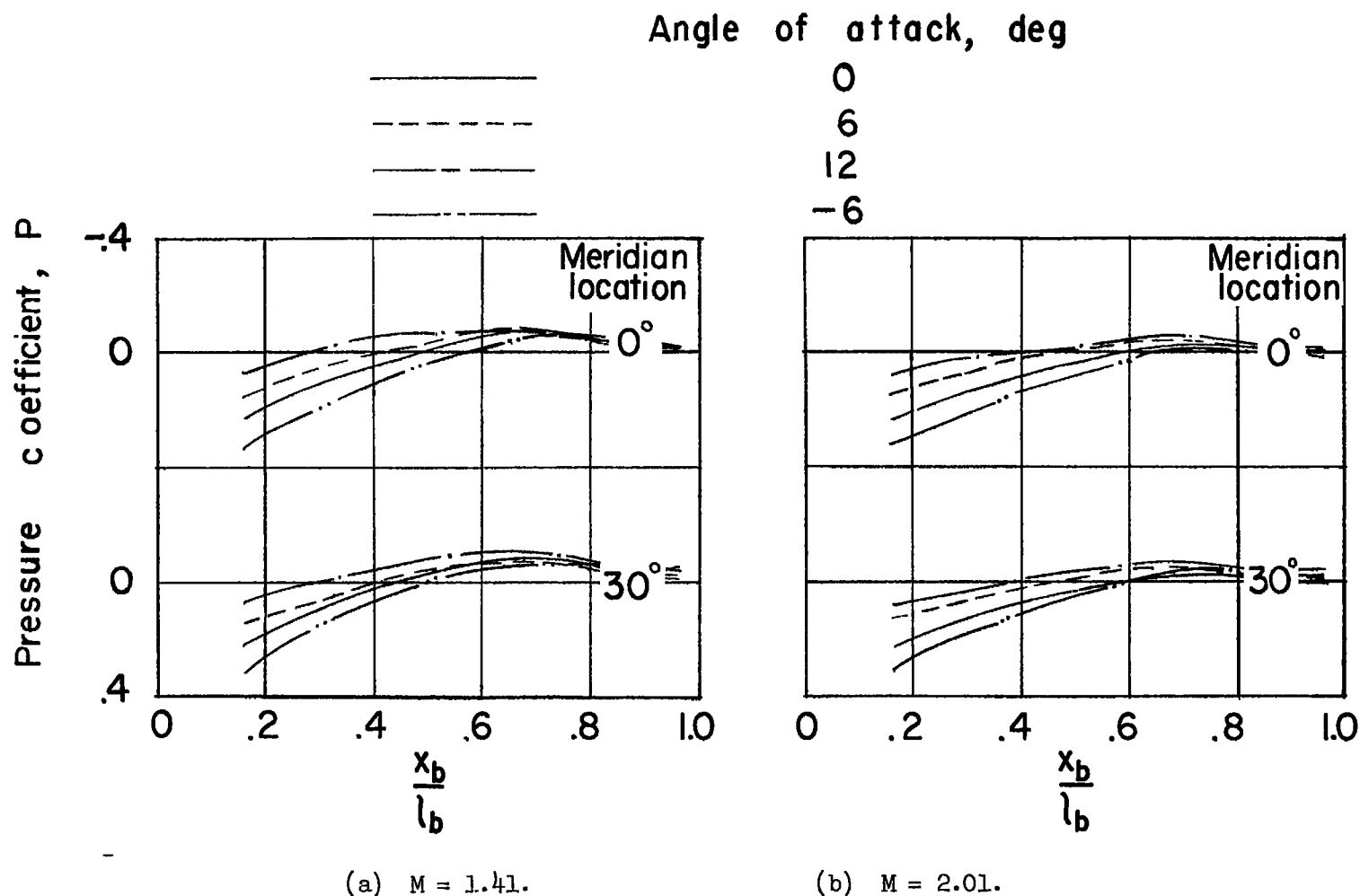
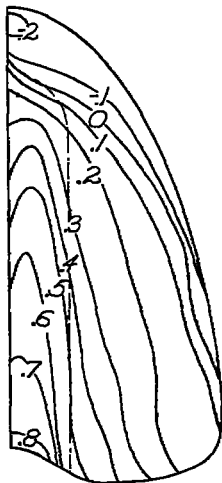
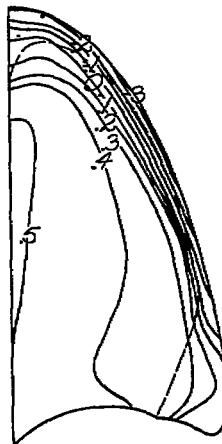


Figure 17.- Pressure distribution on body alone at $M = 1.41$ and 2.01 for various angles of attack and 0.3° sideslip.

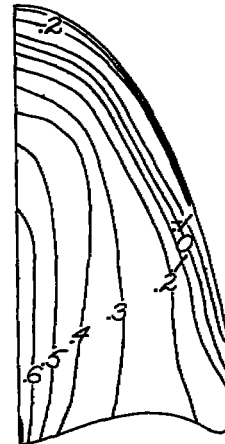
Flat Windshield

 $\alpha = 0.4^\circ$
 $\beta = 0^\circ$

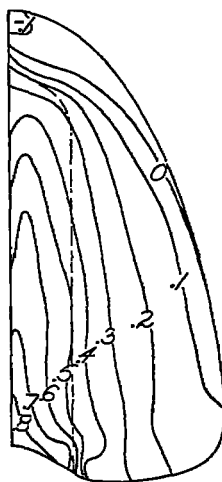
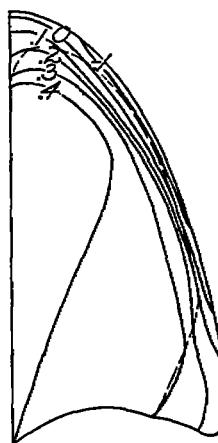
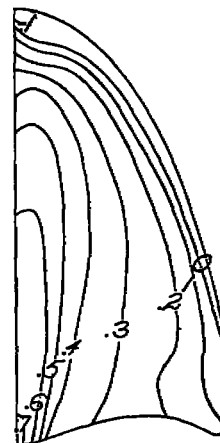
Vee Windshield

 $\alpha = 0.4^\circ$
 $\beta = 0^\circ$

Round Windshield

 $\alpha = 0^\circ$
 $\beta = 0.3^\circ$

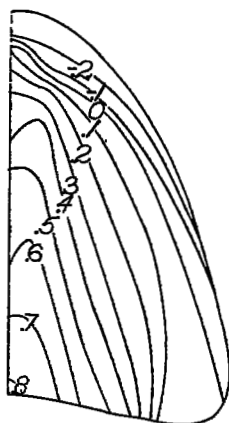
M=1.41

 $\alpha = 0.4^\circ$
 $\beta = 0^\circ$  $\alpha = 0.4^\circ$
 $\beta = 0^\circ$  $\alpha = 0.4^\circ$
 $\beta = 0^\circ$

M=2.01

Figure 18.- Pressure coefficient contours on one-half the frontal projections of each of the large forward-located canopies for $M = 1.41$ and 2.01 .

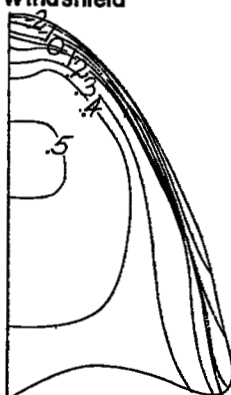
Flat Windshield



$$\alpha = 0.4^\circ$$

$$\beta = 0^\circ$$

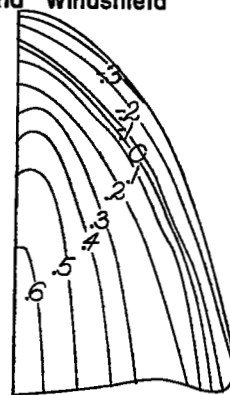
Vee Windshield



$$\alpha = 0.4^\circ$$

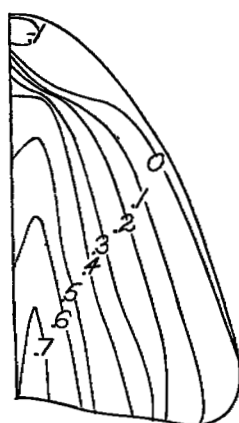
$$\beta = 0^\circ$$

Round Windshield


 $M = 1.41$

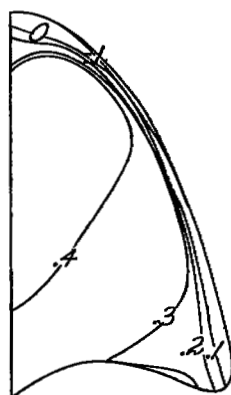
$$\alpha = 0.4^\circ$$

$$\beta = 0^\circ$$



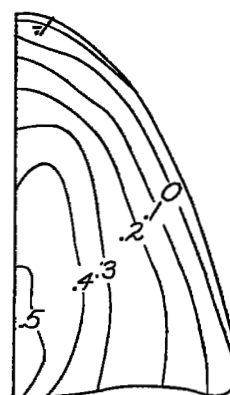
$$\alpha = 0.4^\circ$$

$$\beta = 0^\circ$$



$$\alpha = 0.4^\circ$$

$$\beta = 0^\circ$$


 $M = 2.01$

$$\alpha = 0^\circ$$

$$\beta = 0.3^\circ$$

Figure 19.- Pressure coefficient contours on one-half the frontal projections of each of the large rearward-located canopies for $M = 1.41$ and 2.01 .

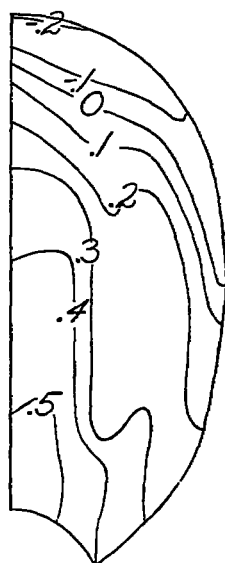
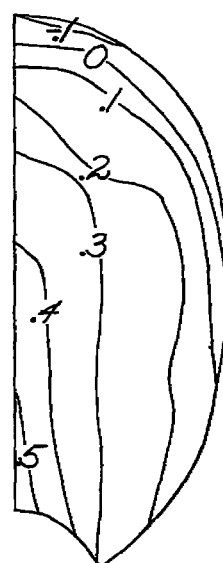
$M=1.41$  $\alpha = 0.4^\circ$
 $\beta = 0^\circ$ $M=2.01$  $\alpha = 0.4^\circ$
 $\beta = 0^\circ$ Forward-
located $\alpha = 0.4^\circ$
 $\beta = 0^\circ$  $\alpha = 0.4^\circ$
 $\beta = 0^\circ$ Rearward-
located

Figure 20.- Pressure coefficient contours on one-half the frontal projections of each of the small canopy configurations at $M = 1.41$ and 2.01 .

